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HYBRID THERMOCOUPLE DEVELOPMENT PROGRAM (U)

FINAL REPORT

by

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RCA Electronic Components

prepared for

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

NASA Lewis Research Center Contract NAS 3-11843 William J. Bifano, Project Manager

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	The design and development of a Hybrid thermocouple, having a segmented SiGe-PbTe n-leg encapsulated within a hollow cylindrical p-SiGe leg, is described. Hybrid couple efficiency is calculated to be 10% to 15% better than that of an all-SiGe couple. A preliminary design of a planar RTG, employing Hybrid couples and a water heat pipe radiator, is described as an example of a possible system application. Hybrid couples, fabricated initially, were characterized by higher than predicted resistance and, in some cases, bond separations. Couples made later in the program, using				
	improved fabrication techniques, exhibited normal resistances, both as-fabricated and after 700 hours of testing.				
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FORWORD

The work described herein was done at the Thermoelectric Products Engineering Department, RCA Corporation, under NASA Contract NAS 3-11843 with Mr. William J. Bifano, Direct Energy Conversion Division, NASA-Lewis Research Center, as Project Manager.

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Section I.

SUMMARY

The objectives of the Hybrid Thermocouple Development Program, as defined by Contract NAS 3-11843, are: (1) the design and development of a Hybrid thermocouple consisting of a segmented silicon-germanium/lead-telluride n-type leg encapsulated within a hollow cylindrical p-type silicon-germanium leg; (2) a preliminary design of a 250-watt (EOL) planar RTG employing Hybrid thermocouples; and (3) the fabrication of two representative module sections of the generator (nine couples in a three-by-three array) for testing at NASA Lewis Research Center. The program, initiated in January, 1969, was divided into five major tasks as follows:

- Task I. Thermocouple Parametric Design Analysis
 - II. Thermocouple Fabrication and Testing
 - III. Preliminary Converter Design
 - IV. Detail Thermoelectric Module Design
 - V. Fabrication of Modules

The design phase of the program (Tasks I, III and IV) is presented in Section III, Design and Analysis. Task II and Task V, dealing with fabrication and testing, are presented, respectively, in Sections IV and V.

A computer program was formulated to perform the detailed parametric analyses required to optimize the Hybrid couple efficiency for the design operating conditions selected. These operating conditions were as follows:

Hot Junction Temperature	1700°F (926°C)
Intermediate Temperature (PbTe Hot Junction)	1000°F (538°C)
Cold Junction Temperature	450°F (232°C)
Incident Heat Flux	2.0 watts/cm ²

The study indicated that the efficiency of the Hybrid couple would be 10 to 15 percent better than an all-SiGe (63 at.% Si) couple. As a result of the detailed parametric analyses performed and the couple development effort conducted, a Hybrid thermocouple reference design was established with the following characteristics:

Performance	Couple Design B
Power per thermocouple*	0.84 watts
Voltage*	0.185 volts
Current*	4.54 amperes
Internal Resistance*	0.033 ohm
Thermocouple Efficiency*	7 . 35 %

* Based on analyses which included 5% extraneous resistance and 7% thermal (shunt) loss. Subsequent programs have indicated that 20% extraneous resistance is more realistic. This would reduce couple design B power, for example, from 0.84 to 0.77 watts per couple.

Geometrics Couple Design B Area ratio: $A_n(n-PbTe)/A_p(p-SiGe)$ 1.6 Area ratio: $A_n(SiGe)/A_n(PbTe)$ 0.7 Radius of n-type PbTe 0.476 cm (0.1875 inch) Wall thickness of p-SiGe 0.117 cm (0.046 inch)Gap between n-PbTe and p-SiGe legs 0.076 cm (0.030 inch)2.86 cm (1.125 inch)Length of p-type SiGe Length of n-type SiGe 1.81 cm (0.714 inch)0.797 cm (0.314 inch)Length of n-type PbTe $5.9 ext{ cm}^2 (0.914 ext{ inch}^2)$ Heat receptor area Heat receptor size $2.44 \times 2.44 \text{ cm}$ $(0.96 \times 0.96 \text{ inch})$ 1.34 cm (0.527 inch)Thermocouple - OD (0.435 inch) Thermocouple - ID 1.1 cmn-type SiGe - OD 0.796 cm (0.314 inch)n-type PbTe - OD 0.953 cm (0.375 inch)

In the development of the thermocouple structure, a number of thermal stress problems were encountered initially in testing of Hybrid thermocouples. Typically, mechanical separation occurred in the cold junction bond of the p-SiGe and the hot junction bond of the n-PbTe segments. Although these same bonds maintained integrity during subcomponent testing, apparently somewhat higher stress levels were encountered in testing of the complete couple assemblies. This problem was corrected by insertion of a gold ring, formed from 0.127 cm (0.050 inch) OD gold tubing, between the p-SiGe tungsten cold shoe and the cold stack prior to the final braze operation to add compliance to the structure.

A total of four couples completed the specified test time of 5000 hours at reference design temperature conditions. Thermal cycling was also performed during these tests.

A preliminary design of a 250-watt (EOL) planar RTG using the reference design Hybrid thermocouples and a water heat pipe radiator was prepared, and its performance is summarized below.

Converter Parameter	Converter B III	
Power Output (BOL)	273 watts	
Efficiency	6.4 %	
Hot Junction Temperature	926°C (1700°F)	
Cold Junction Temperature	232°C (450°F)	
Converter Weight	33.8 kg (76 lbs)	
Heat Pipe Radiator Weight	10.2 kg (23 lbs)	
Fuel Source Weight	32.5 kg (73 lbs)	
Total Generator Weight	78.2 kg (172 lbs)	
Specific Power (BOL)	3.5 w/kg (1.59 w/lb)	

The reference Hybrid converter design is based on the predicted performance of couple design B, above. Although preliminary test results to date are encouraging, much more testing and evaluation of Hybrid couples are required

to establish this concept. Further improvements in Hybrid couple performance might be achieved with additional development effort. Some of the possible design modifications and their corresponding predicted performance are presented below.

	Couple <u>Efficiency</u>	Converter Efficiency	Converter Specific Power (BOL)
Reference Design B-III	7.4%	6.4%	3.5 w/kg (1.59 w/lb)
(1) Thermoelectric Materials 80% n-SiGe - RCA ternary n-PbTe	8.5%	7.5%	4.1 w/kg (1.86 w/lb)
(2) Hot Shoe Temperature - 1000°C and Thermoelectric Materials	9.4%	8.4%	4.58 w/kg (2.08 w/lb)

Hybrid couples, fabricated initially, were characterized by higher than predicted resistance and, in some cases, bond separations. Couples made later in the program, using improved fabrication techniques, exhibited normal resistances both as-fabricated and after 700 hours of testing.

At the conclusion of the program, two module sections of the reference design Hybrid module were fabricated and delivered to NASA Lewis for testing.

Section II.

INTRODUCTION

The thermoelectric materials most often considered for purposes of energy conversion in space are alloys of silicon and germanium and alloys of lead and tellurium. Silicon-germanium alloys are attractive in terms of mechanical strength, machineability and their unprotected operation in either air or vacuum environments at temperatures up to 1000°C (1832°F). The ntype SiGe alloys exhibit a slight change in electrical resistivity and Seebeck coefficient with time because of a temperature dependence of the solid solubility of dopant (see Figure 76, Appendix I). While these effects are self-cancelling to some extent, the electrical resistivity change is considerably greater than the increase in Seebeck coefficient, thus resulting in a slight decrease in thermoelectric performance with time. The largest change occurs at the intermediate temperatures, i.e., 200° C to 600° C, due to a combination of a highly supersaturated dopant-SiGe solid solution and a reasonable rate of dopant diffusion. Metallurgical bonds are employed for both hot and cold contacts in the SiGe thermocouples.

Lead telluride (PbTe) alloys can be operated from room temperature to about 600°C (1100°F); however, they must be protected from the environment in most applications. The main attractiveness of tellurides is their excellent ability to convert heat to electricity, reflected by the so-called "figure-of-merit". Up to temperatures of about 550°C (1022°F), the tellurides have a higher figure-of-merit than the SiGe alloys. Although some problems have been encountered with metallurgically bonded PbTe couples, indications are that the n-type PbTe bonds are reliable.

A careful review of the operating characteristics of both alloys and results of preliminary development of a low temperature "Hybrid" thermocouple conducted by RCA, suggested a high temperature, 926°C (1700°F), Hybrid thermocouple design having a segmented SiGe--PbTe n-leg and a SiGe p-leg. The low temperature, 538°C (1000°F), Hybrid couple, built and tested in 1968, consisted of a n-PbTe leg encapsulated within a p-SiGe cylindrical leg as shown in Figure 1. The high temperature Hybrid thermocouple, described herein, employs SiGe air-vac technology and has a segmented n-leg. n-type PbTe is used up to temperatures of about 538°C (1000°F), and n-type SiGe at higher temperatures. This arrangement takes advantage of the high figure-of-merit of PbTe below 538°C (1000°F) and stable performance of ntype SiGe above 538°C (1000°F). The p-leg is fabricated in the form of a hollow cylinder and used to encapsulate the segmented n-leg, thereby protecting the PbTe. This high temperature Hybrid thermocouple configuration should provide a high performance device with stable electrical characteristics. Details of the proposed Hybrid couple design are shown schematically in Figure 2. Both ends of the p-type SiGe cylinder are metallurgically bonded, one to the silicon-molybdenum hot shoe and the other to the coldstack assembly. The cold-stack assembly is attached to a mounting stud. Two electrical insulators in the cold-stack assembly provide electrical isolation of the two thermocouple legs from each other and from the stud. Each leg is contacted to an electrical connector. The n-type PbTe thermo-

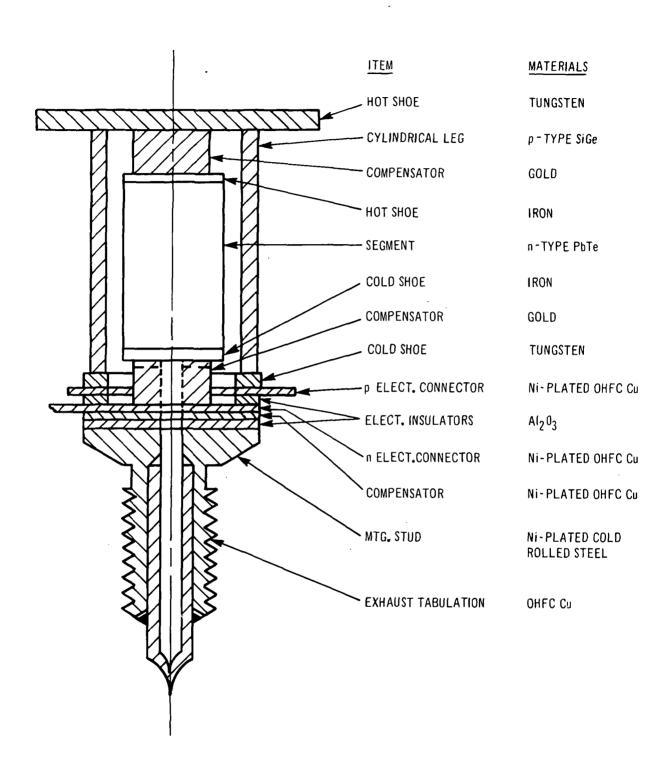


Figure 1. Hybrid Thermocouple Prototype Design

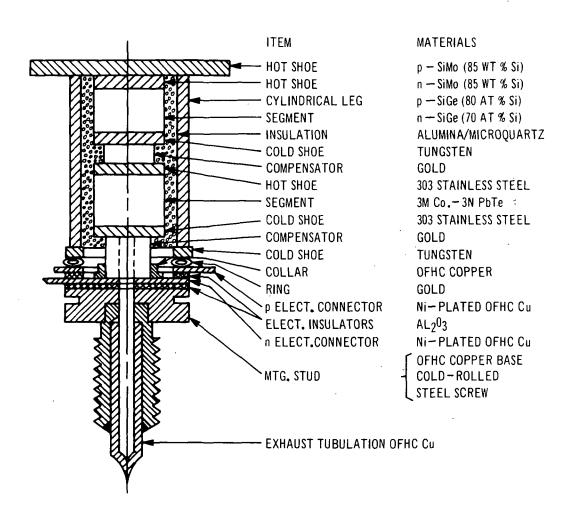


Figure 2. Hybrid Thermocouple Assembly

element is contacted to metal shoes. The n-type PbTe cold shoe is bonded through a stress compensator to the cold-stack assembly. The n-type PbTe hot shoe is bonded through a stress compensator to the cold shoe of the n-type SiGe. The hot end of the SiGe element is bonded to the SiMo hot shoe. The space between the p-type and n-type thermoelements is filled with thermal insulation. Finally, the couple is sealed in an inert-gas atmosphere to inhibit sublimation of the PbTe.

The Hybrid thermocouple efficiency is predicted to exceed that of an all-silicon-germanium thermocouple by approximately 10-15% depending upon the operating temperatures.

On January 23, 1969, a contract was awarded RCA Electronic Components, Harrison, N.J., by NASA Lewis Research Center to design and develop the Hybrid thermocouple concept. The overall objectives were:

- 1. The design and development of a Hybrid thermocouple consisting of a segmented silicon-germanium/lead-telluride n-type leg encapsulated within a SiGe p-type concentric leg.
- 2. The fabrication and delivery to NASA Lewis Research Center of two flat plate module sections of a thermoelectric converter employing Hybrid thermocouples. Each module section was to have a projected area of between 9 and 16 square inches.

To accomplish these objectives, the program was divided into five major tasks as follows:

- Task I. Thermocouple Parametric Design Analysis
 - II. Thermocouple Fabrication and Testing
 - III. Preliminary Converter Design
 - IV. Detail Thermoelectric Module Design
 - V. Fabrication of Modules

The results of Tasks I, III and IV are presented in total in Section II, Design and Analysis; Task II and Task V are presented, respectively, in Sections IV and V.

Section III.

DESIGN AND ANALYSIS

A. Design Requirements

The primary design objective of this program is to develop a flat plate thermoelectric module using Hybrid thermocouples, capable of integration into a 250-watt thermoelectric planar converter. A heat pipe radiator is to be considered analytically (no hardware development) for the heat sink. The performance objectives are given below.

- 1. Converter Power 250 watts (EOL)
- 2. Converter Life

Modules shall produce the design power at the end of five years.

3. Operating Temperatures

Hot Junction - 926°C (1700°F) maximum Interface (PbTe Hot Junction) - 538°C (1000°F) maximum Cold Junction - 316°C (600°F) maximum

- 4. Incident Heat Flux 2 watts/cm²
- 5. Thermoelectric Material Properties

	SiGe	PbTe
Temperature	900°C (1674°F)	500°C (954°F)
Seebeck Coefficient (uV/°C, max)	270	230
Electrical Resistivity (ohms-cm, max)	0.003	0.003
Thermal Conductivity (watts/cm-°C, max	0.045	0.017

6. Performance Degradation

Thermoelectric Material	10%	maximum
Electrical Insulation	5%	maximum
Extraneous Resistance (joints, bonds, etc.)	5%	maximum

7. Environmental Conditions

The module shall be designed to meet the following requirements, but no environmental testing of the module is required under this contract.

a. Acoustic Noise: The module shall be capable of withstanding for five minutes a total integrated sound pressure level of 152 decibels referred to 0.0002 dyne per square centimeter.

b. Vibration: The module shall be capable of withstanding sinusoidal input applied at the mounting points for a period of 15 minutes along each of three mutually perpendicular axes as follows:

5-33 cps at 0.14 inch D.A. displacement 33-140 cps at 8.0 G's peak 140-240 cps at 0.003 inch D.A. displacement 240-2000 cps at 15.0 G's peak

- c. Acceleration: The module shall be capable of withstanding 6 G's acceleration for five minutes in both directions along three mutually perpendicular axes.
- d. Shock: The module shall be capable of withstanding a 20-G shock along each of three mutually perpendicular axes. The waveshape shall be a half-sine pulse of 10-millisecond duration.
- e. Thermal Transient: The module shall be capable of withstanding the following thermal transient test. It shall be heated to an absorber surface temperature of 1700°F as rapidly as possible when subjected to the thermal flux required to achieve this hot junction temperature. Immediately after temperature equilibrium has been established, heating shall cease. The module shall be allowed to cool to an equilibrium temperature by radiating to the surrounding walls of a water-cooled chamber. When a steady temperature has been obtained, the absorber surface shall again be heated to a temperature of 1700°F and allowed to cool to a steady temperature in the water-cooled chamber.

B. Hybrid Thermocouple Design

1. Performance of the Hybrid Thermocouple

The design of the Hybrid thermocouple was undertaken with the object of optimizing its conversion efficiency, consistent with physical and manufacturing constraints. A computer program was established in order to perform the detailed parametric analyses required to optimize the Hybrid couple efficiency for the design operating conditions selected. The thermoelectric materials used in the <u>initial</u> Hybrid couple design analyses are listed below.

```
n-type PbTe segment - 3M Co.'s 3N alloy
n-type SiGe segment - RCA 63.5 At.% Si-SiGe alloy
p-type SiGe cylinder - RCA 63.5 At.% Si-SiGe alloy
```

The Hybrid couple computer design program and the thermoelectric material properties are presented in detail in Appendix II, Sections A and C-1, respectively.

Figure 3 shows the physical model on which the analytical program was based. The thickness dimensions, materials, and geometrical configuration are based on the earlier RCA development effort but do not represent a specific design. Because the design of the cold stack of both the n-and p-type element legs depends on fabrication constraints, availability of materials, and cost, it is not represented as it would be fabricated, but rather as a representative cold stack for analytical purposes. The couple efficiency is not highly sensitive to small variations in cold-stack temperature drop because these variations are small compared to the total temperature difference, Δ T, that appears across the total length of the thermoelectric element. When selecting the final design configuration, every effort was made to minimize the temperature drop across the cold stack.

In addition to the thermoelectric materials, the important input parameters which affect the Hybrid thermocouple efficiency are the incident heat flux density, the hot and cold junction operating temperatures, and the geometry. To permute the many values and variables associated with each of these parameters would result in design data too cumbersome to analyze. Therefore, some of the parameters and their values were fixed in order to limit the number of cases generated.

The values of input parameters used initially are as follows:

a. Operating Conditions

- 1) Heat receptor temperature (THP): 926°C (1700°F)
- 2) Interface temperature of PbTe hot junction (THN2C): 538°C (1000°F)
- 3) Cold junction temperature of p-type element (THP): 232°C (450°F)
- 4) Incident heat flux densities considered (PHI): 1, 3, 5, watt/cm²

b. Couple Geometrics

- 1) P-type element length (FLP): 2.54 cm (1.00 in.), 3.17 cm (1.25 in.), 3.81 cm (1.50 in.)
- 2) Ratio of n-type PbTe area to p-type element area (ANP): 1.0, 1.4, 1.8, 2.2
- 3) Ratio of n-type SiGe area to n-type PbTe area (ANR): 0.4, 0.7, 1.0
- 4) Radius of n-type PbTe element (RRN): 0.406 cm (0.16 in.), 0.508 cm (0.20 in.), 0.61 cm (0.24 in.)

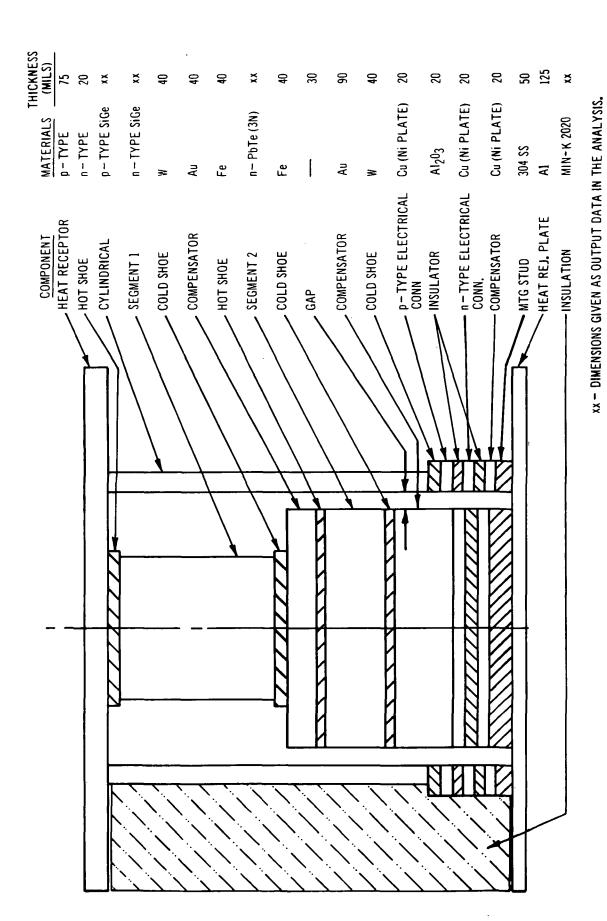


Figure 3. Physical Model of Hybrid Couple for Cases 1B to 1E

The capital letters in parentheses are the code names of the input variables used in the computer program. A complete list of computer symbols is given in Appendix II, Section D-1. Briefly, the program, through an iterative process, successively approximates a consistent set of values for the heat flowing through the element legs and for the proportionate lengths of the n-type segments. Having converged these values to 0.1 percent, the program continues to determine the proper heat receptor size (p-type SiMo hot shoe size) for the desired incident heat flux density. The various input parameters are sequentially permuted to obtain the couple design yielding optimum efficiency. The effects of these input variables on the Hybrid couple design and the resultant output data are presented in Figures 4 through 12.

The following discussion primarily involves couple materials, geometrics, and operating conditions. (Constrains such as part availability, costs, and mechanical limitations will be considered later.)

Figure 4 shows the relationship of incident heat flux density and p-type element length to couple efficiency. In general, it is desirable to make the couple as long as possible and to design for the highest possible heat flux density. Note that as the incident heat flux density is increased (from 1 to 3 watts/cm², for example), the relative gain in efficiency as a result of increasing element length is reduced.

Figure 5 shows the effect of varying heat receptor area on couple efficiency for various values of incident heat flux density and n-type PbTe radius for a p-type element length of 1.25 inches. At the time of the analysis, fabrication considerations limited the SiMo heat receptor size to an area of $14.5~\rm cm^2$ (1.5 in. x 1.5 in. square hot shoe). Subsequent development of hot pressing procedures has essentially removed this constraint.

Figure 6 shows that varying the diameter of the n-type PbTe element has minimum effect on efficiency, whereas the n-PbTe/p-SiGe area ratio, ANP, more significantly affects the efficiency.

Figure 7 shows the effect of varying the incident heat flux density and the ANP ratio on couple efficiency. Generally, as the flux density decreases, so does the effect of ANP on efficiency. In this case, as will be seen later, a lower ANP ratio is desirable in order to maintain adequate p-type SiGe element wall thickness.

Figure 8 shows couple efficiency plotted against the ratio of n-type SiGe area to n-type PbTe area (ANR). The need to consider ANR values less than one becomes apparent when analyzing the relative n-type segment lengths of SiGe and PbTe. (See Figure 12.) Reducing the value of ANR below 1.0 has the effect of lengthening the PbTe segment with an attendant decrease in the n-SiGe segment length, assuming fixed

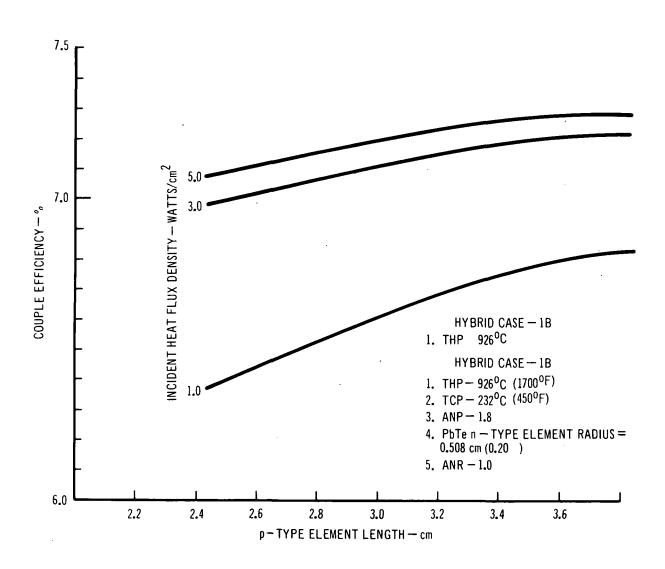


Figure 4. Couple Efficiency Vs p — Type Element Length for Various Heat Flux Densities

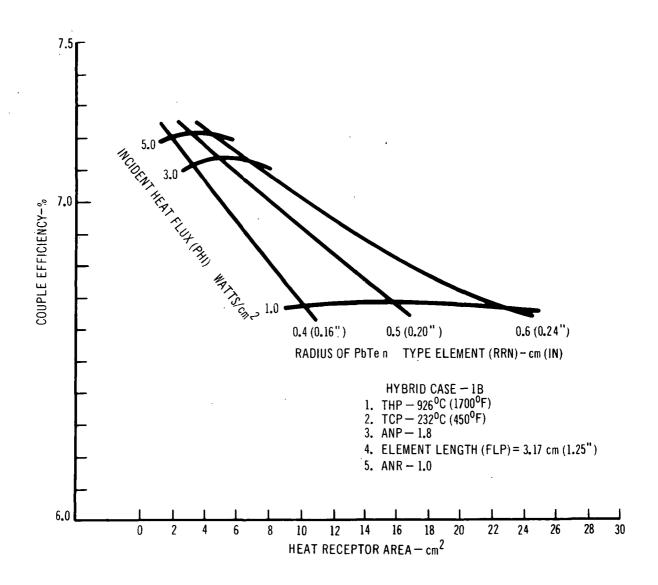


Figure 5. Couple Efficiency Vs Heat Receptor Area For Various Heat Flux Densities and Radii of n PbTe Element

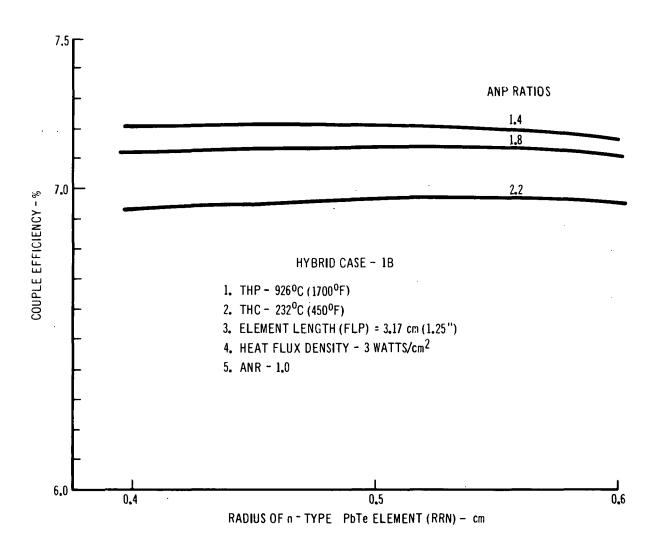


Figure 6. Couple Efficiency Vs. Radius n Type PbTe Element for Various ANP Ratios

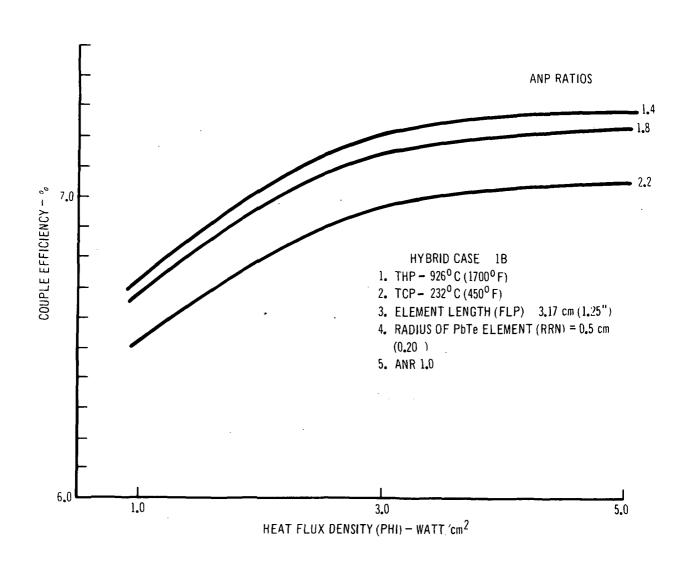


Figure 7. Couple Efficiency Vs. Heat Flux Density For Various ANP Ratios

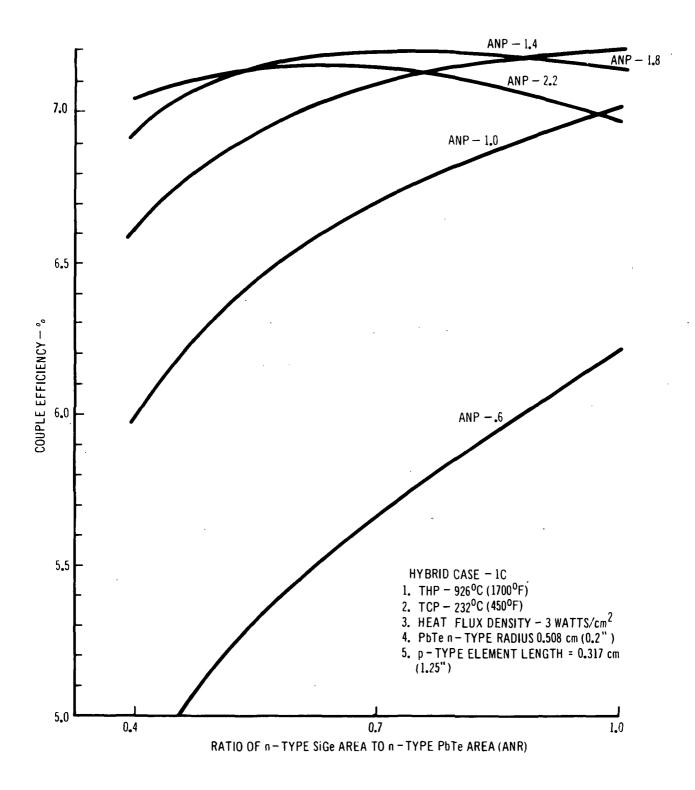


Figure 8. Couple Efficiency Vs. Ratio of n-Type SiGe To n-Type PbTe Area (ANR)
For Various AN TO AP Ratios (ANP)

junction temperatures. With an ANR near 1.0 and an overall couple length of approximately one inch, the PbTe segment assumes the geometric proportions of a tablet. From a thermal stress point of view and in order to minimize the effect of contact resistance, it is desirable to lengthen the PbTe element. Figure 8 was constructed to illustrate the extent to which the n-PbTe length could be increased without compromising couple efficiency.

Figure 9 illustrates the effect of input variables of FLP (p-type element length), ANP, and ANR on couple efficiency. A couple length of 3.17 cm (1.25 inches), an ANP of 1.8 and an ANR of 0.7 represent a reasonable compromise for lengthening the n-type PbTe segment while maintaining a relatively high efficiency.

Figure 10 again shows the effect of ANR and ANP on couple efficiency to give a better definition of where the optimums occur relative to efficiency. For the expanded efficiency scale drawn, ANP values between 1.6 and 1.8 would affect efficiency less than 0.1% for an ANR equal to 0.7. Also, lower ANP ratios would result in an increase in the wall thickness of the p-type element cylinder.

Figures 11 and 12 illustrate the important geometrical relationships involved in fabricating a Hybrid couple. The most important of these curves is Figure 11, relating wall thickness of the p-type element cylinder to ANP and the radius of the n-PbTe segment. The constraining factors here involve fabrication costs and hermeticity; both problems diminish as the wall thickness increases. The lower practical limit of wall thickness is about 40 mils (0.106 cm).

Based on the parametric analyses of the Hybrid couple, three couple designs (see Table I) were proposed as the design basis for initiation of Task II, Thermocouple Development and Testing. The data presented in Table I indicate the operational characteristics of these designs as well as the associated geometrical factors. The intent of these couple designs was to determine the capability of fabricating couples with varying geometries to meet expected requirements of operating conditions while maintaining optimum efficiency.

These designs covered a broad range of geometrical sizes and shapes for couples intended for operation with heat sources providing 1.0, 3.0, and 5.0 watts per cm² incident heat flux densities. Although a range of couple geometries was covered, the operating incident heat flux densities did not appear wholly consistent with heat sources that would be available for use with thermoelectrics in the foreseeable future. Consequently, the Hybrid couple design was reoriented for use with a heat source having a heat flux density of 2.0 watts per cm². This change was proposed by NASA-LeRC as being more representative of future heat source capability. Furthermore, investigation was recommended of various couple diameters as well as of the previously proposed variations in heat receptor sizes and element length. Varying the diameter of the couples, in effect, varies

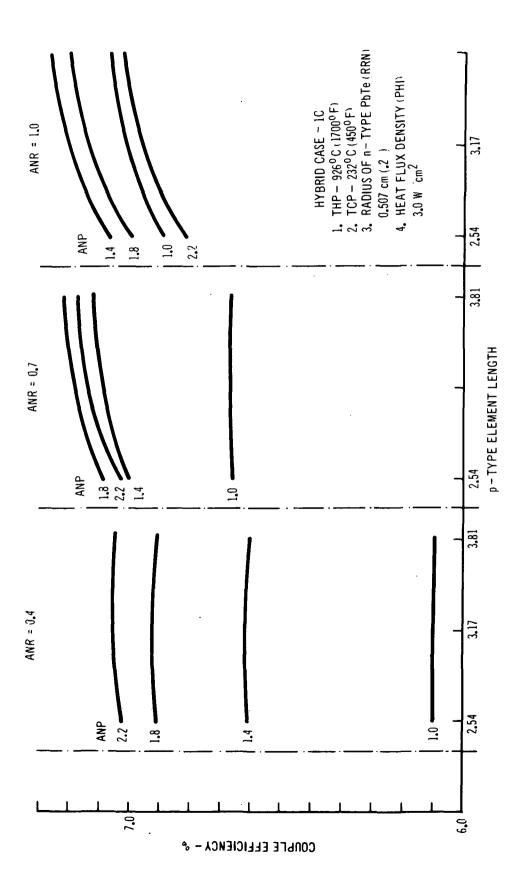


Figure 9. Couple Efficiency Vs. p - Type Element Length for Various n - Type SiGe to PbTe Area Ratios (ANR) and n-Type PbTe Area to p-Type SiGe Area Ratios (ANP)

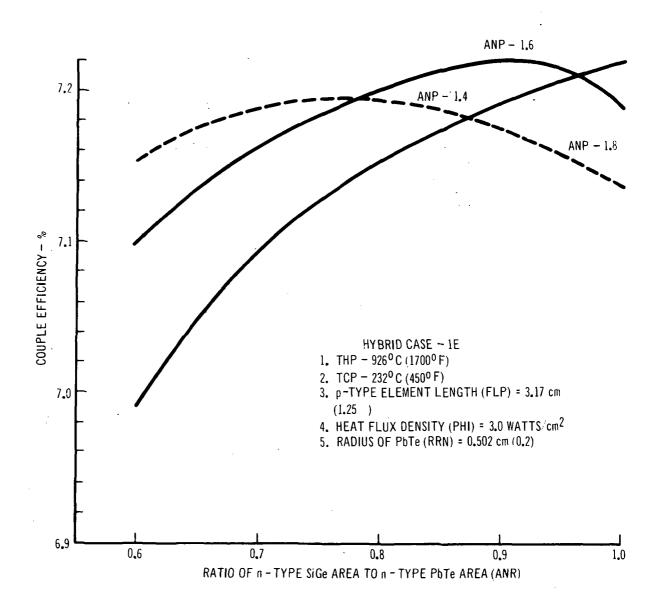


Figure 10. Efficiency Vs. Ratio of n - Type SiGe Area n - Type PbTe Area (ANR) for Various AN/AP Ratios (ANP)

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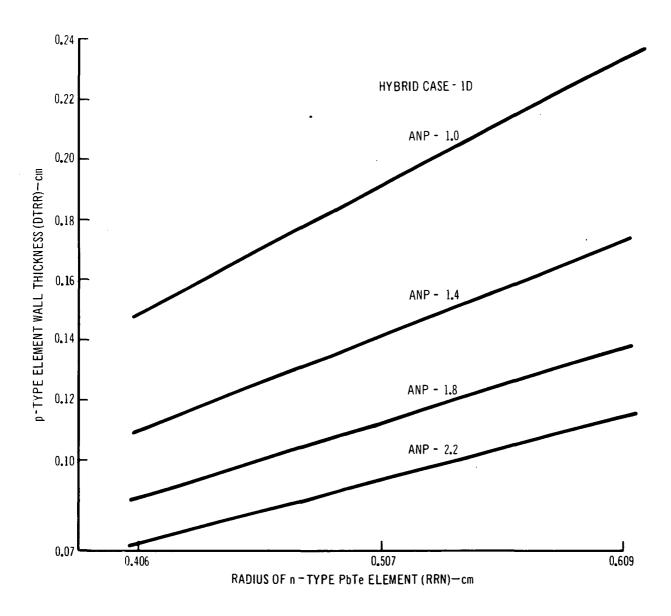


Figure 11. p - Type Element Wall Thickness Vs. Radius of n - Type PbTe Element for Various n - Type PbTe Area to p - Type SiGe Area Ratios (ANP)

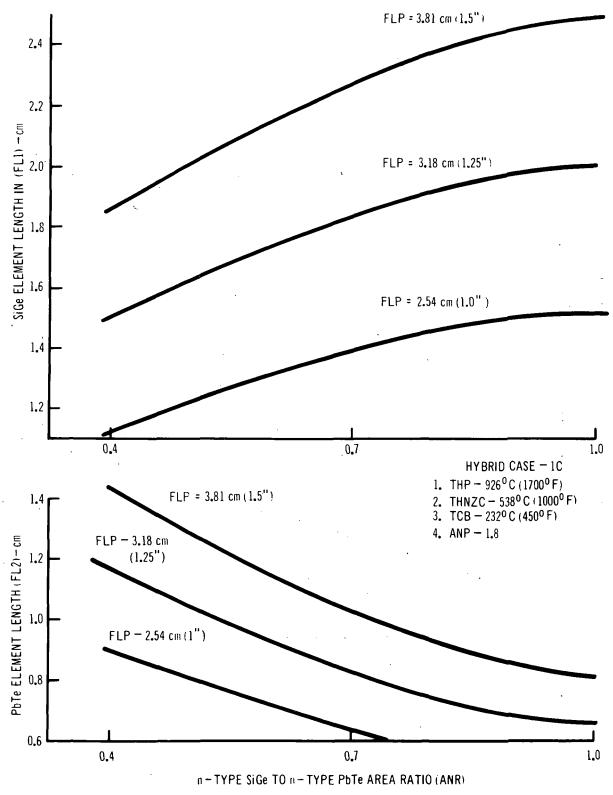


Figure 12. Length of n-Type SiGe and n-Type PbTe Segments Vs. n-Type SiGe to n-Type PbTe Area Ratio (ANR) for Various p-Type Element Lengths

TABLE I
PROPOSED HYBRID COUPLE DESIGNS

<u>DESIGN PARAMETER</u>	S Units	A	<u> </u>	<u>C</u>
<u>Operational</u>				
1) Couple Efficiency 2) Power per Couple 3) Voltage 4) Current 5) Internal Resistan 6) Hot Junction Temp 7) Cold Junction Tem 8) Interface Temp. 9) Incident Heat Flu	watts volt Amps ce ohms °C/°F °C/°F	7.7 0.737 0.186 3.962 0.0393 926/1700 232/450 538/1000 1.0	7.7 0.886 0.186 4.763 0.0326 926/1700 232/450 538/1000 3.0	7.6 1.11 0.186 5.968 0.026 926/1700 232/450 538/1000 5.0
Geometrics				
1) RRN - Radius of no. 2) ANP - An/Ap 3) ANR - An(SiGe)/An 4) DTRR - Wall Thick 5) GAP - Gap between 6) FLP - Length of po. 7) FL1 - Length of no. 8) FL2 - Length of no. 9) AH - Heat Recepto: 10) FH - Heat Recepto: 11) Thermocouple OD (1)	(PbTe) ness (in.) n-p legs (in.) -type SiGe (in.) -type SiGe (in.) -type PbTe (in.) r Area (in.2) r Dimension (in.)	0.1875 1.6 0.7 0.046 0.030 1.500 0.910 0.390 1.500 1.225 x 1.225 0.531	0.1875 1.6 0.7 0.046 0.030 1.250 0.720 0.330 0.590 0.77 x 0.77 0.531	0.1875 1.6 0.7 0.046 0.030 1.000 0.550 0.250 0.460 0.68 x 0.68 0.531

the wall thickness of the p-element cylinder, assuming a constant A_n/A_p ratio. The wall thickness depends upon fabrication limitations which, in turn, becomes a design constraint.

As a result of a design review meeting, a consistent set of Hybrid couple design conditions were again developed. The incident heat flux density was fixed at 2.0 watts per cm², for a hot junction temperature of 926°C (1700°F), a cold junction of 232°C (450°F), and an interface temperature of 538°C (1000°F). In addition, three couple diameters and element lengths were permuted. The design variations for the above operating conditions and geometric inputs are covered in the nine cases of Table II.

Analyses of these designs indicated that the greatest spread of geometrics existed in designs A-3, B-2, and C-1. Based on the current state-of-the-art fabrication techniques, design B-2 comes closest to representing current capability. Cases A-3 and C-1, however, typify designs which expand current technology in terms of element length, wall thickness, and couple diameter. Hence, these three designs were selected as offering the desired variation in couple geometrics.

Closer inspection of designs A-3 and C-1 in Table II revealed that minor changes in dimensions could be made based on certain practical considerations. With reference to the A-3 design, for example, the 0.150-inch radius of the n-type PbTe segment is a non-standard size. Modifying this radius, however, to 0.125 inch would permit purchase of standard material with the attendant benefits of cost and immediate availability. Consequently, design A of Table III was adopted as a practical version of design A-3 of Table II. Moreover, design A enhances the spread of couple geometrics to be investigated.

Design C-l of Table II requires a p-type element cylinder of a diameter which exceeds current "state-of-the-art" ingot sizes, the maximum being 0.600 inch in diameter. Thus, in design C of Table III, the radial dimensions have been accordingly reduced to meet the fabrication constraint placed on the 0.D. of the p-type cylinder.

The three designs of Table III, therefore, represent a compromise in terms of operating conditions, couple geometrics, and practical material considerations, while maintaining near-optimum operating efficiencies for the designs cited. The indicated Hybrid couple efficiencies of 7.1 to 7.5 percent represent a beginning-of-life improvement of 10 to 15 percent over an all-SiGe thermocouple (63 at.% Si alloy) operating at the same junction temperature. These couple designs formed the basis for the subcomponent and thermocouple development performed in Task II-A, 1 and 2.

TABLE II
PROPOSED HYBRID COUPLE DESIGNS

* See explanation at bottom of Table III

TABLE III

HYBRID COUPLE DESIGNS SELECTED FOR SUBCOMPONENT DEVELOPMENT PROGRAM

0.076(0.030 1.42(0.558 1.58(0.622 0.571(0.225 0.142(0.056 0.613(0.241 2.54(1.00) 1.570 0.186 8.45 926(1700) 538(1000 232(450) 0.476(0.1875 0.117(0.046 0.076(0.030 1,87(0,736 1.34(0.527 0.798(0.314 5.69(0.882 3:17(1.25 0.886 4.76 0.186 926(1700 538(1000 232(450) 7:35 1.6 В 0.381(0.150 0.091(0.036) 0.076(0.030 2.32(0.915 0.983(0.387 3.05(0.473 1.10(0.432 3.81(1.50) 0.476 0.186 926(1700 533(1000 0.061 232(450) 2.56 1.6 0.7 7.5 DESIGN PARAMETERS watts/cm² cm2(in.2 cm(in.)cm(1n.) Units Units watts volts $(\mathbf{4}_{\circ})\mathfrak{I}_{\circ}$ amps ohms ઝ્લ n-leg to p-leg area ratio, $\mathrm{An/Ap}$ n-SiGe to n-PbTe area ratio, $A_{ m nJ}$ Sold junction temperature, TCJ Hot junction temperature, THJ Length of n-type PbTe, 12 Interface temperature, TI Length of notype Side, L p-leg wall thickness, t Length of p-type Sige, Gap between n-leg and] Operational Geometrics Internal resistance* Heat receptor area p-Sige cylinder OD Incident heat flux Jouple efficiency* Power per couple* PbTe element, rn Voltage* Current*

Based on analysis which included 5% extraneous resistance and 7% thermal (shunt) loss. Subsequent programs have indicated that 20% extraneous resistance is more realistic. This would reduce design B power, for example, from 0.886 to 0.77 watts per couple.

*

2. Stress Analysis of Hybrid Thermocouple Components

Two significant areas that are related to the effect of thermal stress on the Hybrid couple are discussed below.

(1) First, stresses are induced in the thermoelements as a result of differential expansion between the couple legs. The stresses induced axially by the differential expansion were relaxed by the use of gold, a highly ductile material. How gold yields in this particular situation is not known quantitatively; however, the percentage of elongation through which the gold must move can be calculated and compared to acceptable limits. Moreover, previous RCA development efforts in the Hybrid program demonstrated the ability of gold to sufficiently relax the axially-directed stresses caused by thermal expansions. An early-design low-temperature Hybrid couple which contained gold withstood more than 3000 hours of stable operational life after enduring several thermal cycles to room temperature.

The differential expansion between both couple legs was calculated to determine the percentage of elongation or of contraction through which the gold would have to move. The calculation was based on the differential expansion that occurs between 400°C, the final bonding temperature where axial stresses are minimal, and room temperature where the stresses are more severe. Another situation exists in which the couple is operating at typical temperatures of 926°C and 232°C, and producing thermally=induced stresses, but they are less severe than those in the isothermal case at room temperature; consequently, this case was not pursued.

For the case in which the couple temperature was isothermally lowered from 400°C to 25°C, a differential expansion of 0.00278 inch between couple legs would result if the legs were not constrained. At room temperature, the differential expansion is such that the stress on the inner n-type leg is tension while the stress on the p-type cylinder is compression. If the gold yields completely, the compression or the elongation, relative to its axial length, would produce an elongation of gold by about two percent. This result is within the limits of the 4 to 30 percent elongation cited for gold.

However, if the gold were replaced with a material which was not ductile and would not yield to the thermally-induced stresses, an axial stress of 33,360 psi could result. This stress was calculated from the equation

^{1.} Metals Reference Book, 3rd Ed., Vol. 2, 1962, Smithells, Colin J., p.893.

$$\sigma = \frac{\Delta L E_p}{L}$$

where σ is the axial stress in psi, ΔL is the differential expansion (0.00278 inch), L is the length of the p-type leg (1.500 inches) and Ep is the modulus of elasticity of p-type SiGe (1.8 x 10^{0} psi). Such a stress would cause fracture and separation in the weakest couple component which is the n-type telluride segment in tension. (Tensile strength of the tellurides is typically 1500 psi.) Therefore, inasmuch as the initial Hybrid couples have shown successful application of gold as a leg component, it is assumed that the gold sufficiently relaxes the induced stress caused by differential expansion between the couple legs.

(2) The second area of potential mechanical weakness is that region in the p-type SiGe cylinder near the bond with the SiMo hot shoe. Because this bond is normally made at a temperature in excess of the hot shoe temperature, the induced stresses that result from the interaction of both materials increase as the couple temperature is reduced. As a limiting case for maximum stress, the isothermal situation of the couple at room temperature has been investigated.

The nature of these stresses at room temperature would place the SiMo hot shoe in compression along radial and tangential vectors lying in the plane of the hot shoe and emanating from a point on axis with the cylinder (see Figure 13). Conversely, the p-SiGe cylinder contracts more than the SiMo upon cooling to room temperature and therefore results in a hoop stress tension (tangential component) and a shear stress (radial component) directed normal to the cylinder wall. Of the two stress cases, the radially-directed shear stress is of greatest significance and interest, and is discussed further.

Assumptions made for this case include a SiGe cylinder with a mean radius of 0.241 inch and a wall thickness of 0.046 inch bonded to a SiMo heat receptor. The differential radial component due to thermal expansion between the SiMo and cylinder at room temperature was calculated to be 1.30 x 10^{-4} inches. This value was then used in the stress equation

$$\sigma = \frac{\Delta RG}{R}$$

where σ is the shear stress, R is the radius of the SiMo heat receptor inscribed by the cylinder at 25°C, and G is the value for shear modulus. Substituting R = 0.241 inch and G = 7.8 x 10^{6} psi in the equation yields the radial shear stress of 424 psi. In the absence of test data on the shear strength of SiGe material, the calculated

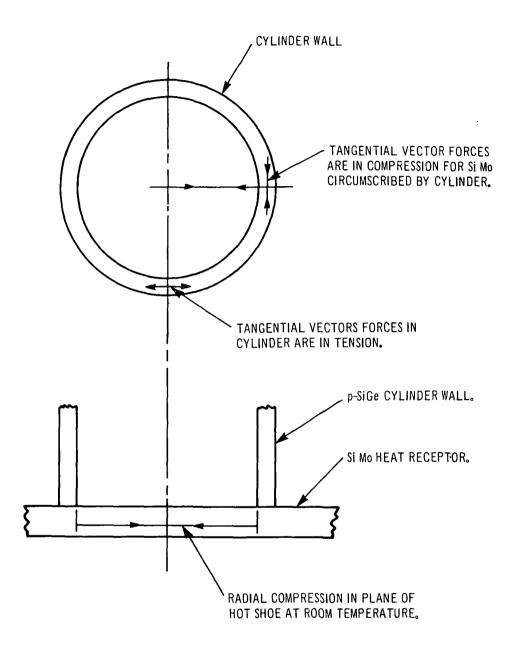


Figure 13. Room Temperature Stresses in Bond Area p-Type SiGe Cylinder to p-Type Si Mo Heat Receptor.

value of 424 psi was compared to tensile test data of p-type SiGe which yields at 3900 psi.

Because the shear modulus is approximately 0.4 times the modulus of elasticity, it was assumed that the shear strength would be 0.4 times the ultimate tensile strength; i.e., the shear yield point would be approximately 1600 psi. Comparison of the calculated shear stress of 424 psi to the ultimate shear strength derived from the tensile test data demonstrated that the calculated shear stress was sufficiently below the shear stress limit and would not fracture the cylinder wall. This conclusion is further strengthened by the fact that couples previously fabricated by RCA of similar materials and geometries have not shown fractures at this bond joint.

In addition to the aforementioned thermal stress cases, a third one which analyzes the capability of the Hybrid couple to withstand lateral acceleration of 6 g's has been considered.

Figure 14 shows the direction of forces, pivot point P, and point Q for which the extreme fiber stress has been calculated. The plane in which points P and Q lie is in the plane of the SiGe-to-tungsten cold shoe bond. From a lateral acceleration point of view, this bond is considered the most likely place for failure to occur since the combination of factors of induced stress to material strength would be least favorable at this point.

Initially the stress (σ) in the extreme fiber at point Q was computed from the equation shown below which gives the stress for the Hybrid couple viewed as a solid cylinder.

$$\sigma = \frac{3a \rho_t \ell^2}{8R} + \frac{6a V_{hr} \rho_{hr} (\ell + \frac{T_{hr}}{2})}{8\pi R^3}$$

where R is the radius of the cylinder, $\rho_{\rm t}$ is the density of SiGe, a is the acceleration, ℓ is the element length, $V_{\rm hr}$ is the volume of the heat receptor, $\rho_{\rm hr}$ is the density of the heat receptor. The values used were:

However, the Hybrid couple is a thin-walled cylinder and provides a reduced bonding area to the tungsten cold shoe. This reduced area has the effect of increasing the value of the unit fiber stress at a point, Q, relative to the stress for the solid cylinder. An approximation of stress increase could be obtained by applying the ratio of bonding area

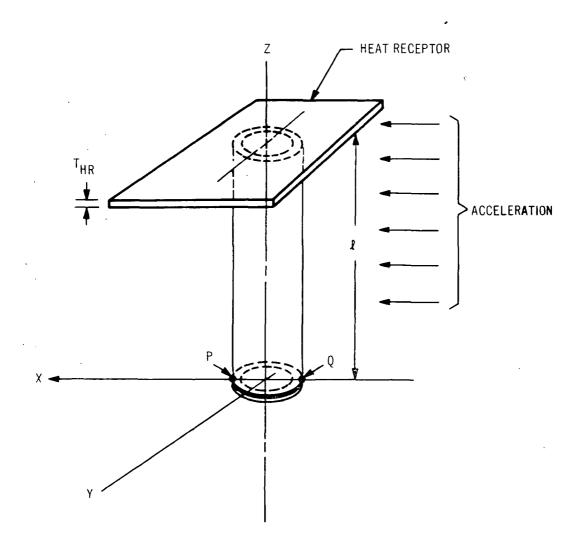


Figure 14. Load Distribution in Direction of Radius

of the solid cylinder to that of the thin-walled cylinder, and to increase the unit fiber stress of the solid cylinder by this factor. This calculation, considering a wall thickness of 0.050 inch, yielded the unit fiber stress of 4 psi at Q.

From a comparison of this stress level to the tensile strength of SiGe (typically 3900 psi for p-type SiGe) which is considered the weakest material in tension in the p-type cylinder leg, the conclusion was drawn that this form of mechanical testing would not cause fractures or separations in the p-type SiGe near this bond.

C. Generator Design

1. Hybrid Planar Converter Designs

A parametric analysis of a 250-watt, 28-volt planar converter employing the Hybrid thermocouple was performed using a computer program. The objective of the program was to define a specific thermocouple geometry and a thermocouple layout configuration that would maximize efficiency for given operating temperatures. In addition to efficiency, specific power also is computed. Other outputs of the program include operating conditions of various converter segments, thermocouple and converter geometries, and converter performance.

The analytical concept of the Hybrid converter and the assumptions made in developing the model can be made clearer by referring to Figure 15. (A converter is defined as the power generating system without the isotope heat source, and a generator as a converter plus the isotope heat source.)

The converter consists of a planar array of Hybrid thermocouples (arranged as nearly square as possible) which are attached to a radiating, finned heat sink with fuel capsules distributed above the hot shoes. Insulation of the multifoil type surrounds the thermocouples and the entire converter. For weight consideration as well as cost, the foil insulation is assumed to be a composite of molybdenum, nickel, and aluminum foils with an opacified paper separator between foils. Initially an effective thermal conductivity of 6.6 x 10⁻⁵ watts/°C-cm was used for the Hybrid thermocouple analysis and for other thermoelectric programs. More recent test data from other programs conducted at RCA, however, suggest that the effective foil insulation thermal conductivity is the same or somewhat higher than the predicted value. Current design programs, therefore, were modified to use the more conservative foil thermal-conductivity value of 1.6 x 10⁻⁴ watt/°C-cm.

The heat balance equations, as in previous analytical models, take into account the thermocouple conduction, Peltier, and Thompson heats. Temperature gradients that normally occur in the hot shoe and radiating fin also have been taken into account. In the case of the fin, it was assumed that all heat to be rejected radiates directly to space. This

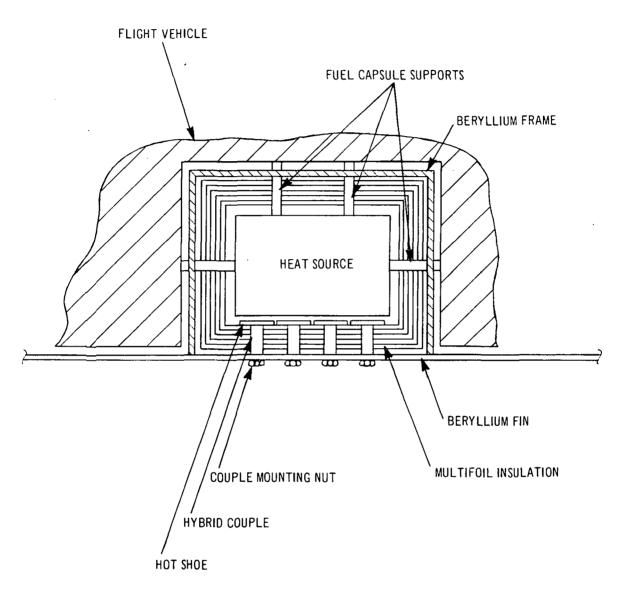


Figure 15. Planar Converter Recessed into Flight Vehicle

assumption has the effect of requiring a slightly larger fin area than actually may be required, because some heat is lost from fin radiation and body conduction to the flight vehicle into which the converter is recessed. Vehicle and deep-space temperatures were taken as 298°K and 73°K, respectively.

Materials comprising the Hybrid couple are those discussed previously for the thermocouple design phase. The use of 63.5 at. % SiGe material data in this analysis, which is consistent with the prior thermocouple analysis, gives a conservative view of converter performance. The thermoelectric SiGe materials used to fabricate the Reference Design Hybrid Couples, as described in Section IV, have a higher silicon content, 80.0 at.% silicon for the p-type cylinder leg and 70.0 at.% for the n-type leg. This value of silicon content enhances mechanical stability at bond interfaces between SiMo hot shoes and element legs. Measurements of material properties on alloys containing 80.0 at.% silicon showed a slight increase in the figure-of-merit relative to the lower silicon content material for the phosphorus-doped n-type materials. (The solubility of phosphorus in the n-type SiGe material increases proportionately with the silicon content of the alloy.) figure-of-merit of the p-type material, on the other hand, remains essentially unaffected because the boron-doping level is not limited by solubility in the matrix. Although the final Hybrid thermocouple design used thermoelectric materials with higher per cent silicon alloys, the resultant changes in thermoelectric properties are small, and the performance predicted by the original analytical model (assuming lower silicon content alloys) is still valid.

Other significant assumptions include the use of beryllium for the fin and protecting converter cover to keep weight to a minimum. The additional weight of mounting hardware and fuel capsule support was compensated by increasing the computed converter weight by 10 per cent. The specific power includes the weight of the capsule or the fuel to be used with the converter, based on 60 thermal watts per pound. The total computed heat required for the converter was increased by 5 per cent to account for the heat-shunt loss through the capsule supports to the flight vehicle.

In summary, the planar converter computer-aided design program, known as HYGEN (see Appendix II, Sections B, C, and D), establishes converter geometries for various inputs of thermocouple geometries and operating temperatures. The more significant outputs of the program, in addition to converter geometries, are the power output, efficiency, number of thermocouples, voltage, specific power, and temperature profiles of various components of the converter.

As in other thermoelectric programs, the weight and specifics of heatsource design were not included in the converter design model. Only tentative assumptions can be made until a heat source can be specified, i.e., allowing sufficient volume space to accommodate the fuel capsule(s). It generally appears feasible to have heat source designs approximately 3.5 inches in diameter and 6 to 12 inches in length. Safety usually is the limiting consideration in defining heat source designs. In the present Hybrid converter design, 3.5 inches plus clearances were allowed normal to the plane of the thermopile hot shoes to accommodate the heat sources. The volume space allowed would be adequate for a fuel loading of about 4000 watts. The required radiant-heat flux from one side of the capsule would be on the order of 2 to 3 watts/cm², which is within the reasonable capability of fuel capsules currently being developed.

Three converter designs were established based on variation of the cold junction temperatures, 122°C(250°F), 177°C(350°F), and 232°C (450°F), respectively, for each of two Hybrid couple designs selected initially in Task II. The preliminary Hybrid Converter Designs are presented in Table IV and the Hybrid Couple Designs in Table V. The agreement between both computer programs, the Hybrid Couple Program (HYBRID) and the Hybrid Converter Program (HYGEN) is very close. This is evidenced by comparing Couple Design B of Table V with Converter Case III-B of Table IV, and Couple Design B-l of Table V with Converter Case VI of Table IV. In addition to describing the couple, the HYGEN program arranges the appropriate number of couples in a square array and computes converter parameters such as specific weight, efficiency, component weights, geometries, electrical performance, etc.

The HYGEN computer program treats the converter model as a square array of Hybrid couples arranged with two series strings of couples electrically connected in parallel for greater reliability. When the couple strings are electrically paralleled, the output voltage and power requirement at end of the specified five-year design life should be 28 volts and 250 watts, respectively. To achieve this, the beginning of life (BOL) power requirement has been increased 10 per cent to 275 watts. This assumes a fuel source power decay of 6 per cent and 4 per cent for the converter. A beryllium fin was used to reject waste heat to deep space (from one side of the converter only). To reject several thousand watts of waste heat, as in the case of this converter, a large fin having an area of about 8500 in.2 (fin effectiveness 0.2) is required. This size fin becomes too heavy and large for a practical generator. The high calculated fin weights (WFIN) and low specific powers (WGPO) indicated for the cases presented in Table IV confirm the assumption made at the beginning of the program that a solid finned radiator might not be suitable for application to this converter.

In general, all cases shown meet or exceed within a per cent the BOL power requirement of 275 watts. It is desirable to minimize the fuel loading in order to reduce fuel cost and fuel source weight. Case B-II of Table IV appears attractive in this respect but the load voltage would be too marginal for BOL consideration. Case B-I is the same couple but operates at a lower cold side temperature. This design results in a voltage of 29.2 volts which is marginal since 30 volts is assumed necessary at BOL. Although Case B-I-IV exceeds 30 volts,

TABLE IV

PRELIMINARY HYBRID CONVERTER DESIGNS

Solid Fin Beryllium Radiator - 0.25" Thick

		Program			Д			H-1	
S	CONVERTER PARAMETERS	Symbol*	Units	I	II	III III	ΛI	IV VI	I
1)) Efficiency**	GETA		7.15	6.79	6.38	6.93	6.57	6.20
2)	Power Output** (BOL)	GPO		312	277	273	335	298	293
3)	Fuel Loading	GQT		4274	7007	4363	4835	4531	4730
(7) Heat Loss through Case	රි ලි රිසප		312	31.2	337	333	333	360
5)	Heat Rejected by Fin	1		3526	3308	3462	3937	3684	3853
9	Number of Couples	GCPL		289	583	324	289	586	324
2) Load Voltage**	GEL		26.5	28.0	29.9	59.6	28.3	30.3
8) Weight of Case	W Case		13.6	13.6	14.7	14.5	14.5	15.7
6	Weight of Fin	W Fin		7.00	230	85	245	265	160
10)) Weight of Foil	W Foil		56.9	56.9	29.5	28.9	28.9	31.4
11)	Weight of Couples	WGCPL		29.1	28.6	32.0	29.8	29.5	32.8
12)	Specific Power**, xx	WGPO	_	.511	.755	1.17	.481	.719	.918
13)	13) Fuel Capsule Temperature	TFCC		1034	1028	1023	1026	1020	1014

xx Fuel Source Weight based on 60 thermal watts per pound

PARAMETERS
COUPLE

	THP	ပ	956	956	926	926	956	956
ure	THN2C	ပ္	538	538	538	200	500	500
perature	TCP	ပ	122	177	232	122	177	232
	ETA	₽€	8.1	7.7	7.3	7.9	7.5	7.0
	邑	volts	.202	.194	.185	.205	.196	.187
	껖	ohms	.0316	.0327	.0337	.0301	.0311	.0322
	ANP	ı	1.6	1.6	1.6	1.3	1.3	1.3
	DTRR	inch	970.	970.	970.	.055	.055	.055
	FLP	inch	1.25	1.25	1.25	1.25	1.25	1.25
	FL1	inch	.658	969.	.734	.706	.746	.784
	FL2	inch	.393	.354	.315	.343	307	.264
	FH	inch	096.	096.	096.	1.01	1.01	1.01
	RRN	inch	.1875	.1875	.1875	.1875	.1875	.1875
4) Couple Elliciency, 7 5) Closed Circuit Voltage** 6) Resistance** 7) An/Ap 8) Wall Thickness 9) Length of p-type Sige 10) Length of n-type Sige 11) Length of n-type PbTe 12) Heat Receptor Dimension 13) Radius of n-PbTe Element		ELA R ANP DTRR FLP FLP FL1 FL2 FH		EL R R ANP DIRR FLP FL2 FL2 RRN	EIA % EL volts R ohms ANP - DTRR inch FLP inch FL1 inch FL2 inch FR inch FR inch FR inch	EL volts .202 R ohms .0316 ANP - 1.6 DTRR inch .046 FLP inch .658 FL inch .658 FR inch .393 FH inch .960 RRN inch .1875	EL volts .202 .194 R ohms .0316 .0327 ANP - 1.6 1.6 DTRR inch .046 .046 FLP inch .658 .696 FL2 inch .960 .960 RRN inch .1875 .1875	EL volts .202 .194 .185 R ohms .0316 .0327 .0337 ANP - 1.6 1.6 1.6 DTRR inch .046 .046 .046 FLP inch 1.25 1.25 FLI inch .658 .696 .734 FL2 inch .393 .354 .315 FH inch .1875 .1875

^{*} See Appendix II, Section D-2
** See explanation at bottom of Table III

TABLE V
PRELIMINARY HYBRID COUPLE DESIGNS

DESIGN PARAMETERS

Operational	<u>Unit</u>	<u>Design B</u>	Design B-1
Thermocouple efficiency*	%	7.35	7.11
Power per thermocouple*	watt	0.867	0.934
Voltage*	volt	0.185	0.187
Current*	amperes	4.69	4.99
Internal resistance*	ohm	0.033	0.031
Hot junction temperature	°C/°F	926/1700	926/1700
Cold junction temperature	°C/°F	232/450	232/450
Interface temperature	°C/°F	538/1000	500/932
Incident heat flux	watts/cm ²	2.0	2.0
Geometrics			
A_n/A_p (ANP)	_	1.6	1.3
$A_n(SiGe/A_n(PbTe) (ANR)$	-	0.7	0.7
Radius of n-type PbTe (RAN)	inch	0.1875	0.1875
Wall thickness (DTRR)	inch	0.046	0.055
Gap between n-p legs (GAP)	inch	0.030	0.030
Length of p-type SiGe (FLP)	inch	1.250	1.250
Length of n-type SiGe (FL1)	inch	0.736	0.788
Length of n-type PbTe (FL2)	inch	0.314	0.262
Heat receptor area (AH)	inch ²	0.914	1.019
Heat receptor dimension (FH)	inch	0.96×0.96	1.01×1.01
Thermocouple OD	inch	0.527	0.545
Thermocouple ID	inch	0.435	0.435
n-type SiGe OD	inch	0.314	0.314
n-type PbTe OD	inch	0.375	0.375

^{*} See explanation at bottom of Table III

the excess power and lower efficiency indicate that this case is far from being optimum. Improved converter performance, however, can be obtained through better matching of converter performance to the required power and voltage at BOL.

From the foregoing converter analyses, it is evident that the solid 0.25-inch-thick beryllium accounts for the majority of the converter weight which results in low specific powers. The converter specific power for the six cases cited ranges from 0.48 to 1.17 watts per pound. To improve the specific power, alternate heat light-weight heat rejection systems were considered.

2. Analysis of Light-Weight Heat Rejection Systems

A preliminary study of a honeycomb-heat pipe radiator was conducted. RCA's Defense Electronics Division made available an existing analytical computer program for determining the minimum weight configura-In all, 756 cases were treated. The analytical program considered the following variables: skin materials of beryllium and aluminum; white-coated surface and second-surface mirrors; water heat pipe diameters of .25 inch (.64 cm), .37 inch (.94 cm), and .50 inch (1.27 cm); and nine evaporator gas temperatures ranging from 68°C (155°F) to 200°C (392°F). In addition, the effect of external energy impinging on the radiator surface is treated by means of an environmental parameter, which is varied over a range from 0 through 0.6. See Figure 16 for the definition of environmental parameter. Radiator weight, radiator area, spacing between adjacent heat pipes, skin thickness, and total length of heat pipe required were outputs of the program. Other assumptions were that the heat pipe was a straight, stainless steel tube, having a .010-inch wall thickness, and operating isothermally along its entire length. The total radiator size and weight were calculated assuming heat rejection from one side of the radiator only.

Results of this analysis indicated that an aluminum honeycomb radiator was, in general, slightly lighter than its beryllium counterpart. Hence, aluminum was chosen as the radiator material. Mechanical and structural constraints, however, might alter this conclusion.

The choice between a white-coated radiating surface or a surface of second-surface mirrors, having a lower absorptance, depends greatly on the radiator heat rejection temperature and the external energy impinging on the radiator surface. These quantities of external energy and radiator surface temperature are related by the environmental parameter (PE) and significantly affect the size and weight of the radiator required to reject the converter waste heat. The environmental parameter is defined as the ratio of the external energy absorbed by the radiator to the total heat which would be radiated by an isothermal radiator at the heat pipe temperature. Figure 16 shows the increase in the environmental parameter as the radiator temperature decreases as a function of several levels of heat flux absorbed by the radiator. In essence, the external impinging heat flux on the radiator should be made as low as possible through orientation of the radiating surface to the flight vehicle and celestial bodies and/or using low absorptance radiating surfaces. Although second-surface mirrors have lower absorptance characteristics than the white-coated surfaces, they do weigh more on a per unit area basis. Hence, a tradeoff exists in selecting a surface finish to obtain minimum weight at a given radiating temperature. For a typical mission with the radiator at 200°C and using second-surface mirrors (absorptance, 0.1 or less. Using a white-surfaced radiator under similar conditions, the environmental parameter would increase typically to a value of 0.3

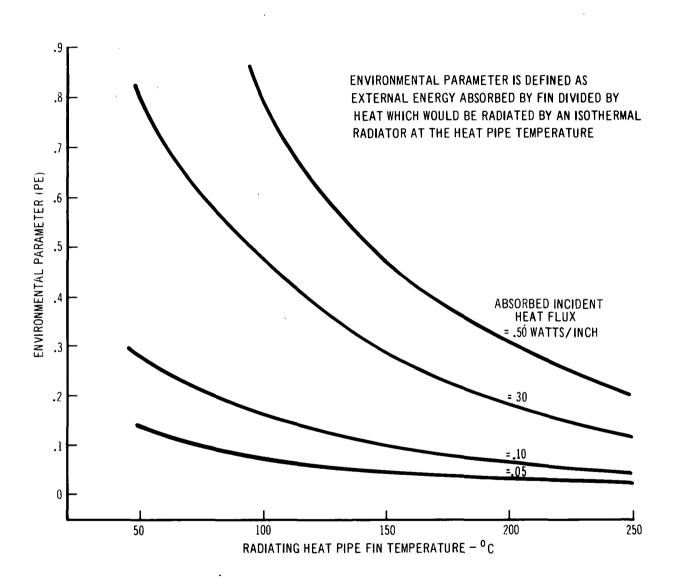


Figure 16. Environmental Parameter (PE) Vs Radiating Heat Pipe Fin Temperature

due to the higher solar absorptance. In these cases the external energy impinging on either surface has been assumed constant. Although the environmental parameter numbers are different, the weights of both type radiators are nearly the same at this temperature, indicating the area of the second-surface mirror radiator is less than the white-coated surface radiator. Further study would be required in this area to assess the effects of mission, radiator orientation, surface finish, and heat rejection temperature on weight.

As an example of the importance of the environmental parameter on radiator weight, a typical case for a 0.37-inch OD heat pipe-honey-comb radiator is presented in Figure 17. This figure indicates the minimum weight condition for the proper combination of adjacent heat pipe spacing (CCS) and skin thickness (T). The total length of a single heat pipe is calculated based on the total heat rejection area required. In actual practice the total heat pipe length would be subdivided into a number of smaller practical size lengths. This would yield a number of pipes which would be spaced adjacent to each other to form a practical, minimum-weight honeycomb heat pipe radiator assembly (see Figure 18).

Another significant variable affecting weight is the size heat pipe used. Figure 19 shows the advantage of using the smallest practical heat pipe diameter to keep radiator weight low. Constraints to using very small heat pipes would involve the practical fabrication limitation of such pipes and the radial heat transfer capability, i.e., temperature drop from evaporator surface of heat pipe to vapor inside. The dashed line in Figure 19 indicates for one case the penalty paid in weight due to the increased temperature drop as the input heat flux increases inversely to pipe diameter. For the assumptions cited, even the small 0.25-inch heat pipe seems acceptable as heat flux or temperature drop is not a constraint. From practical considerations, the size and material for the heat pipe needs further study for optimum heat pipe operation. At present, water is the best working fluid for this temperature range; however, its compatibility with stainless steel is questionable. Consequently, pipes of copper or copper-lined stainless should be studied further for determination of reliability and minimum weight. Copper heat pipes using water have been operated by RCA at 150°C (302°F) and have shown good performance after 15,000 hours of life with negligible degradation. Extending equivalent performance to 200°C (392°F) using copper in contact with water appears quite feasible.

Based on the above analysis, a typical honeycomb-heat pipe radiator can now be nominally characterized in terms of weight and interfaced with one of the present converter designs. Figure 18 shows a typical radiator that would be used with converter design B-III. To keep the temperature drop as low as possible, an arrangement of 16 0.37-inch diameter stainless steel heat pipes joined by a honeycomb structure and aluminum skin has been used. The tubes have been fanned out to meet the spacing requirements between pipes for minimum weight.

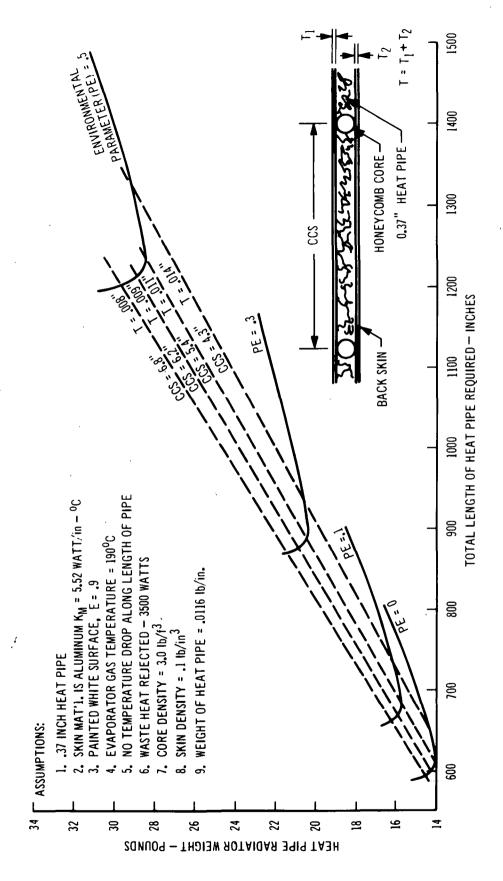
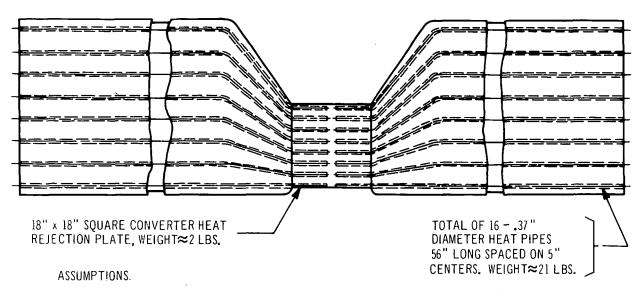


Figure 17. Heat Pipe Weight Vs Total Pipe Length for Several Environmental Parameter Ratics



- 1. WHITE SURFACE RADIATOR
- 2. ALUMINUM SKIN≈.012"THICK
- 3. ENVIRONMENTAL PARAMETER = .3
- 4. WASTE HEAT REJECTED FROM ONE SIDE OF RADIATOR ONLY = 3500 WATTS
- 5. EVAPORATOR GAS TEMPERATURE = 190°C
- 6. UNIFORM TEMPERATURE MAINTAINED ALONG PIPE LENGTH

Figure 18. Typical Heat Pipe Radiator for Converter Design, B - III

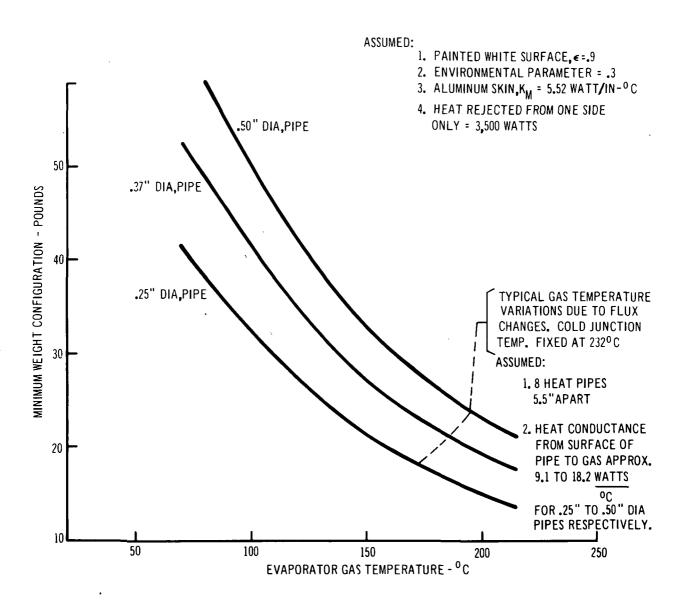


Figure 19. Weight Vs. Evaporator Gas Temperature OC for Three Heat Pipe Sizes

Approximately 896 inches of total length of heat pipe was indicated from the computer analysis. Subdividing this by 16 permits a reasonable individual heat pipe length of 56 inches. This is considered practical for heat pipe design and fabrication. (RCA has fabricated and operated 0.25 0D inch pipes, 48 inches long, and 0.50 0D inch pipes, 96 inches long.) In practice, however, temperature drops along the length of the pipe, while small, plus some performance degradation due to bending of the heat pipes, have not been taken into account in the present analysis. These effects are real and their significance should be investigated further.

3. Selection of Converter Reference Designs

As a result of the development concluded in Task II, Thermocouple Development and Testing, Design B (see Table V) was selected as the Reference Design Hybrid Couple. Therefore, the three preliminary converter designs selected were based on the Hybrid Reference Design Couple (Design B) and the use of the honeycomb-heat pipe radiator system (see Table VI).

TABLE VI SELECTED GENERATOR DESIGNS

		<u>B-I</u>	<u>B-II</u>	<u>B-III</u>
Power output (BOL)	watts	312	277	273
Efficiency	%	7.2	6.8	6.4
Hot junction temperature	°C	926	126	926
Cold junction temperature	°C	122	177	232
Converter weight	lbs	70	69	76
Heat pipe radiator weight	lbs	53	34	23
Fuel source weight*	lbs	71	68	73
Total generator weight	lbs į	194	171	172
Specific power	watts/lb	1.60	1.62	1.59

^{*} Fuel source weight based on 60 thermal watts per pound

The above generator designs indicate that the specific power remains essentially constant for the specified range of cold junction temperatures when using the heat pipe radiator; therefore, the lower cold side temperatures become attractive only in terms of efficiency.

4. Projected Improvements in Hybrid Planar Converter Designs

Further improvement in the performance of both the Hybrid couple and the Reference Generator Design could possibly be obtained by several modifications as noted in the following:

- a. Substituting improved thermoelectric materials:
 - (1) 80 at.% SiGe for 63.5 at.% SiGe.
 - (2) RCA n-PbTe--GeTe Alloy* for 3M Co. 3N Alloy.
- Increasing T_c to 1000° C. Further optimization of converter weight.

The potential improvement in converter performance, based on the use of the improved thermoelectric alloys, is presented in Table VII for the three selected generator designs. The material properties of the 80 at. \$ SiGe and RCA n-type lead telluride alloy have been listed in a new data table, HDATAl (see Appendix II, Section C-2).

TABLE VII

PROJECTED HYBRID COUPLE AND CONVERTER EFFICIENCIES 63.5 At. SiGe AND 3M PbTe COMPARED TO RCA 80 At.% Sige AND n-PbTe THERMOELECTRIC MATERIALS

	•	Thermoelectri	c Materials
	Converter Design B	63.5 at.% SiGe/ <u>3M PbTe</u>	80 at.% SiGe/ RCA n-PbTe
I.	Couple Efficiency, % Converter Efficiency, %	8.1 7.2	9.6 8.5
II.	Couple Efficiency, % Converter Efficiency, %	7.7 6.8	9.1 8.0
III.	Couple Efficiency, % Converter Efficiency, %	7.4 6.4	8.5 7.5

Increasing the hot junction temperature to 1000°C results in an increase of converter efficiency of 10%. The effect on the three selected generator designs is given as follows.

^{* &}quot;Improved Compatible Material for Thermoelectric Power, " Final Report, August 1969; J.P. Dismukes, I. Kudman, and H.I. Moss. TID-4500, Category UC-33; Contract AT(30-1)-3886.

Hot Shoe Temperature*

	Converter Design B		926° C	1000°C
I.	Couple Efficiency, % Converter Efficiency,	%	8.1 7.2	8.9 7.9
II.	Couple Efficiency, % Converter Efficiency,	%	7.7 6.8	8.5 7.6
III.	Couple Efficiency, % Converter Efficiency,	%	7.3 6.4	8.1 7.1
			*T _a = 232°C	

A review of the converter weight has indicated that by (1) simplifying the cold stack, (2) optimizing the multi-foil insulation structure, and (3) improving fuel capsule supports, the present estimate of 76 pounds might be reduced to 56 pounds. This would increase the specific power for Converter Design B-III from 1.59 to 1.80 watts/pound. Moreover, by increasing the hot junction temperature to 1000°C, the specific power could be increased to 2.1 watts/pound and, by substituting the improved thermoelectric alloys, the specific power is increased to 2.4 watts/pound.

A summary of the effect of these improvements on Converter Design B-III is presented in Table VIII. These improvements represent the possible growth potential of the Hybrid Couple concept.

D. Module Reference Design

1. Module Design

The B-III Generator Design requires that 324 Hybrid couples in an 18inch by 18-inch planar array be electrically connected in a seriesparallel circuit. This requirement could be met in two different ways. First, each series string could consist of 162 couples with paralleling electrical connections between both series strings made at the terminals for series-parallel operation. The second is similar to the first except that electrical connections are provided between equal potential points of each couple string. With this method, should a couple degrade during life, there exists an alternate current path which permits continued generator performance with a decrease of only 0.5 per cent in total power. In the first method, should the couple degradation go to the limit of being an open, the generator power would be halved. Although the second method of paralleling is more complex and causes greater generator weight due to the many interconnections than the first, it is recommended because of its greater reliability over the first.

TABLE VIII

COMPARISON OF PRESENT AND PROJECTED B-III CONVERTER DESIGN

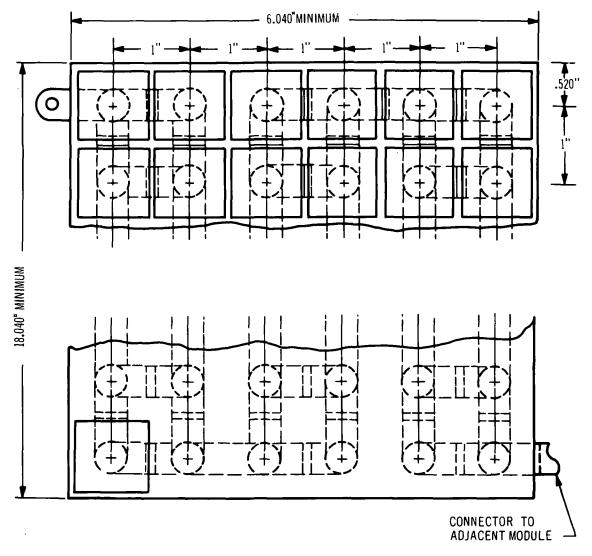
	Converte	Converter Modification			Converter	ter
Sige Alloy	Pbre Alloy	$^{\mathrm{T_h}}_{\circ_{\mathbb{C}}(^{\circ_{\mathbf{F}}})}$	$\frac{\mathrm{T_{\mathbf{c}}}}{\mathrm{O}\mathrm{G}\left(\mathrm{^{O}F}\right)}$	Weight lbs (kg)	Specific Power W/lb (W/kg)	Efficiency %
63.5 at.% S1	3M - 3N	926 (1700)	232 (450)	172 (78)	1.59 (3.50)	7.9
63.5 at.% Si	3M - 3N	926 (1700)	232 (450)	154 (70)	1.80 (3.96)	7.9
63.5 at.% Si	3M - 3N	1000 (1832)	232 (450)	154 (70)	2.10 (4.64)	7.2
80.0 at.% Si	RCA - Pbre-Gere	1000 (1832)	232 (450)	154 (70)	2.40 (5.28)	8.4

A 6 x 18 array of Hybrid couples was assumed for the module, as shown in Figure 20. It requires three module sections of 108 couples each to meet the requirements of the generator. Using more modules of fewer couples each would require more mounting hardware and cause greater difficulty in integrating the numerous modules into a converter. Total gap area between adjacent modules would also increase, thus creating greater heat shunt losses. In the other extreme, the converter could be built as a single unit; however, should replacement of a couple be desired after converter assembly, this would necessitate rebuilding and handling of the entire converter, which would detract from its initial reliability. The module size shown in Figure 20 is seen as the best compromise of the above considerations and was selected as the Module Reference Design.

Other important considerations regarding module design are the processes used to insert the thermal insulation and the technique used to join the electrical interconnections. With regard to the fibroustype insulation, this would be machined as shown in Figure 21. The insulation would be dovetailed to the adjacent piece in order to decrease any line of sight radiation through any voids that might occur at the interface of adjacent strips of insulation. Alternately, a line of 18 couples and strips of insulation would be affixed to a mounting fixture. The mounting fixture would be similar to the mounting plate but have additional holes in it to permit a special crimping tool to cold-weld the copper electrical interconnectors together. No significant difficulties are seen in this area but development is required since the standard connection technique relies on the compression of adjacent connectors using a nut and screw.

Figure 23 also shows this type electrical connector. Typical of past connector designs used on previous programs at RCA, the linking electrical connector relieves the stresses between adjacent couples through the use of a small bend or "omega" in the connector. In addition to the geometry of the connector, relieving stress build-up within the module or converter, the connector itself, being copper, has low inherent stress because the final processing step of the couple anneals the copper and leaves it soft. Consequently, the stress developed between adjacent couples is not considered significant in view of the geometry and softness of the copper connector and the cantilevered air-vac couple construction.

Ultimate construction of the module panel could be simplified by development of a "plug through" Hybrid couple similar to the present air-vac construction (see Figure 22) which would permit a solid piece of insulation with appropriate holes to receive the couples. The electrical interconnection would then be crimped as previously described. With this approach, either fibrous-type insulation or multifoil insulation could be used.



NOTES:

- 1. THIS SHOWS A 6x18 COUPLE ARRAY WHICH IS ONE OF 3 MODULES USED IN THE CONVERTER B III DESIGN REQUIRING 324 COUPLES.
- 2. COUPLE INTERCONNECTIONS PROVIDE A SERIES PARALLEL ELECTRICAL CIRCUIT.
- 3. COUPLES ARE ARRANGED AS COUPLETS PROVIDING MAXIMUM RELIABILITY FOR SERIES PARALLEL ELECTRICAL CIRCUIT.

Figure 20. Module for Hybrid Converter Design, B - III

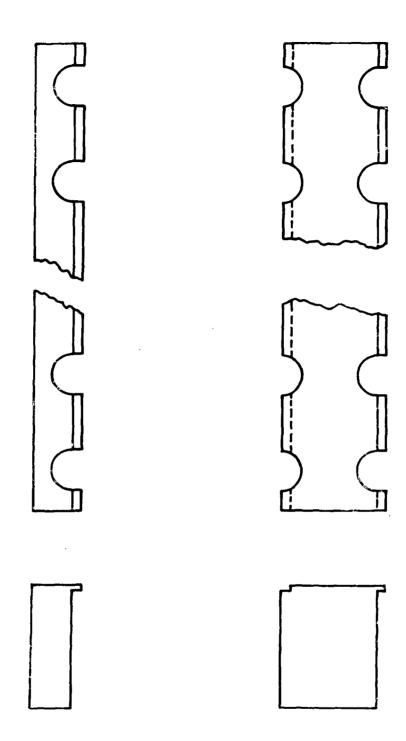


Figure 21. Module Strip Insulation

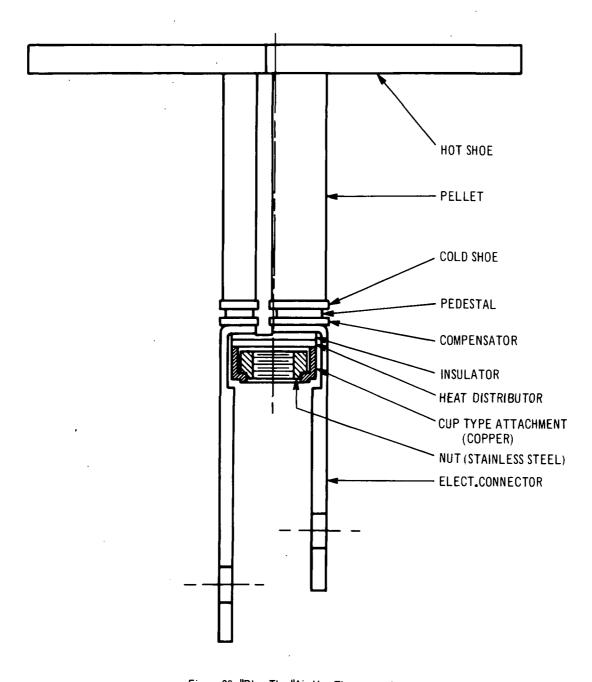


Figure 22. "Plug-Thru"Air-Vac Thermocouple

2. Test Panels

Of the two types of radiator panels considered for the Hybrid couple panel sections, it was decided to use the mounting and cooling plate assembly Design No. 2 (Figure 23). Nine thermocouples would be attached to the mounting plate to form the module section and a water-cooled or finned auxiliary heat exchanger would be subsequently bolted onto the mounting plate during testing for control of the cold junction temperature. This approach provides more flexibility as the water cooling or fin are first attached to the "cooling plate" and the cooling plate is then bolted onto the module section. The expected test panel operating conditions and projected performance are given in Table IX.

TABLE IX
PROJECTED TEST PANEL PERFORMANCE

Operation	n Temperature Conditions	°C_	_°F
(THRC) (THPC) (THNC) (TCNC) (THN2C) (TCPC) (TCN2C)	p-SiMo hot shoe edge temperature p-SiGe leg hot junction temperature n-SiGe leg hot junction temperature n-SiGe leg cold junction temperature n-PbTe leg cold junction temperature p-SiGe leg cold junction temperature n-PbTe leg cold junction temperature	941 926 923.5 541 538 232 233.85	1725 1700 1694 1006 1000 450 452
(TCBC)	Heat rejection base plate temperature	223.3	434

Thermoelectric Performance Characteristics

		<u>Units</u>	<u>Couple</u>	<u>Panel</u>
(ETA) (PO)	Thermocouple Efficiency	%	7.35	7.35
(EL)	Power per Thermocouple Closed Circuit Voltage	watt volt	.77 .166	6.93 1.5
(R)	Internal Resistance	ohm	.030	0.30
(CUR)	Current	amperes	4.64	4.64

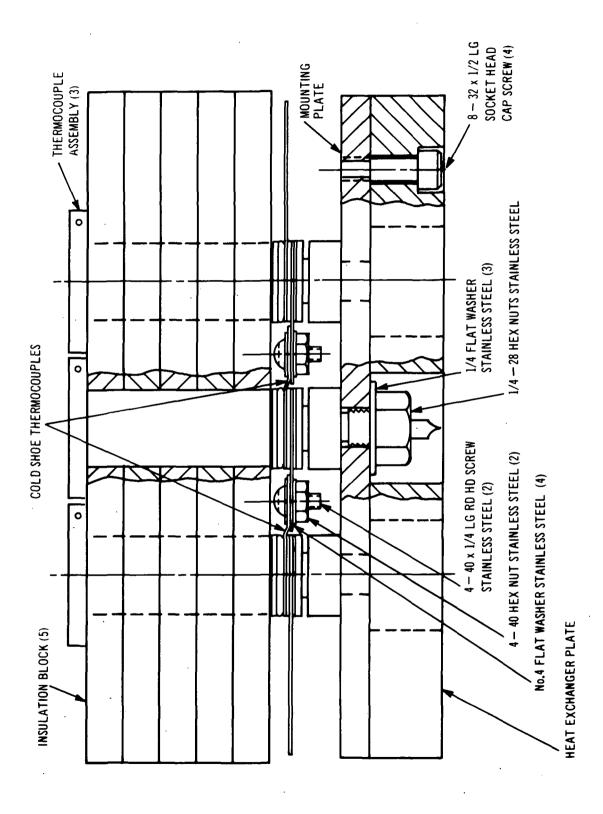


Figure 23. Hybrid Structure Module Panel Assembly

Section IV

THERMOCOUPLE DEVELOPMENT AND TESTING

A. Subcomponent Development

The objective of this phase of the program was to evaluate bond systems and further define fabrication constraints for the subcomponents to be used in the construction of the three Hybrid couple designs (Table III). The following subcomponents were evaluated (see Figure 24):

- (a) <u>p-SiGe element</u> metallurgically bonded to a p-SiMo heat receptor plate and a tungsten cold shoe.
- (b) <u>n-SiGe element</u> metallurgically bonded to a n-SiMo heat receptor plate and a tungsten cold shoe.
- (c) <u>n-PbTe element</u> (3N PbTe) metallurgically bonded to hot and cold stainless steel shoes.
- (d) <u>Cold stack</u>; electrical connectors, insulators, compensators, mount stud and a copper exhaust tubulation brazed together.
- (e) <u>Intermediate bond system</u>; consists of a gold disc bonded between a stainless steel PbTe hot shoe and a SiGe tungsten cold shoe.

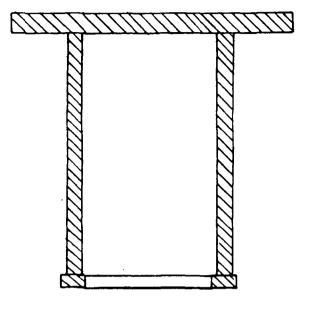
1. p-SiGe Element

a. p-SiGe Cylinder Fabrication

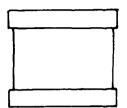
With the initiation of this program, studies were made to determine the practical range of cylinder lengths and wall thickness that could be fabricated by the Elox (spark dischrage) method, and to consider other possible alternate methods if required.

SiGe pellets ranging in alloy composition from 63.5 at.% Si to 80.0 at.% and lengths from 1.42 cm (0.560"), 1.9 cm (0.750") and 2.54 cm (1.0") were drilled by the Elox technique; in each case, mechanically strong, leak-tight cylinders resulted when machined at the slower drilling rates. At faster drilling rates, failures occurred in the form of microcracks and/or complete fracture of the cylinder wall. The leak checks were made using (1) a Vecco Model MS-9 mass spectrometer with helium and (2) a penetrating dye placed inside the cylinder which was temporarily sealed at one end. The outside of the cylinder was coated with a carbonate (white) material. The exact location of leaks could then be determined by direct observation of dye marks.

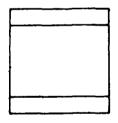
A. p-SiGe ELEMENT



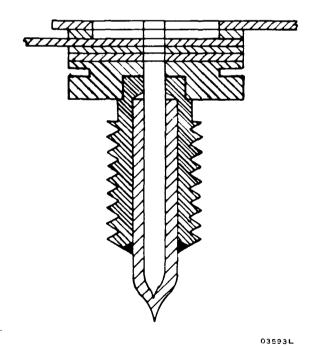
B. n-SiGe ELEMENT



C, n-PbTe ELEMENT



D. COLD STACK



E. INTERMEDIATE BOND SYSTEM

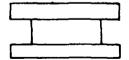


Figure 24. Subcomponent Assemblies

With the definition of the developmental Hybrid couple geometries (Table III), cylinder lengths and diameters indicated were, in some cases, larger than any of the cylinders fabricated to date. Many problems were encountered in attempting to fabricate the larger cylinders by Elox drilling in one operation with lengths up to 3.81 cm (1.5"); namely, cracking and non-uniform wall thickness. Because of the long cylinder length, the machining was done in two stages; first, the blank was drilled half-way though; then the blank was reversed and the remaining half drilled. Although some of the problems were relieved through the redesign of the holding fixtures, they were not fully eliminated.

As a result of the difficulty in obtaining the longer length cylinders, several alternate methods were investigated, including ultrasonic machining, hot pressing, isostatic pressing and sintering, and diamond core drilling. The ultrasonic drilling technique was not feasible, as a non-uniform wall thickness was obtained and excessive cracking of the SiGe wall occurred. Several attempts were made at hot pressing the cylinders. A special graphite die was fabricated and several runs were made, all of which were only partially successful because of equipment problems. The pressure could not be maintained at temperature, and resulted in porous cylinders of non-uniform density. Initial attempts to fabricate cylinders by isostatic cold pressing and sintering also resulted in cylinders with high porosity, 30-40%. Because of the difficulties encountered in the above three alternate approaches, they were discontinued.

The machining of the cylinders was then attempted by an outside vendor using a special diamond core drilling technique. The first cylinders fabricated were 2.54 cm (1.0") in length and 1.13 cm (0.444") in diameter with a 0.101 cm (0.040") wall thickness. Leak checks with the Veeco MS-9 leak detector and dye penetrant showed no leaks, with the exception of where defects (micro cracks) in the starting material were present. Visual inspection of these parts showed a smoother surface than obtained by the Elox technique. The dimensional tolerances were also found to be better. The initial success with the diamond core drilling prompted a more intensified study comparing the two techniques. The evaluation of the two techniques showed the diamond core drilling technique produced higher quality cylinders with better yields than the Elox electrical discharge machining process. It also was capable of producing good cylinders with lengths of 3.81 cm (1.5"). It was for these reasons the diamond core drilling technique was chosen for the fabrication of the cylinders. A summary of fabrication cylinder results is given in Table X. The cylinder produced by diamond core drilling is shown in. Figure 25.

TABLE X

RESULTS OF p-TYPE SIGe CYLINDER FABRICATION

	Disposition	Used for bonding tests	Remaining 4 used for bond test	Remaining used for bond tests	1 Met Lab; 2 bond test	Pinhole leaks	Circumferential cracks		8 bonded	All bonded	To be bonded	1 bonded	Couple bonding
Test	Dye (2)	All passed	l leak	All passed 1 leak All passed	All passed	All leak	All leak		2 leak	All passed	All passed	All passed	All passed
Leak	Veeco(1)	All passed	l leak	All passed 1 leak All passed	All passed	All leak	All leak	17 leak	2 leak	All passed	All passed	All passed	All passed
	Appearance	Good	One broken circumferential	Good Good Good	Hot-pressed pellets - good	3 broken	Good		Good	Good	Good	Good	Good
	Fabrication Method	Elox	Elox	Elox Elox Elox	Elox	Elox	Elox		Diamond-core drilling				
Max. Wall Thickness	(mils)	50	50	70 60 70 70	50	45	45		50	50	50	50	90
00	(inch)	0.444	0.444	0.444 0.444 0.444	0.384	0.384	0.527		0.500	0.500	0.527(3)	0.500	1.25 0.527(3)
	(1nch)	6.0	0.75	0.75	0.5	1.5	1.25	ק	1.5	1.25	1.25	1.25	
Batch		₩	9	ω	3	60	60	Subtotal	10	60	11	9) 168
	Batch	п	N	6	4	2	9		7	€0	6	10(4)	11(4)

NOTES:

Veeco leak detector, Model MS-9
 RCA dye penetrant test
 Machined from large-size ingot (0.600-inch OD)
 80 at.\$ Side alloy

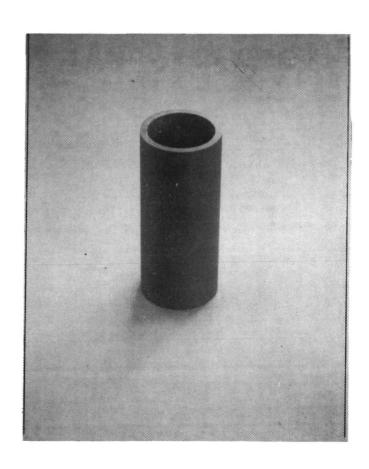
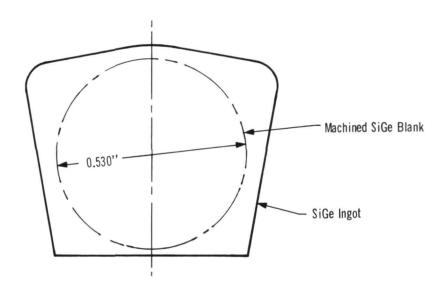


Figure 25. Diamond Core Drilled p-SiGe Cylinder

The cylinders had all been machined from the standard size SiGe zone-leveled "loaf-shaped" ingot which has a maximum diameter of 1.35 cm (0.530 in.) (see sketch below).



In order to provide a larger size SiGe blank for the cylinders, the SiGe zone leveling process was modified. The largest size ingot that could be fabricated on the available equipment had an OD of 1.52 cm (0.600 in.).

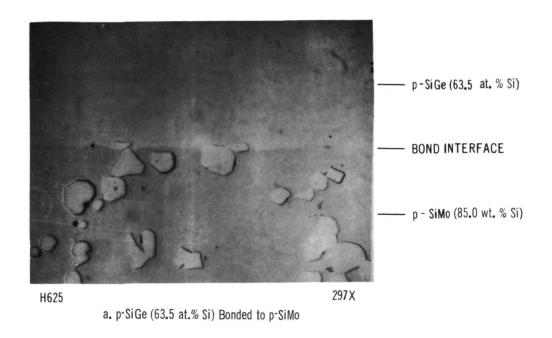
Attempts to fabricate cylinders of couple design A (Table III), 3.81-cm (1.5-in.) length and 0.091-cm (0.036-in.) wall thickness, were unsuccessful due principally to radial cracking. Cylinders of couple design C (Table III), 2.54-cm (1.0-in.) length and 1.58-cm (0.622-in.) OD, were not possible to fabricate since there was insufficient material in the largest size ingot, 1.52-cm (0.600-in.) OD, that could be fabricated in presently available equipment. As a result, all testing of p-SiGe element subcomponents was conducted with couple design B cylinders (Table III), i.e., 3.17-cm (1.25-in.) length, 1.34-cm (0.527-in.) OD, and 0.117-cm (0.046-in.) wall thickness.

b. p-SiGe Cylinder--p-SiMo Heat Receptor

The p-SiGe cylinder must be joined to the p-SiMo heat receptor so as to achieve a sound metallurgical bond with a hermetic seal. Two designs were initially considered; the first employed a conventional butt joint in which the p-SiGe cylinder is bonded directly to the

flat surface of the p-SiMo heat receptor; while the second employed a compression joint, in which a shallow cut (5 to 25 mils deep) was made into the SiMo heat receptor to accept the SiGe cylinder, thus putting the bond in compression. Leak-tight cylinders of 63.5 at.% Si composition were bonded to SiMo (85 wt. Si) heat receptors, using the standard "butt-type" joint technique. Cylinder lengths ranged from 1.42 cm (0.560 in.) to 3.18 cm (1.250 in.) with a wall thickness of 0.127 cm (0.050 in.). Similar results were obtained with cylinders of 80 at. % Si composition. Figures 26a and 26b show these to be metallurgically sound. Bonds made with the compression joint were all unsuccessful. Depression cuts of 5 to 10 and 25 mils were made in the p-SiMo with cylinders inserted into the depression and bonded per the normal practice. In each case, the p-SiGe cylinders cracked in an area 10 to 50 mils above the top lip of the depression cut. The butt-type bond was therefore selected for the bonding of the p-element subcomponent.

Normal bonding practice calls for close matches in thermal expansion among materials bonded at high temperatures unless the seal area is very small. Experimental evidence accumulated during the initial stages of this program showed that consistent leak-tight seals were obtained in the bonding of 63.5 at.% SiGe cylinders to the 85 wt.% SiMo heat receptors at temperatures in excess of 1093°C (2000°F). Based upon these results it was expected that the thermal expansion match would be close enough to permit operation of the device at design temperature. Life testing of the p-SiGe element was conducted at accelerated test conditions, 982°C (1800°F) for 300 hours and at 1090°C (2000°F) for four hours, and thermal cycling tests from 982°C (1800°F) to 150°C (300°F) to assess the mechanical and thermal capability of the p-SiGe to p-SiMo bond. Due to equipment limitations, the accelerated thermal cycle test was more severe than the test defined in the Hybrid Thermocouple Development Program Plan. A comparison of both thermal cycles is shown in Figure 27. No changes in electrical resistance or microstructure were noted in the life-tested samples; however, microcracks developed in the bond region resulting in loss of hermeticity in those specimens subjected to both high temperature soaking and thermal cycling. It was evident that a closer thermal expansion match was required. Figure 28 is a plot of the thermal expansions of various SiGe and SiMo alloys as a function of temperature. These data indicate that the SiGe alloy composition required to match SiMo is in the range of 80 at. # Si. Because the thermoelectric properties of the two p-type SiGe alloys (i.e., 63 and 80 at. \$\mathbf{1}\$) are essentially the same, one alloy can be substituted for the other without requiring a change in design calculations. This new alloy-bond combination was successfully subjected to ten accelerated (see Figure 27) thermal cycles from 982°C (1800°F) to 150°C (300°F). The bonds were leak-tight as evidenced by helium leak check and the dye penetrant tests. The tensile strength of as-fabricated and life-tested specimens (982°C (1800°F) for 300 hours and 1090°C (2000°F) for four hours) averaged 3300 psi and 2550 psi, respectively. These values are typical for SiGe alloys with the fractures occurring in the bulk SiGe.



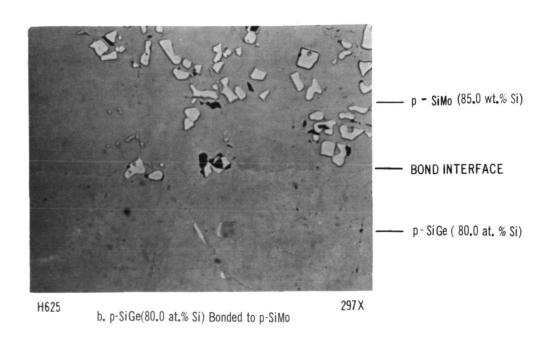


Figure 26
p—SiGe Cylinder Bonded to p—SiMo Heat Receptor

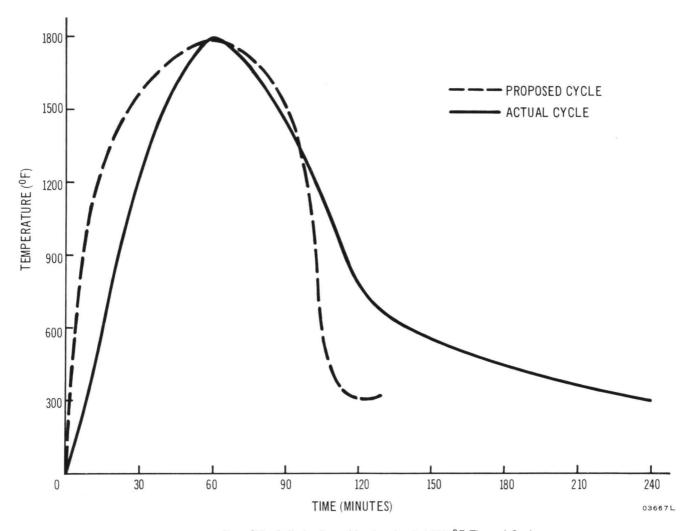


Figure 27. p-Type SiGe Cylinder Assembly, Accelerated 2000°F Thermal Cycle

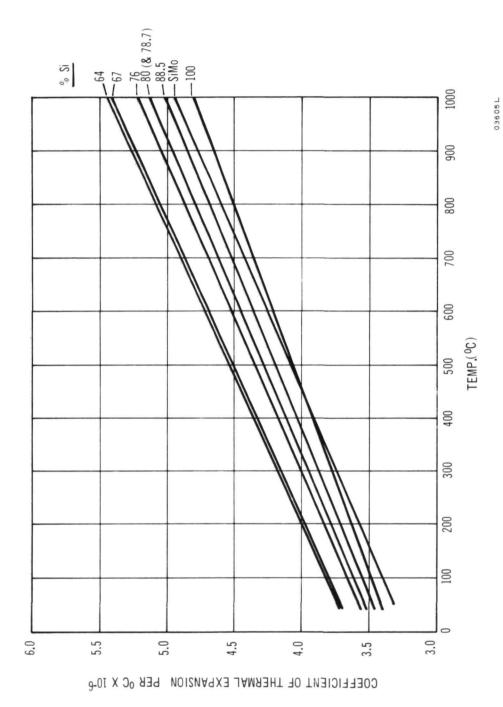


Figure 28. Coefficient of Thermal Expansion as Function of Temperature for SiGe and SiMo Alloys

The bonding of the tungsten cold shoe ring to the p-SiGe (63.5 at.% Si) cylinder initially involved a direct eutectic-type diffusion bond, which forms a WSi2 intermediate layer. Tests conducted with this system did not result in hermetic seals; therefore, development of an alternate attachment technique was undertaken. A low temperature braze technique was developed in which a nickel coating was applied to the SiGe surface via an electroless plating technique. This braze system produced hermetic seals at the p-SiGe element cold shoe interface.

Thermal cycling of subcomponents with the tungsten shoe brazed at the cold end resulted in leak-tight seals after ten cycles from 316°C (600°F) to 93°C (200°F).

In subsequent development of the p-SiGe--p-SiMo subcomponent, the SiGe alloy was changed from 63.5 at.% Si to 80 at.% Si. In order to obtain improved strength, a return to the eutectic-type diffusion bond (WSi2) at the cold end was made. The resultant bonds were metallurgically sound and formed hermetic seals (see Figure 29), both as-fabricated and after thermal cycling tests. The tensile strength of the as-fabricated and life-tested specimens, 316°C (600°F) for 4032 hours, averaged 2300 psi with the break occurring in the bulk SiGe. The cold shoe bond was tested for 6200 hours at 316°C (600°F) with no observable change in bond resistance. This type bond had been previously used in the SNAP 10 thermoelectric modules and has been tested in excess of five years at 500°C (938°F) with stable electrical characteristics.

As a result of these subcomponent tests, the composition and bonds for this segment of the Hybrid couple were selected as follows: an 85 wt.% p-SiMo heat receptor bonded directly to an 80 at.% p-SiGe cylinder leg which is simultaneously bonded at the cold end to a tungsten shoe.

2. n-SiGe Element

The n-SiGe element consists of a n-SiGe pellet metallurgically bonded to a n-SiMo hot shoe on one end and a tungsten cold shoe on the other end; see Figures 30a and 30b. These metallurgical bonds have been used extensively in the past and have exhibited good stability and life capability at the Hybrid couple design operating temperatures.

A limited number of subcomponents were fabricated and subjected to various screening tests. Life tests at 982°C (1800°F) for 300 hours and 1090°C (2000°F) for four hours showed no change in electrical or physical properties. The tungsten-n-SiGe bond was not included in these test samples due to the temperature limitation of the WSi2 bond. Metallographic examination indicated normal n-SiGe--n-SiMo bonds with no discernible change from initial bonds (see Figures 30a and 30b).

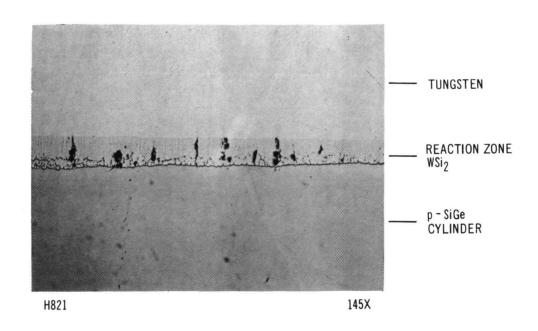


Figure 29. p-SiGe (80.0 at. % Si) Bonded to Tungsten Cold Shoe

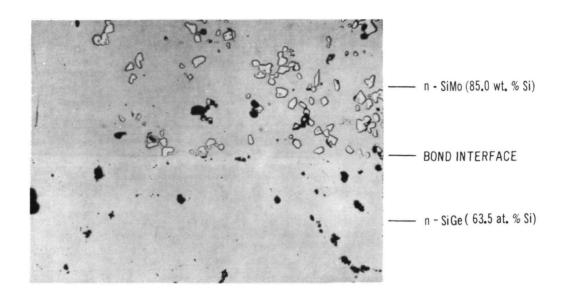


Figure 30a. Hot Shoe Bond n - SiGe Leg to n - SiMo Hot Shoe

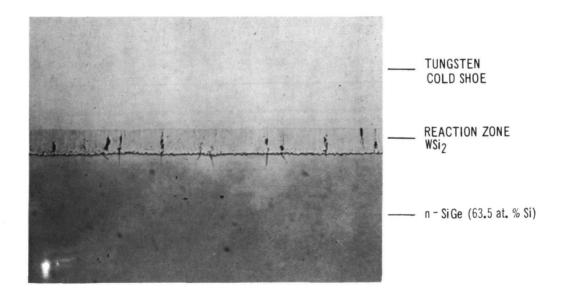


Figure 30b. Cold Shoe Bond n - SiGe Leg to Tungsten Cold Shoe

A summary of the tensile and shear strength tests made on as-fabricated and life-tested specimens is given in Table XI. These strength values are normal for SiGe with the fractures occurring in the bulk SiGe material.

TABLE XI

TENSILE AND SHEAR STRENGTH OF n-SiGe ELEMENTS

		Tensile Strength*, psi		Shear Strength*, psi		
	Number of		After 300 hours @ 982°C	After 4 hours @ 1090°C		After 300 hours @ 982°C
Design	Specimens	As-bonded	(1800°F)	(2000°F)	As-bonded	(1800°F)
A	15	3600-4000	3500-3800	3500-3800	2000-2500	2000–2500
В	15	3600-4000	3700-4000	3700-4000	2300–2500	2000-2500
C	15	3600-4000	3700-4000	3700-4000	2300-2600	2000-2500

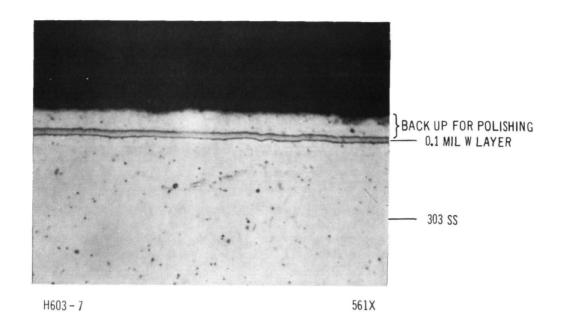
^{*} Measured at room temperature

3. n-PbTe Element

The n-PbTe element consists of a 3N PbTe pellet metallurgically bonded to 303 stainless steel shoes, which have been treated with a tungsten diffusion barrier layer and a minimum of bonding medium layer.

The shoe is an electrical contact which is chosen for its thermal expansion match with the PbTe element. In this application, it is supplemented with a chemically stable tungsten barrier. Therefore, the chemical purity of the PbTe is maintained to permit its normal life without the poisoning effects of foreign elements.

The tungsten is applied by chemical vapor deposition using commercial tungsten hexafluoride. A plating thickness of 0.1 to 0.5 mil is adopted for standard practice; Figure 31. These tolerances are maintained by sampling one of each batch of five shoes (Figure 32). The barrier is restricted to the actual shoe bond area by the die. This thin tungsten barrier should maintain good adherence and soundness throughout the life expectancy of the PbTe element.



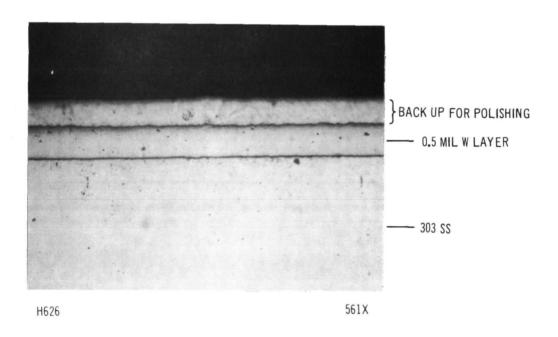


Figure 31. Thickness Limits of Chemical Vapor Plated Tungsten on 303 Stainless Steel Shoes

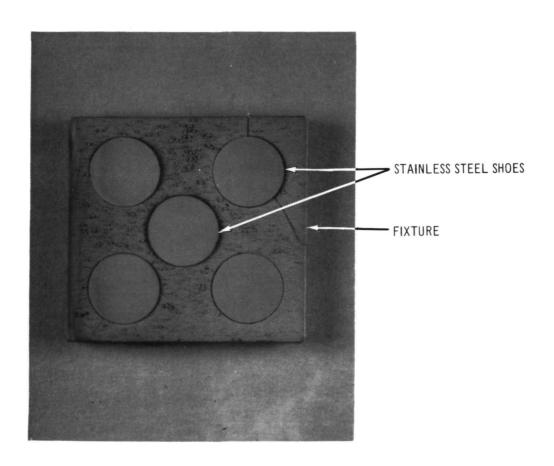


Figure 32. Tungsten Vapor Plated 303 SS Shoes One Run of Five Shoes .375D X .040

The shoe is bonded to the PbTe pellet with a procedure which forms a complete void-free interface and one free of surface oxides developed on the surface of the pellet. This solves many of the recognized problems with a W-PbTe bond which includes (1) poor wetting characteristics of the PbTe surface, (2) rapid surface oxide formation on a mechanically cleaned surface, and (3) negligible solubility of PbTe into tungsten for bonding purposes.

The procedure uses nickel as a bonding medium and a die to provide alignment and mechanical pressure (Figure 33). In the bonding cycle, with an inert atmosphere, the nickel diffuses into the tungsten and dissolves a thin surface layer of PbTe to form the bond. With the die and pressure rods fixing the position of the shoes relative to the pellet, additional pressure is applied to eject the excess melt and, along with it, any residual oxides and foreign matter. The bond overflow is contained in the immediate area of the bond, as shown in Figure 34, and discarded in the finishing operation. What is left is a perfectly sound, clean and void-free bond. Appropriate photomicrographs are presented in Figures 35 and 36.

The tensile and shear strength of as-fabricated PbTe elements of 0.635 cm (0.250 in.) and 0.951 cm (0.375 in.) OD are given in Table XII.

TABLE XII

TENSILE AND SHEAR STRENGTH OF n-PbTe ELEMENTS

Specimen OD cm (in.)	Specimen Number	Tensile Strength psi
	Pellet	1080
0.635 (0.25)	Element 1 2 3	712 1570 *
0.951 (0.375)	" 4 " 5 " 6	361 * 729
		Shear Strength psi
0.951 (0.375)	" 7 " 8	284 203

^{*} broke while attaching tensile fixtures

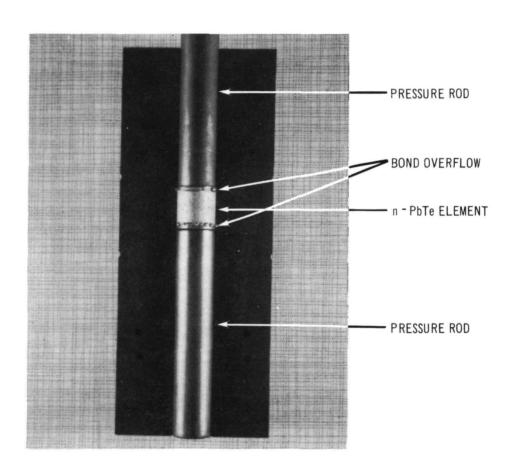


Figure 33. PbTe Element in Shoe Bonding Die

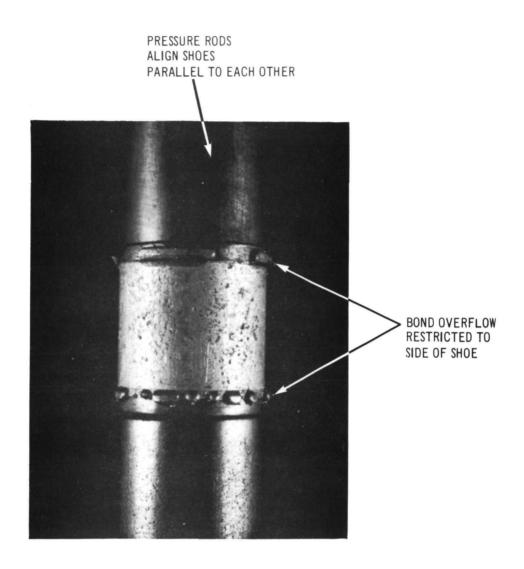


Figure 34. n - Type PbTe Element - Bond Overflow

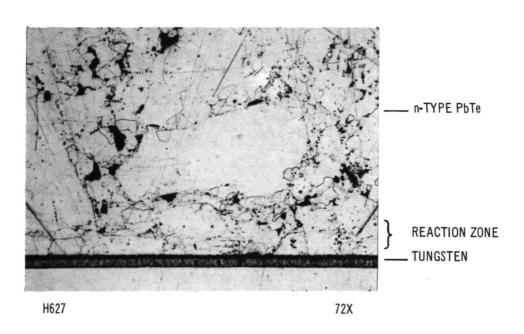
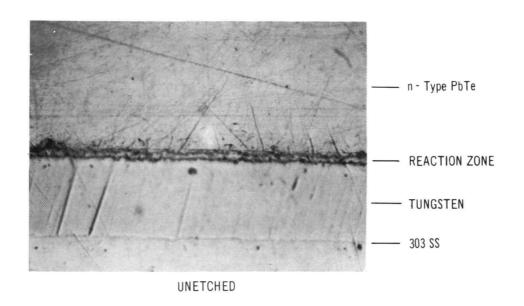


Figure 35. PbTe 3N Element Bond Structure



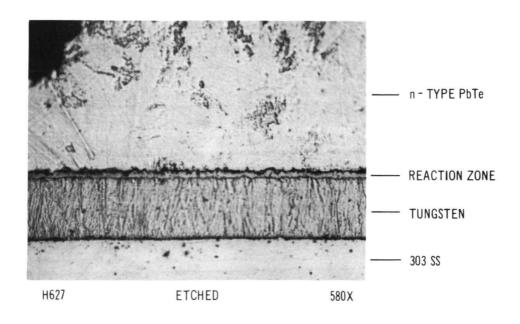


Figure 36. PbTe 3N Element Bond Structure

Note that the strength of the elements is lower than that of the pellets. The element strength, however, should be adequate for the stresses encountered in the device. The current technique of shear testing was not suitable for elements with 0.635 cm (0.250 in.) OD and no shear strength data were obtained for these elements. Resistance measurements made before and after isothermal life tests of the 0.635 cm (0.25 in.) OD and 0.951 cm (0.375 in.) OD PbTe elements are tabulated in Table XIII. Note that the sum of the PbTe bulk and contact resistances do not equal the total measured resistance since some of the bulk resistance is unavoidably included in the measurement of the contact resistance. Only minimum changes occurred in the electrical resistance and physical appearance of the elements subjected to isothermal life tests. Metallographic examination of the life-tested specimens showed sound structures with very little discernible change in the PbTe-W barrier nickel-303 stainless steel bond structure (Figures 37 and 38). In general, the n-PbTe elements subjected to life tests have shown good stability.

4. Cold Stack

The cold stack subcomponent consists of the n- and p-electrical connectors (copper), ceramic insulators (alumina (Al203)), mount stud (copper and steel), and copper exhaust tubulation brazed together into a single subassembly. Previous studies have shown that the use of a T-mount copper stud (Figure 39a) rather than a tapered mount steel stud (Figure 39b) results in an improved temperature drop (10°C versus 17°C). The T-mount stud also was found to minimize stresses which may be transmitted to the ceramic insulator when the stud is secured to the module mounting panel. These stresses could ultimately fracture the insulator and produce a separation in the cold stack. Shock and vibration measurements made on the T-mount design have also indicated the superiority of this design concept. Accordingly, this concept was pursued in the development of the cold stack for the Hybrid couple.

Two methods of fabricating the T-mount stud were investigated; (1) a single piece copper stud including the threaded section (Figure 39a), and (2) a multipiece stud, copper with a brazed steel threaded section (Figure 40). Strength tests of the "as fabricated" cold stacks, utilizing the single piece copper stud including the threaded section (Figure 41a) were made and compared with those of the multipiece T-mount copper stud employing brazed threaded steel section (Figure 41b). Force was applied to the nut with a torque wrench in increments of 10 inchpounds, starting at 10 to 50 in.-lbs. Deformation of the one piece stud was noticeable in the lower half of the stud and the thread stripped at 45 in.-lbs on one sample. No distortion of the parts was detected in the multipiece stud with the brazed threaded steel section at 50 in.-lbs. The two-piece T-mount was therefore selected for the cold stack construction.

A total of nine cold stack structures of the multipiece stud design were fabricated and subjected to the following tests:

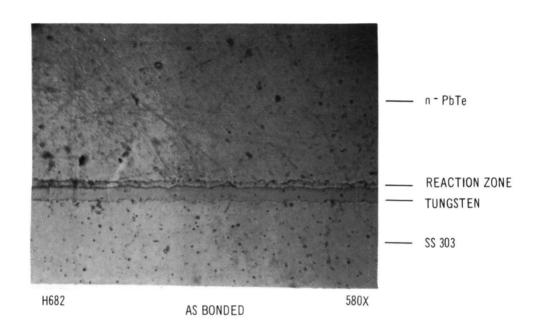
TABLE XIII

RESISTANCE OF PbTe ELEMENTS
SUBJECTED TO ISOTHERMAL LIFE TESTING IN ARGON

Test	Element OD cm (in.)	Soak Temperature <u>°C (°F)</u> RT	Time at Temperature hours AB	Total 0.313		e, mill acts 2 0.020	Bulk PbTe 0.30
1		540 (1000)	330	0.341	0.020	0.020	0.32
2		RT	AB	0.32	0.025	0.025	0.30
	0.951 (0.375)	540 (1000)	330	0.335	0.025	0.020	0.30
	0.751 (0.575)	RT	AB	0.34	0.020	0.020	-
3		593 (1100)	138	0.385	0.025	0.020	0.345
		593 (1100)	252	0.365	0.025	0.030	0.330
4		RT	AB	0.32	0.015	0.020	-
4		593 (1100)	138	0.343	0.015	0.020	
5		RT	AB	0.92	0.015	0.025	0.910
		540 (1000)	475	0.99	0.030	0.040	0.960
6		RT	AB	0.92	0.022	0.035	0.905
	0.635 (0.250)	540 (1000)	475	0.98	0.020	0.030	0.950
~		RT	AB	0.93	0.020	0.025	0.890
7		593 (1100)	159	0.935	0.025	0.030	0.900
8		RT	AB	0.94	0.020	0.030	0.915
8		593 (1100)	159	0.97	0.030	0.040	0.935

RT Room Temperature

AB As Bonded



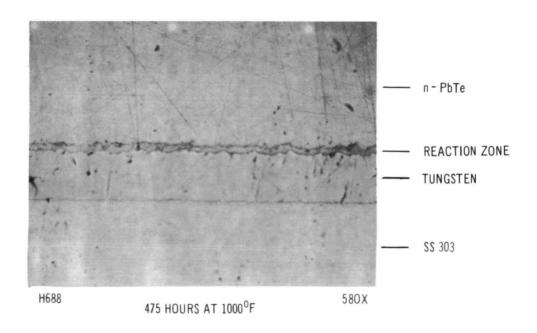
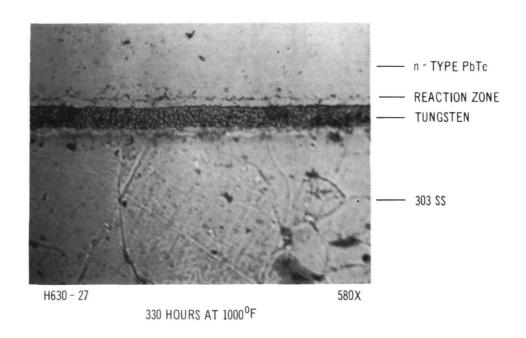


Figure 37. 3N PbTe Element Shoe Bond Structure



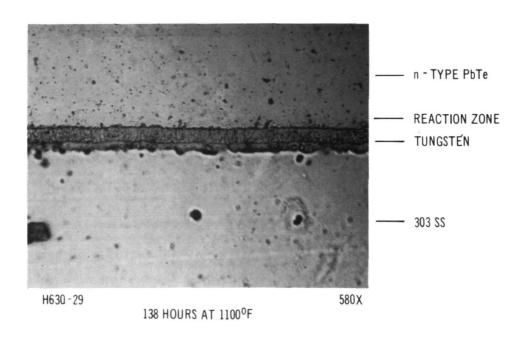
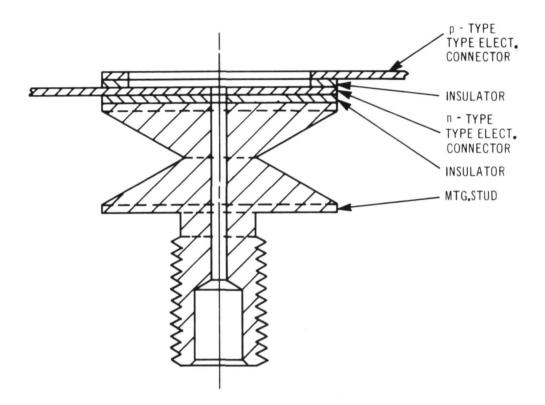
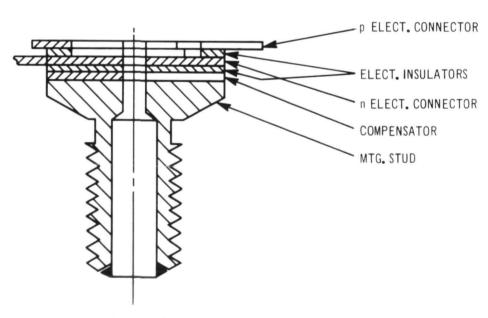


Figure 38. 3N PbTe Element Shoe Bond Structure



a. One Piece Copper T-Mount Stud-Cold Stack Design



b. Tapered Mount Steel Stud Cold Stack Designs

Figure 39. Experimental Cold Stack Designs

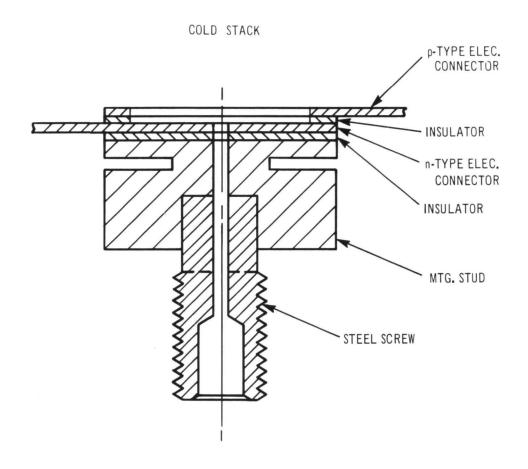


Figure 40. Two Piece T-Mount Stud — Cold Stack Design (Copper T - Steel Screw)

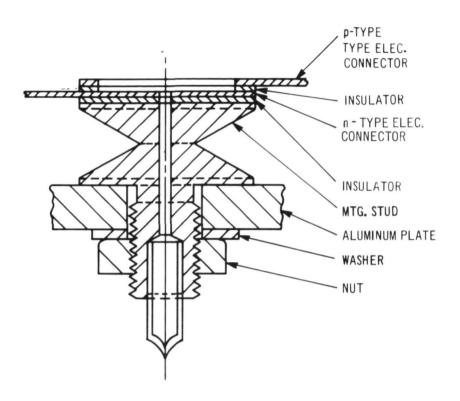


Figure 41a. One Piece Copper T- Mount Stud - Cold Stack Design - Aluminum Plate

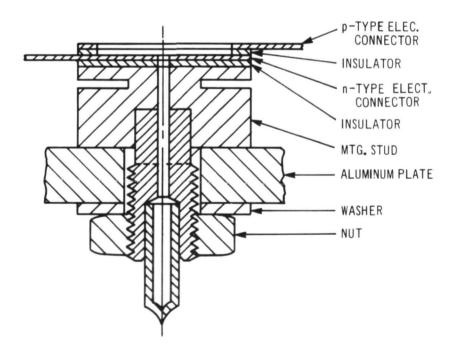


Figure 41b. Cold Stack Design (Copper T - Steel Screw) Aluminum Plate.

Number of Samples	Type of Test
1	Metallographic examination
2	Tensile test
2	Shear test
2	300 hours life test
2	Accelerated life test

<u>Metallographic analysis</u> of the structure showed excellent brazes between all component parts.

Tensile test: This test was performed by exerting a longitudinal force between a tungsten disc and screw (see Figure 42). The forces required to break the samples were 320 pounds (6400 psi) to 400 pounds (8000 psi). Both breaks occurred through the ceramic insulator ring, showing typical tensile cracks (Figure 43).

Shear test: This test was performed by exerting a force at right angles to the longitudinal plane of the structure (Figure 44). One sample was lost when tightened into the Instron fixture. The other sample failed at 355 pounds (7100 psi). The mechanism of failure in both samples was through the ceramic insulator ring.

<u>Life tests</u>: Two samples completed 170 hours of testing at 200°C and another two samples 170 hours at 300°C, under isothermal conditions.

All the above samples were checked for hermeticity after life testing and were found to be leak-tight. Metallographic analyses of these samples showed sound, void-free structures.

5. Intermediate Bond System

The intermediate bond system includes (1) the top gold compensator bonded between the n-SiGe tungsten cold shoe and the n-PbTe 303 stainless steel hot shoe, and (2) the lower gold compensator bonded to the n-PbTe 303 stainless cold shoe. The intermediate bond subcomponent consists of a gold disc diffusion bonded between a nickel-plated 303 stainless steel disc and a nickel-plated tungsten disc (Figure 45). This bond system consists of a nickel-gold diffusion bond which has been used in a number of air-vac thermoelectric devices with excellent results. The intermediate bond test coupons were fabricated according to the process HTS-4 described in Section V-A. They were subjected to the following screening tests:

Tests	Number of Samples	Temperature °F	Time hours
Tensile	2	room temp.	-
Shear	2	room temp.	-
Isothermal life	2	1100	330
Isothermal life	2	1200	4
Metallographic evaluation	1	room temp.	

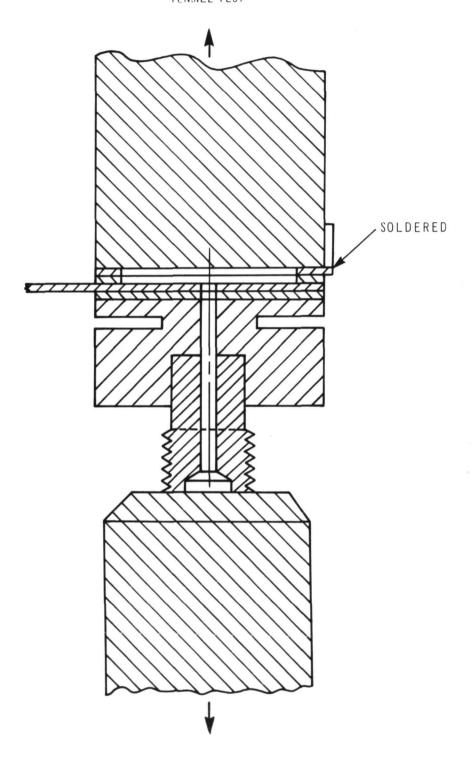


Figure 42. Two Piece T - Mount Stud - Cold Stack Design (Copper T - Steel Screw)

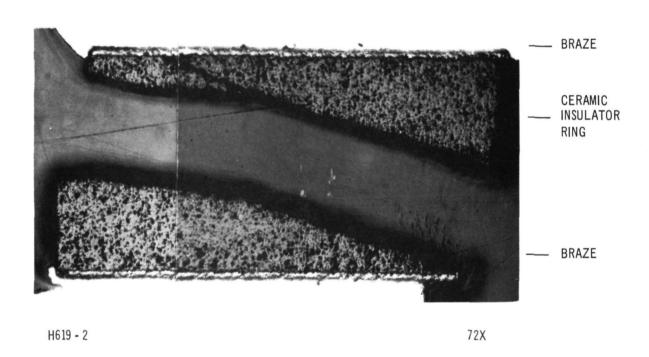


Figure 43. Tensile Tested at 320 lbs.(6400 psi)

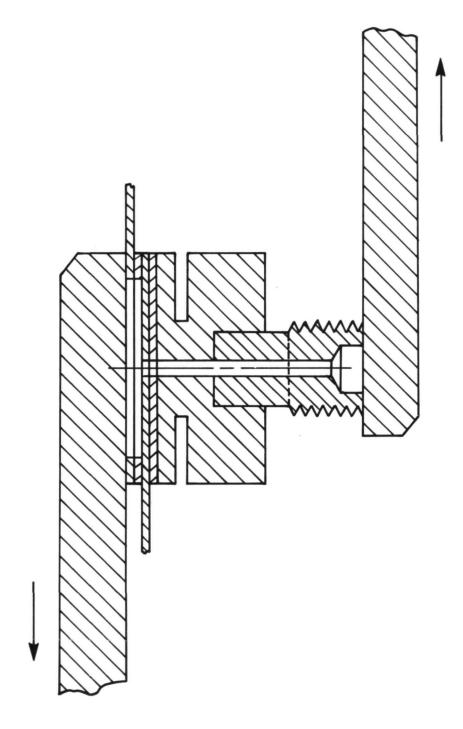


Figure 44. Two Piece T- Mount Stud - Cold Stack Design (Copper T - Steel Screw)

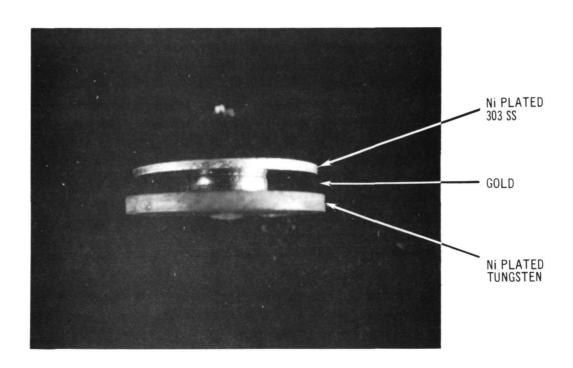


Figure 45. Intermediate Bond System Coupon

The tensile testing resulted in the separation of the specimens in the gold-nickel bond region, as expected. Specimens subjected to accelerated life testing had approximately the same strength as as-bonded specimens (i.e., 6200 and 11,900 psi as-bonded compared to 8600 and 6400 psi after life test). The tensile results are summarized in Table XIV.

TABLE XIV

TENSILE STRENGTH OF INTERMEDIATE BOND COUPONS

Specimen Number	Soak Temperature °C (°F)	Time at Temperature hours	RT Tensile Strength psi
1	RT	AB	6,200
2	RT	AB	11,900
3	593 (1100)	330	8,600
4	640 (1200)	4	6,400
5	593 (1100)	481	9,660
6	649 (1200)	4	11,890*

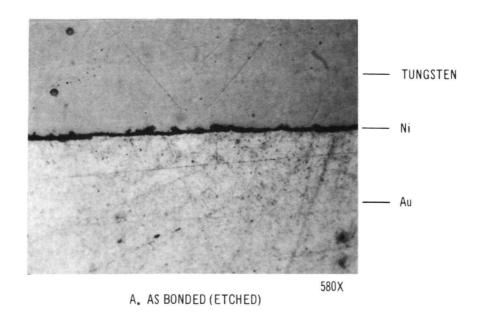
^{*} attachment failure; coupon did not rupture

Shear strengths of two as-fabricated specimens were 11,500 and 13,800 psi. Metallographic examination showed that the bonds were sound, both before and after life test; see Figures 46 and 47. Overall integrity of this bond system was excellent.

B. Thermocouple Development

Hybrid thermocouple assembly sequences and material and part specifications are presented in detail in Section V-A. The preliminary thermocouple assembly sequences selected are described below (see Figure 48).

1. Heat Receptor Subassembly (A). This assembly consists of an n-SiMo disc, 0.320 in. OD, 0.040 in. thick, metallurgically bonded via a diffusion barrier to a 0.960 in. x 0.960 in. x 0.125 in. thick p-SiMo heat receptor plate.



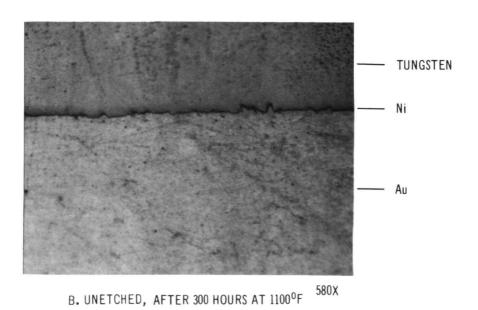


Figure 46. Intermediate Bond Coupon, Tungsten - Gold Bond Structure

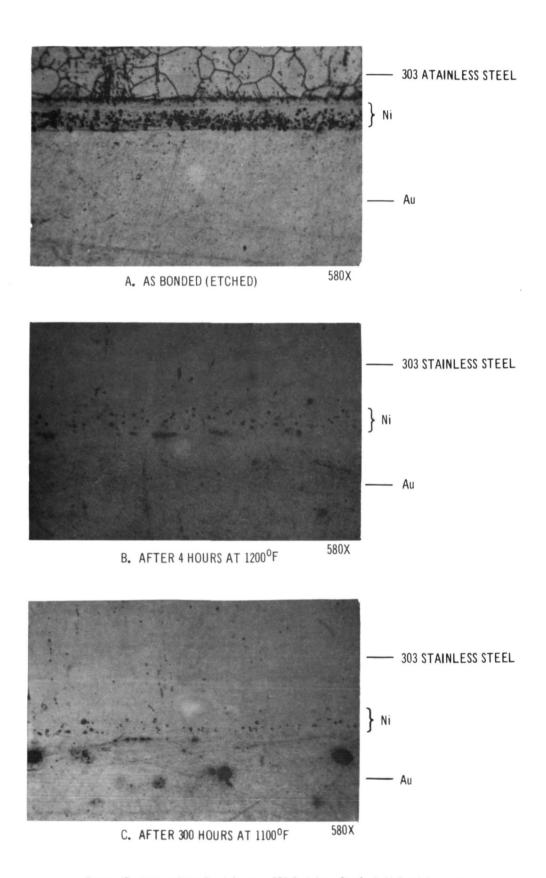


Figure 47. Intermediate Bond Coupon, 303 Stainless Steel-Gold Bond Structure

- 2. <u>SiGe Subassembly (B)</u>. The heat receptor assembly (A), p-type SiGe cylinder, n-type SiGe pellet, and n- and p-type SiGe tungsten cold shoes are simultaneously metallurgically bonded together.
- 3. n-type PbTe Subassembly (C). This subassembly is identical to sub-component (C), n-PbTe element, discussed in Section IV-A-3.
- 4. Thermocouple Subassembly (D). The SiGe subassembly (B), n-PbTe subassembly (G) and the gold compensators are diffusion bonded together.
- 5. <u>Cold Stack Subassembly (E)</u>. This subassembly is identical to subcomponent (D) discussed in Section IV-A-5.
- 6. Final Assembly (F). The thermocouple subassembly (D) and the cold stack subassembly (E) are brazed together to form the Hybrid thermocouple. Prior to this final assembly step, the gap between the n-segment and the wall of the p-SiGe cylinder is filled with 99.5% aluminum oxide powder and microquartz insulation to reduce heat transfer through this space. Following this final braze operation, the couples are evacuated and back-filled with argon, and the copper tubulation is pinched off.

Based on the results of the subcomponent screening tests described in Section IV and preliminary investigation of thermocouple assembly techniques, two preliminary Hybrid Couple Designs, B and B-1 (see Table V), were selected as the designs to be used in the development of the thermocouple fabrication techniques. These two designs are identical except B-1 has an increased p-SiGe cylinder wall thickness. As a result of the effort conducted in this study, couple design B was selected as the Hybrid Reference Design Couple.

The development effort conducted in arriving at a preferred method for fabricating the Hybrid Couple and selecting the Reference Design involved many variations of the initial assembly methods described above. A summary of the three principal fabrication methods employed throughout this development phase and the final method selected are presented in Table XV.

Assembly technique A (Figure 48) was the initial method; techniques B (Figure 49) and C (Figure 50) were intermediate techniques; and technique D (Figure 51) was the final method selected for fabrication of the Hybrid module panel sections.

A list of other important fabrication modifications made throughout the development program, in addition to those given in Table XV, are presented chronologically in Table XVI. A detailed discussion of the bonding processes and assembly techniques investigated in arriving at the final assembly technique D (Figure 51) is presented in the following.

TABLE XV
SUMMARY OF FABRICATION METHODS - HYBRID THERMOCOUPLE

Assembly Method D Fig. 48* Fig. 50* Fig. 51* 1. Heat Receptor Subassembly (A) X X Heat Receptor Subassembly (B) 2. X SiGe Subassembly (A) X 3. SiGe Subassembly (B) X 4. Heat Receptor-SiGe Subassembly X PbTe Element Subassembly X 6. X X X X X X 7. Thermocouple Subassembly X 8. Cold Stack Subassembly X X X X 9. Final Assembly X X X X

^{*} See Figures 48 through 51 for detailed description of assembly steps

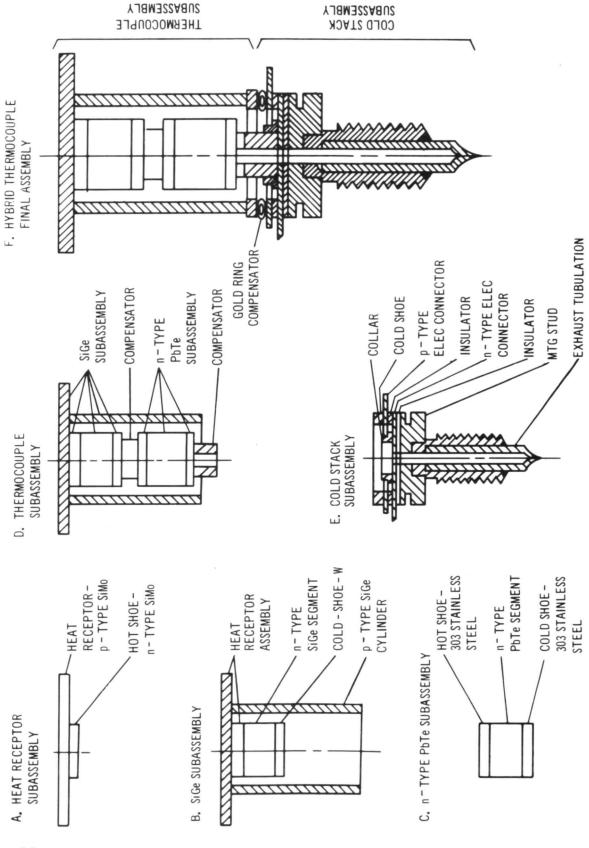


Figure 48. Hybrid Thermo couple Assembly Sequence - A

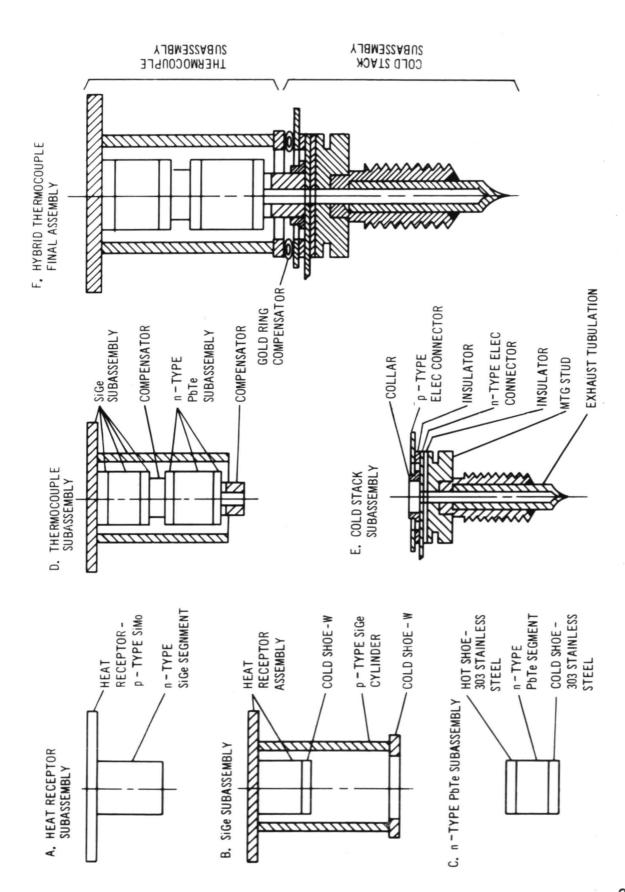


Figure 49. Hybrid Thermo couple Assembly Sequence - B

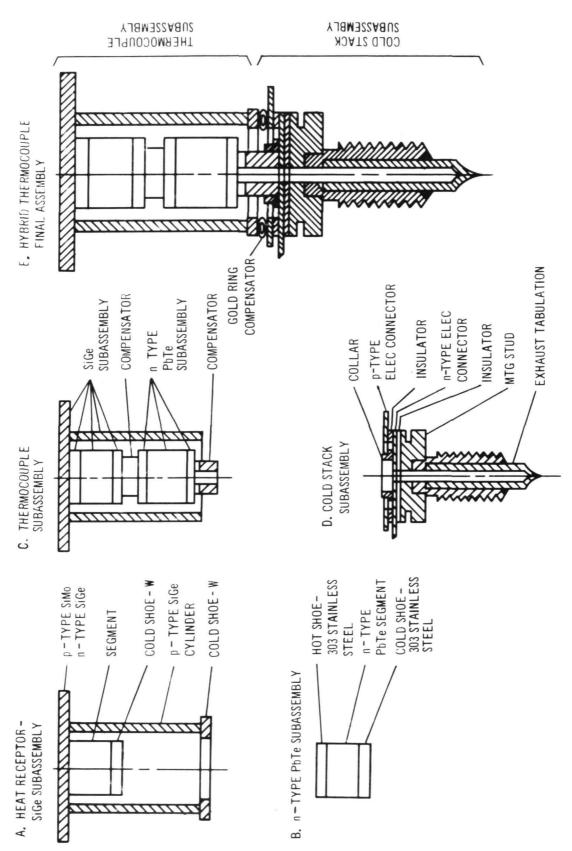


Figure 50. Hybrid Thermocouple Assembly Sequence - C

Figure 51. Hybrid Thermocouple Assembly Sequence - D

TABLE XVI

CHRONOLOGICAL FABRICATION MODIFICATIONS - HYBRID THERMOCOUPLE

Results	Both gold compensator bonds to (1) n-SiGe cold shoe and n-PbTe hot shoe and (2) n-PbTe cold shoe were excellent.	p-SiMo heat receptor leak-tight; eliminated small fissures.	p-Sige-tungsten cold shoe bond leak- tight; eliminated small fissures.	Sound bond interface with good mechanical integrity.	Obtained excellent metallurgical bonds with variations in initial electrical resistances.	Obtained excellent metallurgical bonds with low electrical resistances.
Fabrication Modification	Changes from "one step" bonding of gold compensators (2) to "two step" bonding operation.	Used bonding support "pad" same diameter as n-SiGe to replace large (p-SiMo) "hot shoe" area pad.	Introduced "gold ring" between p-SiGe-tungsten cold shoe and p-electrical connector.	Bonded n-Sige leg via titanium barrier directly to p-SiMo heat receptor.	Bonded n-SiGe, p-SiGe (including respective tungsten cold shoes) to p-SiMo heat receptor in one bonding operation.	Increased p-SiMo thickness to 0.125 in. and n-SiMo thickness to 0.040 in.
Subassembly Operation	Thermocouple Subassembly		Final Assembly	Heat Receptor Subassembly	Combined Heat Receptor and Sige Subassemblies operations	Heat Receptor Subassembly
Couple Number	B-4		B-8	B-10	B-18	* P-1

* Fanel Couples

1. Heat Receptor Subassembly

Two different bonding methods were investigated during the development of the heat receptor subassembly. The first utilized the standard air-vac technology of hot pressing the p- and n-types SiMo heat receptors with a titanium diffusion barrier. The second method employed a high temperature diffusion bond again using titanium.

Initially the component dimensions used in both studies were: p-SiMo, 2.54 cm $(1.00 \text{ in.}) \times 2.54 \text{ cm} (1.00 \text{ in.}) \times 0.19 \text{ cm} (0.075 \text{ in.})$ thick, and the n-SiMo hot shoe, 0.635 cm (0.250 in.) 0D by 0.051 cm (0.020 in.) thick. Both of the materials were vacuum cast with a composition of 85 wt % Si, 15 wt % Mo.

Attempts to hot press the 0.051 cm (0.020 in.) thick n-SiMo disc to the p-SiMo heat receptor plate resulted in good metallurgical bonds; however, extensive cracking of the thin n-SiMo hot shoe occurred. The hot press pressure control was not sensitive enough to apply the required pressure to such a small area part and the excessive pressure caused the cracking. Because of equipment limitations and the fact that only one subassembly could be processed at one time, this approach was eliminated. Initial attempts at diffusion bonding the thin n-SiMo hot shoe to the p-SiMo heat receptor resulted in a sound metallurgical bond. However, due to the thinness of the n-SiMo, the diffusion barrier material diffused through to the outer surface which resulted in a poor bond in the subsequent subassembly operation (SiGe subassembly) at the n-SiMo/n-SiGe interface. In an attempt to eliminate this difficulty, a direct bond of n-SiGe-(diffusion barrier) to p-SiMo was made, thus eliminating the thin n-SiMo disc.

Two variations were investigated; (1) Assembly Sequence B, heat receptor subassembly (see Figure 49, part A); and (2) Assembly Sequence C, heat receptor-SiGe assembly (see Figure 50, part A). Both methods resulted in sound metallurgical bonds initially; however, some variations in initial electrical resistances were observed. During life testing of couples using these bonds, anomalies in the electrical resistance occurred. Because of the uncertainties regarding the n-SiGe/p-SiMo bond, it was decided to return to the conventional air-vac n-SiMo to p-SiMo bond system. In order to alleviate the cracking problems encountered earlier with this system, the thickness of n-SiGe disc was increased from 0.051 cm (0.020 in.) to 0.102 cm (0.040 in.), and that of the p-SiMo heat receptor plate from 0.19 cm (0.075 in.) to 0.318 cm (0.125 in.). Also, hot pressed n-SiMo was substituted for the vacuum case material (n-type only). These modifications resulted in sound metallurgical bonds in the Hybrid couple with low electrical resistances. A photomicrograph of this bond is shown in Figure 52. This bond system was employed in the fabrication of the Hybrid panel couples. Figure 51, part A, shows the final heat receptor assembly configuration.

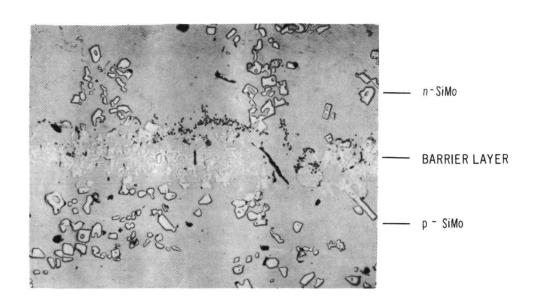


Figure 52. Heat Receptor Bond — p-SiMo — n-SiMo

2. SiGe Subassembly

The principal variations investigated in the fabrication of the SiGe subassembly were necessitated by difficulties in the heat receptor assembly, as discussed previously. The attempt to make this subassembly in one step as opposed to two (process C versus process B), showed this approach to be feasible, but the use of the air-vac heat receptor subassembly for the final assembly technique (Figure 51, part A) precluded the use of this approach.

A minor modification made in the SiGe subassembly was the inclusion of the tungsten cold shoe ring of the p-SiGe cylinder in the bonding operation, process D (Figure 51, part B).

3. n-Type PbTe Subassembly

No changes or modifications were made in the assembly technique developed for the n-type PbTe subassembly (subcomponent (C)). Details of this subassembly are presented in Section IV-A-3.

4. Thermocouple Subassembly

Two difficulties were experienced in the development of the assembly technique for this subassembly; (1) fissures developed in the p-SiMo heat receptor during the diffusion bonding operation, and (2) the bond of the upper gold compensator to the n-SiGe cold shoe and n-PbTe hot shoe was marginal. It was found that by using a support pad (under the p-SiMo heat receptor), having the diameter of the n-SiMo disc rather than the area of the heat receptor, the fissures were eliminated and the p-SiMo heat receptor was leak-tight; The upper and lower gold compensators have different diameters (different areas), the upper having the larger area, and it was found that in the bonding operation insufficient pressure was being transmitted to the upper compensator, thus resulting in a poor bond. The process was changed from a "one step" to a "two step" bonding operation. First, the upper compensator bond was made, then the lower compensator was bonded with the bonding pressure adjusted for each area. All resultant bonds were excellent.

5. Cold Stack Subassembly

Only a minor modification was made in the cold stack as developed in the subcomponent (D), Section IV-A-5: a "collar" or flange was introduced to mate with the lower gold compensator in the final assembly, thus providing the thermal and electrical path for the n-segment.

6. Final Assembly

The only modification made in the final assembly was the incorporation of a gold ring, made from hollow gold tubing, 0.050 in. 0D with a 0.010-in. thick wall, between the p-SiGe tungsten cold shoe and the cold

stack. This modification eliminated stresses which caused fissures in the p-SiGe--tungsten bond area. The resultant bond area was sound metallurgically and leak-tight.

A summary of the thermocouples fabricated in this task and their disposition is given in Table XVII. The test results of the couples are discussed in the following section IV-C. The final assembly technique developed, assembly sequence D, Figure 51, was used to fabricate the Hybrid couple for the module panels.

C. Thermocouple Testing

Testing of the developmental Hybrid couple structures was conducted simultaneously with the thermocouple structure development phase to provide rapid feedback so that modifications could be made to either the processing or components, as required. Both design and accelerated life testing and thermal cycling tests were employed in these evaluation.

The design test conditions for the Hybrid couple are hot junction temperature, $T_{\rm HJ} = 926^{\circ}\text{C}$ (1700°F); interstage-n-PbTe hot shoe temperature, $I_{\rm I} = 538^{\circ}\text{C}$ (1000°F); and cold junction temperature, $T_{\rm CI} = 232^{\circ}\text{C}$ (450°F).

The first Hybrid couple tested was couple B-l-l, which developed high resistance after 228 hours and three thermal cycles at design test conditions. Upon removing the couple from the test stand, a separation occurred in the brazed bond between the p-SiGe and the tungsten ring of the cold stack. This was the only couple tested of the B-l design (i.e., having the thicker p-SiGe wall) as success in fabrication of the thinner wall p-SiGe cylinders allowed the selection of couple design B as the Hybrid Reference Design thermocouple.

A summary of the Reference Design couples tested (Design B) is given in Table XVIII comparing "initial" (20-hour) data, 1500-hour reference design point for SiGe, and longest data point for each individual couple.

Couples B-7 and B-9 experienced an abnormal increase in couple resistance after a relatively short time on test; 328 hours and 656 hours, respectively. Couple B-7 separated on demounting at the brazed joint between the p-SiGe and gold compensator ring similar to the failure exhibited in couple B-1-1. This obviously weak bond, gold compensator brazed directly to SiGe, was eliminated starting with couple B-8. The tungsten ring (p-SiGe cold shoe) was first bonded directly to p-SiGe and the gold compensator ring was subsequently brazed between the tungsten ring and the p-electrical copper connector of the cold stack.

Couple B-9 had been operated at an accelerated hot shoe temperature, 960°C (1760°F) versus 941°C (1725°F). The couple was demounted and an X-ray radiograph taken in order to examine the internal couple structure. It showed that a separation had occurred in the n-leg between the n-PbTe and its hot shoe. Upon further examination, a ring of material was noted on

TABLE XVII SUMMARY OF HYBRID COUPLES FABRICATED

Dogiam No	Pro-	Dignogition	Roma rika
2 6 7 8 9	A 2 3 3 4 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5	Disposition Life tested Completed TC subassy Completed TC subassy Completed TC subassy Completed TC subassy Life test Completed TC subassy Completed TC subassy Metallographic exam.	Remarks p-SiMop-SiGe brazed bond leaked p-SiMop-SiGe brazed bond leaked p-SiMop-SiGe bond leaked p-SiMop-SiGe bond leaked Metallographic examination n-SiMop-SiMo separation n-SiMop-SiMo separation Fissures braze p-SiGeW
B 3 3 4 5 5 6 6 7 5 6 6 7 6 7 6 7 6 7 6 7 6 7 6	A 2 3 3 4 5 5 5 6 7 7 8 8 6 5 6 7 7	Completed TC subassy Not tested Completed TC subassy Not tested Metallographic exam. Life test Life test Life test Life test Completed TC subassy Life test	Met. exam; poor bond upper Au comp. Leaked brazed bond p-SiGeW PbTe hot shoe separation High res. n-leg (n-SiGe to p-SiMo) PbTe hot shoe bond poor Bonds OK n-SiGep-SiMo bond separation n-SiGep-SiMo bond separation n-SiGep-SiMo bond separation n-SiGep-SiMo bond separation
G 1	2 3 4 5 5		Bonds OK Bonds OK for test and evaluation for test and evaluation
BX ⁽¹⁾ 1	2	Life test Life test Life test Life test	
P(2) 1-2			9-couple Hybrid Module Panels. is for their test and evaluation.

NOTES: (1) Special resistance test couples (2) Couples fabricated for Hybrid Module Panels.

TABLE XVIII

HYBRID COUPLE PERFORMANCE SUMMARY

	PML	0.52	0.58	0.62	0.76	0.67	77.0	0.54	09.0	
	Res.	39.1	32.8	37.0	30.6	38.4	43.4	36.6	8.07	
leted	P O O	228	237	229	229	220	231	226	234	
Test Completed	T S	921	897	932	951	941	406	698	939	
Tes	Test Time,	7845	328	2868	959	6284	5888	9689	5550	
	No. of Thermal Cycles	N	1	М	11	5	10	10	6	
	PML	0.65	1	69.0	1	19.0	0.45	0.55	0.63	0.62
	Res.	33.6	1	32.4	1	37.0	43.7	37.3	39.3	32.5
500 Hr.	o H	230	1	235	1	216	228	226	230	231
Nominal 1500 Hr.	E o	915	1	938	1	937	806	872	934	903
Nom	Test Time, Hrs.	1506	1	1553	1	1482	1834	1494	1382	1171
	No. of Thermal Cycles	0		0		2	9	7	9	N
	PML	0.70	0.61	0.81	0.83	0.71	67.0	0.59	99.0	0.73
ŀ.	Res. PML mohm watt	30.8	30.3	26.6	27.0	32.54	39.12	31.56	36.10	78.8
1 20 H	o H	229	233	233	231	216	223	219	225	200
Nomina	E o	935	910	176	096	938	706	863	076	106
	No. of Termal Ts Tc Res. P _N Sycles $\frac{\circ}{C}$ $\frac{\circ}{C}$ mohm we	0	0	0	0	1	5	5	5	0
	Couple No.	B-6	B-7	B-8	B-9	B-12 ¹	B-14 ¹	$B-16^{1}$	$B-17^{1}$	B-18 ¹

Note 1: These data have been normalized to the design test conditions via the $\Delta \, \mathrm{T}^2$ relationship as described below:

Normalized
$$P_{Norm} = P_{ML} \times \frac{(\Delta T_N)^2}{(\Delta T_X)^2}$$
 $T_N - Design Temperature$

the inside surface of the p-SiGe cylinder wall adjacent to the n-PbTe hot shoe bond. A chemical analysis of this deposit showed the composition to be Pb and Te. No effect on the microstructure or the integrity of p-SiGe cylinder was noted. It would appear that the PbTe hot shoe bond was operated above the 538°C design temperature and PbTe sublimation occurred.

Couples B-6 and B-8 completed 4842 and 2868 hours of life testing at nominal design operating temperatures. Couple B-6 developed high resistance after a severe thermal cycle due to a power failure. Separation occurred in the brazed bond region between the p-SiGe and gold compensating ring and at the PbTe hot shoe bond. Both of these areas were modified in later couples, initiating with couples B-8 and B-10, respectively. Further inspection indicated that the PbTe shoe bond separated first, while contact was sustained by the p-leg until separation ultimately occurred there also. The life test station in which couple B-8 was being tested slowly developed a leak which degraded both the heater and the instrumentation attached to the couple. This couple was not initially vacuum-tight due to a problem in brazing the gold ring to the tungsten cold shoe at the time of fabrication; therefore, the inside of the couple was exposed to oxygen. The life test station was cycled down following the last measurement. When cold, the couple showed an abnormally high resistance, in ohms, in the n-leg. Subsequent X-ray examination of this couple showed that PbTe material from the hot shoe bond area had deposited on the inside of p-cylinder wall. Various X-ray views of this bond, however, did not show physical separation as was apparent in B-9, reported above. high resistance was evidently a result of the loss of PbTe material from the hot shoe bond area and probably oxygen contamination of the PbTe element.

A program test goal was to operate a number of thermocouples for a 5000hour life test, prior to building the final Hybrid module test panels, to assure that the couple structure and performance were satisfactory. The next four couples tested, viz., B-12, -14, -16 and -17, were all fabricated using process B. Although the initial resistances of these couples were somewhat high, ranging from 31 to 39 mohms, reasonable stability was exhibited during the 5000-hour test. Subsequent p-series Hybrid couples, fabricated for the 9-couple panels using process D, exhibited much lower initial resistance, ranging from 25 to 28 mohms, and also exhibited improved stability after about 700 hours of testing. The resistance of 7 of the 9 p-couples ranged from 28 to 30 mohms at 700 hours. (The P-Hybrid couple data is based on NASA-LeRC test results, to date, on panel #1.) The increase in resistance of the individual couples as a function of test time is presented in Figure 53. Note the high resistances exhibited by the B couples compared to that of the P couples. The sharp increase in the resistance of couples B-6 and B-8 indicates the onset of the bond separation described above.

The original design data obtained early in the program predicted a power output of 0.87 watts/couple. Since that time, data gathered in both the Silicon-Germanium Thermoelectric Material and Module Development Program, Contract AT(29-2)-2510, and the Multi-Hundred Watt RTG Program, Contract

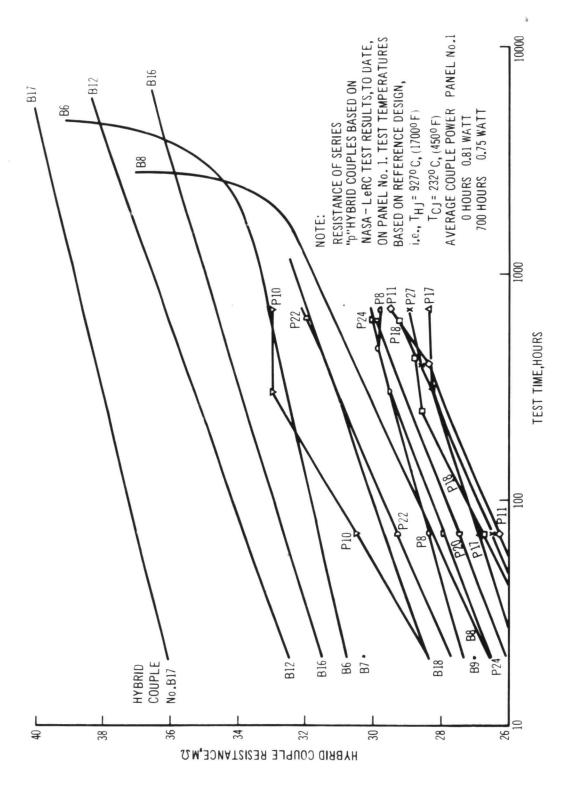


Figure 53. Hybrid Couple Resistance Plotted Against Test Time

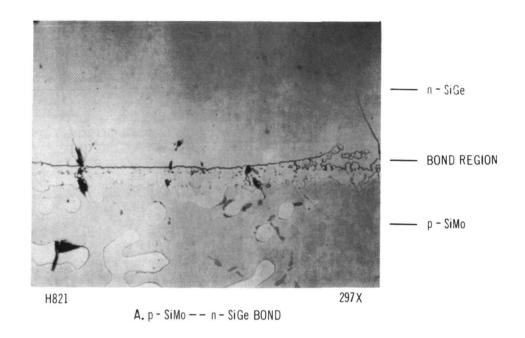
AT(29-2)-2831, have established the "contact" resistances of air-vac couples at 20% of the SiGe material resistance. This compares with approximately 5% assumed at the start of the Hybrid Program and reduces the predicted design value at the 1500-hour stabilization point from 0.87 watts to 0.77 watts per couple. Measured power of the four B couples is approximately 0.63 watt at 1500 hours while the P couples average 0.75 watt at 700 hours compared to the predicted (1500 hour) design of 0.77 watts per couple.

Couples B-12, B-16 and B-17, with 6284, 6396 and 5550 hours of life testing, respectively, were examined by taking radiographs to determine their condition. Couple B-14 with 5888 hours of life testing was damaged upon removing it from the test stand and was not submitted for examination. The radiograph of couple B-12 showed no cracks or separations in the cylinder wall or bond areas.

The radiograph of B-16 also showed no separations in any of the bonds. However, the start of deterioration of the bond between the stainless steel hot shoe and the lead telluride is evident. This deterioration is manifest by the erosion of the lead telluride at the edges of the pellet at the interface of the hot shoe and the telluride. Also evident in the radiograph is a misalignment of the gold compensator between the cold shoes of the n-SiGe and the hot shoe of the n-lead telluride. The radiograph of couple B-17 shows a definite separation between the stainless steel hot shoe and the n-lead telluride with evidence of some telluride material deposited on the inner wall of the cylinder in the vicinity of the separation. This deposit is evident in most couples; however, it appears to be heavier in the case of couple B-17. This heavy deposit of material on the cylinder wall seems to indicate the loss of hermeticity in the couple and a higher than normal operating temperature.

Based on the radiographs, couple B-12 was selected for metallurgical examination. The examination of B-12 revealed no cracks in the p-SiGe cylinder wall or p-SiMo hot shoe. No separation was found at any of the bonds including the braze joints of the cold stack. Figures 54a, 54b and 55 are photomicrographs of bond areas of p-SiMo--n-SiGe, n-SiGe to tungsten, and stainless steel hot shoe n-lead telluride, respectively. Each shows sound bonds with no evidence of deterioration. The barrier layer in p-SiMo to n-SiGe and stainless steel hot or cold shoe to lead telluride bonds is clearly evident and is continuous for the entire length of the bond.

Concurrent with the testing period of the above four couples, it was decided to fabricate and specially instrument a Hybrid thermocouple so that individual resistances and temperature measurements could be monitored for the following: (1) n-leg, individual SiGe and PbTe segments; (2) n-SiGe--p-SiMo segment; (3) p-SiGe--p-SiMo segment; and (4) overall couple resistance. Accordingly, couple B-18 was instrumented and placed on test. Figure 56 shows the location of the attached wires for both resistance and temperature measurements. All wires except that at location



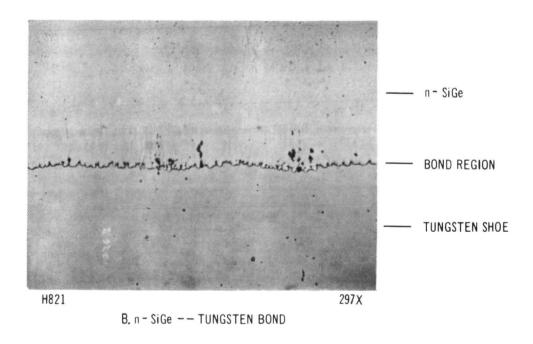


Figure 54. Couple B-12, 6284 Hours — T_S 938°C (1720°F) — T_C 216°C (418°F)

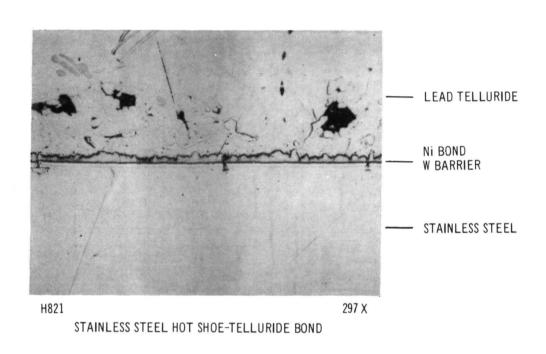


Figure 55. Couple B - 12, 6284 Hours – T $_{\rm S}$ 938°C (1720°F), T $_{\rm C}$ 216°C (418°F)

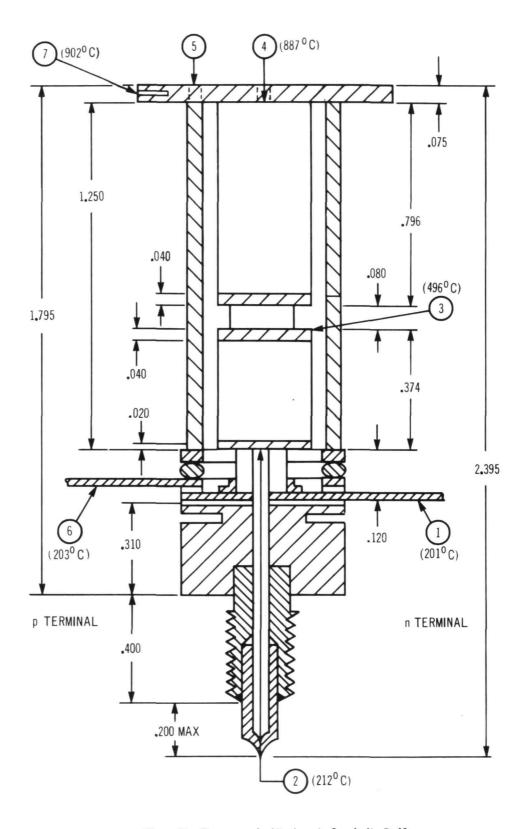


Figure 56. Thermocouple Attachments Couple No.B-18

#5 denote coincident temperature and resistance measurement locations. Location #5 was for resistance measurements only. All resistance measurements were made using a milliohmeter after the couple temperatures had stabilized at open circuit conditions. Comparison of resistance design data to actual measured resistances shows that the greatest deviation occurs in the segment between n-PbTe hot shoe and the p-SiMo hot shoe (location 3-5). Comparable resistances between locations (see Figure 56) are shown in Table XIX.

TABLE XIX

SPECIALLY INSTRUMENTED THERMOCOUPLE B-18

			Resistan	(mohms)			
Segment	Location	<u>Design</u>	20 <u>hours</u>	226 ³ hours	514 hours	835 hours	1172 hours
n-PbTe	1-3	2.10	2.08	2.40	2.73	2.48	2.56
n-SiGe	3-4	9.05	9.42	14.70	15.03	15.34	17.94
n-SiGep-SiMo	3-5	9.85	10.78	16.75	17.19	17.20	20.08
p-SiGep-SiMo	5-6	15.94	9.19 ²	9.66	10.09	10.87	13.08
Total Couple							
a. at temp.	1-6	28.3 ¹	26.82 ²		30.01	30.56	31.86
b. room temp.	1-6	-	11.27		-	_	16.95

Note <u>l</u> assumes 5% contact resistance; based on 1500 hour SiGe material properties

² p-SiGe material used had a lower resistivity than the value in the computer data table

³ severe thermal shock due to power failure

The reason for the high resistance of the n-SiGe--p-SiMo segment was not apparent. The initial "as bonded" room temperature resistance of this segment was 5.7 mohms, and the initial resistance at temperature was 10.78 mohms which was in good agreement with the design data. A resistance increase from 10.78 mohms to 16.75 mohms occurred after a severe thermal cycle caused by a power failure at 226 hours. The total resistance increase for this segment was 9.3 mohms, from 10.78 mohms to 20.08 mohms, while the total couple resistance increase was 5.04 mohms, from 26.82 mohms to 31.86 mohms. It is felt that the discrepancy in the $\triangle R$ of the couple compared to that of the n-segment is due to instrumentation.

Of further note is the temperature profile. (See temperatures indicated in brackets of Figure 56.) Although the test operates at a hot shoe "edge" temperature of 902°C (1656°F), the PbTe hot shoe temperature of 496°C (925°F) when extrapolated to design operating condition would be 526°C (979°F) as compared to 538°C (1000°F) design. Further, the 15°C (27°F) drop from hot shoe corner to hot junction also compares favorably with analysis. Thermally, therefore, the measured data shows good agreement with the design analysis.

Based on the test results of couple B-18, it was decided to further investigate the n-SiGe to p-SiMo segment. These tests were planned to provide some information as to (1) possible changes in the hot junction region, n-SiGe--p-SiMo; (2) possible sublimation effects of PbTe element upon the structure; and (3) a possible reaction of the alumina insulating material with SiGe resulting in the resistance increase observed in life-tested couples.

Accordingly, four specially constructed Hybrid couples, BX-1, -2, -3 and -4, were fabricated and placed on test. The construction details of these four couples are shown in Figures 57 through 60, and a description is given below:

- 1. Couple BX-1 (Figure 57) has a 70 at.% p-SiGe pellet substituted for the SiGe-PbTe n-segment. The insulating alumina (Al₂0₃) material has been eliminated from the inside of the structure.
- 2. Couple BX-2 (Figure 58) has a 70 at.% n-SiGe pellet in the inside. The PbTe n-pellet has been eliminated as well as the insulating Al₂O₃.
- 3. Couple BX-3 (Figure 59) is similar to BX-2 except that insulating alumina (Al₂O₃) powder has been included.
- 4. Couple BX-4 (Figure 60) is identical to the standard Reference Design B structure except that the insulating alumina (Al₂0₃) powder has been omitted.

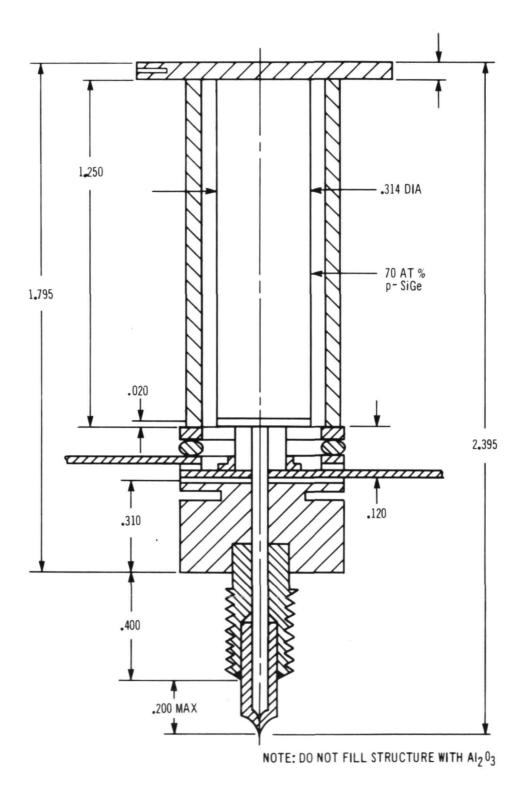


Figure 57. Hybrid Thermo coup le Assembly - Design BX - 1

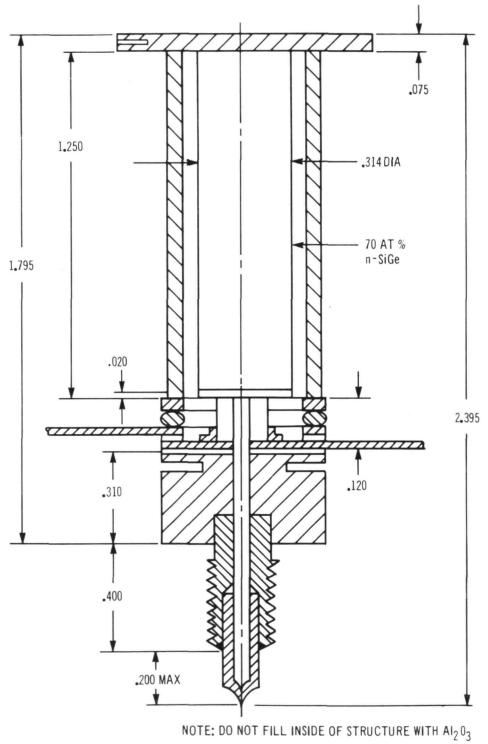
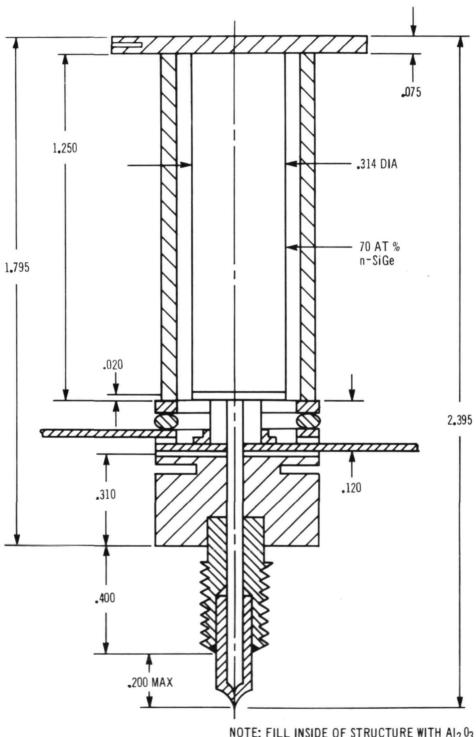


Figure 58. Hybrid Thermocouple Assembly - Design $\,\mathrm{BX}-2\,$



NOTE: FILL INSIDE OF STRUCTURE WITH $\mathrm{AI}_2\,\mathrm{O}_3$

Figure 59. Hybrid Thermocouple Assembly - Design $\,$ BX - 3

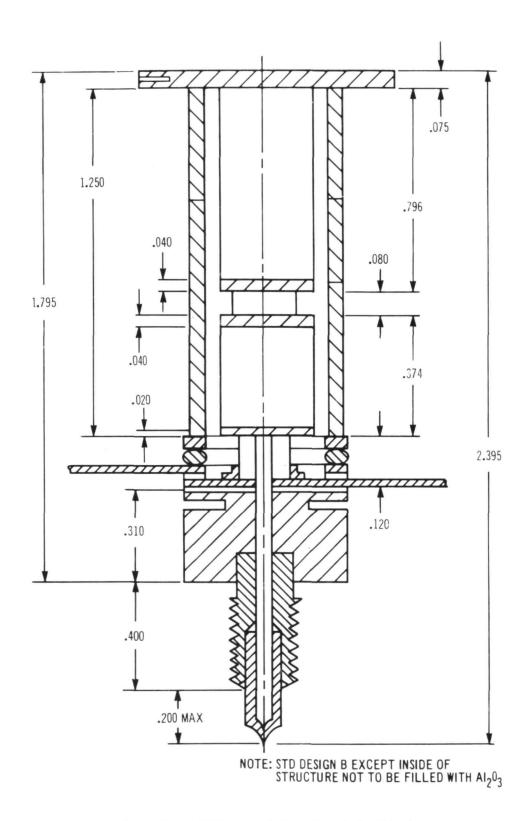


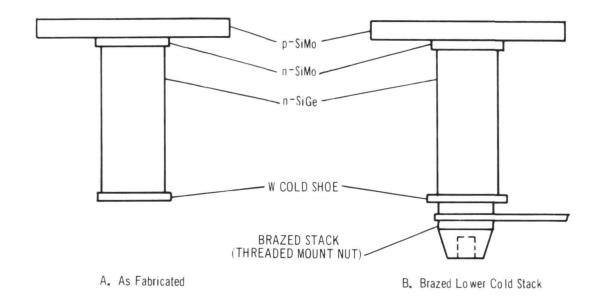
Figure 60. Hybrid Thermocouple Assembly – Design BX – 4

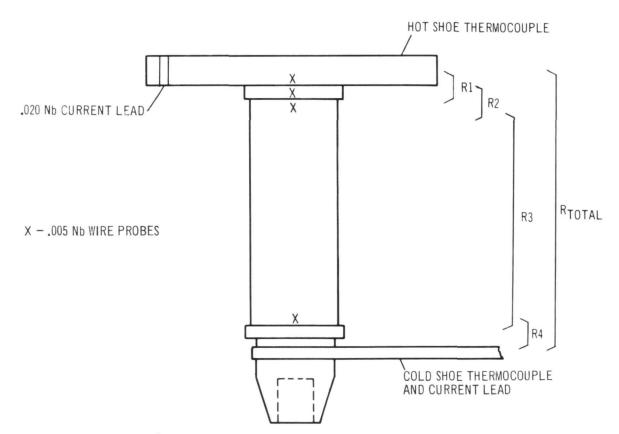
These couples were tested for a total of 2328 hours at design test conditions. Couple BX-4, a standard Hybrid couple without alumina powder insulation fill (see Figure 60) exhibited a resistance pattern similar to previously tested Hybrid couples, 29.46 mohms at zero hours increasing to 33.94 mohms after 2805 hours. Examination of the data of couples BX-1, -2 and -3 after 2328 hours showed that the resistance of the p-SiGe leg remained essentially constant with time and that the increase in resistance was primarily due to the n-SiGe segment, as noted in couples BX-2 and -3.

Based on these test results, it was decided to stop fabrication of the Hybrid couples, Series G, for the Hybrid panels, and return to the conventional air-vac hot shoe bond, p-SiMo--Ti-barrier layer--n-SiMo--n-SiGe (process D, Figure 51) rather than the modified air-vac bond, p-SiMo--Ti-barrier layer--n-SiGe (process C, Figure 50) in the fabrication of the Hybrid couples. This is fully discussed in Section IV-B-1 and involves an increase in thickness of both the p-SiMo heat receptor plate (from 0.19 cm, 0.075 in., to 0.318 cm, 0.125 in.), and the n-SiMo disc (from 0.051 cm, 0.020 in., to 0.102 cm, 0.040 in.). A number of special test samples were fabricated, employing the conventional air-vac bond system but with the thicker p-SiMo and n-SiMo heat receptor. Also, hot pressed versus vacuum cast n-SiMo was investigated.

In order to perform ingradient testing of these special test specimens, it was necessary to attach a cold stack to the specimen as shown in Figures 6la and b. A description of the specimens is given below.

No. of Test Samples		Fabrication Procedure		Temperature Resistance
4	Step 1:	p-SiMo, 0.125 in. thickness (vacuum cast material) - barrier - n-SiMo, 0.040 in. thickness (vacuum cast material)	Avg.	0.10 mohms
	Step 2:	Subassembly 1 + n-SiGe and tungsten cold shoe	Avg.	0.08 mohms
4	Step 1:	p-SiMo, 0.125 in. thickness (vacuum cast material) - barrier - n-SiMo, 0.040 in. thickness (hot pressed material)	Avg.	0.10 mohms
	Step 2:	Subassembly 1 - n-SiGe and tungsten cold shoe	Avg.	0.08 mohms





C. Instrumentation Arrangement - Special Test Samples

Figure 61. Special Test Samples

The bond resistances of the special test samples "as fabricated" and after being brazed with the lower stack structure are presented in Table XX. No significant changes occurred in any of the bond resistances as a result of this bonding operation.

TABLE XX
BOND RESISTANCES OF SPECIAL TEST SAMPLES

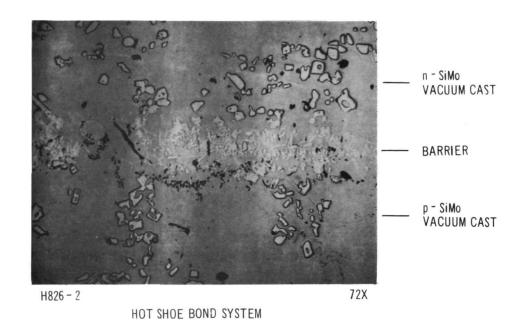
	After Bonding					Aft	er Bra	zing L	ower S	tack	
Sample	No.	R ₁	R ₂	R ₃	R_4	R_{total}	_R ₁ _	R ₂	R ₃	R ₄	R_{total}
				(mohms)				(mohms		
1	(a)	0.20	0.15	3.7	0.75	5.1	0.20	0.23	4.1	0.26	5.0
3	(a)	0.20	0.18	4.1	0.18	5.7	0.24	0.23	4.2	0.27	5.1
4A	(b)	0.20	0.20	4.3	0.15	5.2	0.30	0.25	4.4	0.24	5.3
4B	(b)	0.25	0.28	5.7	0.15	6.3	0.30	0.47	5.21	0.20	6.3
$R_1 = p-SiMon-SiMo$											
			R_2	= n-	SiMo	n-SiGe					
			R_3	= n-	SiGe						
			R_4	= n-	SiGe	tungsten	cold s	hoe			
			Rtota	_ = p_	SiMo	tungsten	cold s	hoe			

Notes:

- (a) vacuum-cast n-SiMo
- (b) hot pressed n=SiMo

Photomicrographs of the bonds for the special test samples made with (a) vacuum-cast n-SiMo and (b) hot pressed n-SiMo, are shown, respectively, in Figures 62 and 63. All bonds were sound and are representative of typical air-vac bonds.

The instrumentation and wiring arrangement of the special test samples are given in Figure 64.



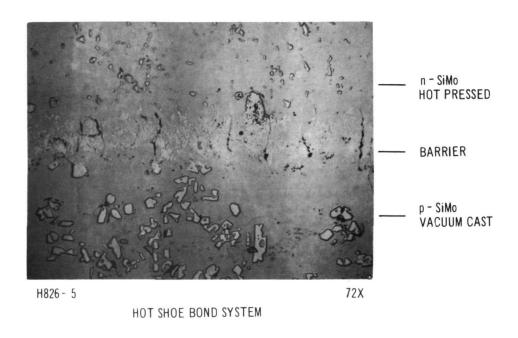
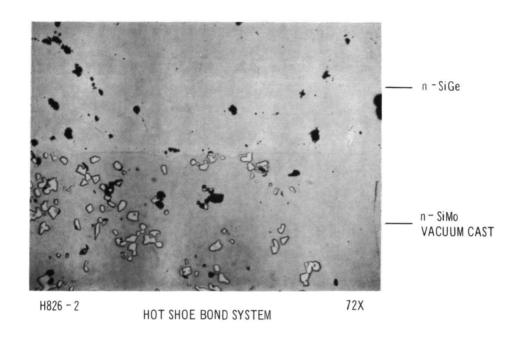


Figure 62. Test Coupon Bond Systems



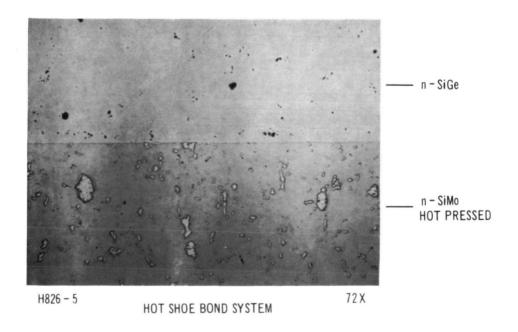


Figure 63. Test Coupon Bond Systems

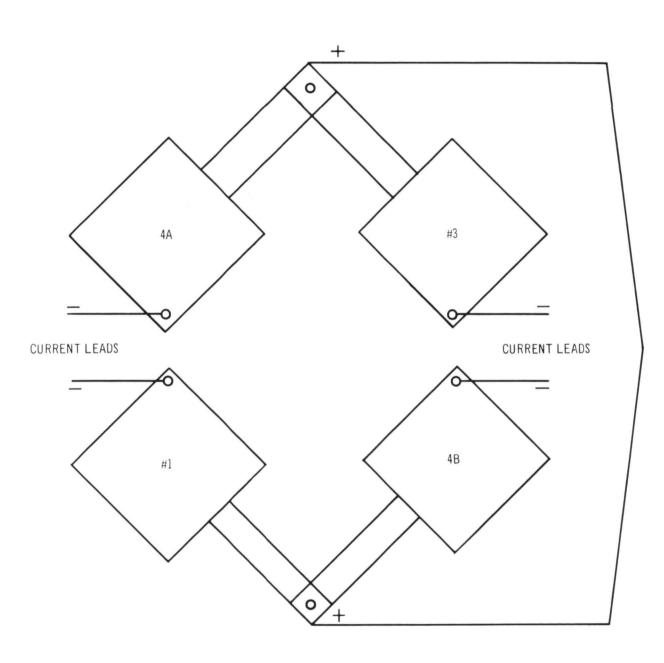


Figure 64. Wiring Arrangement-Special Test Samples

The room temperature and the high temperature bond resistances of the four special test samples, as measured in the life test station up to 381 hours, are given in Table XXI. The various bond resistances appear normal for air-vac couples; no abnormal increase in the resistance of either the n-SiGe or various bonds has occurred.

A leak occurred in the life test station in which these samples were being tested, and it was necessary to transfer them to another life test station. Room temperature resistance readings, taken in the test station after cycling to room temperature and in the new station prior to high temperature operation, were normal. The increase in the n-SiGe resistance with time is normal and is due to a temperature dependence of the solid solubility of the dopant. The higher the silicon content of the alloy and the higher the temperature of operation, the faster equilibrium is attained. Generally, the majority of the change occurs below 600°C. The contact resistance values always contain a small amount of the bulk resistance of the SiMo or SiGe as the attachment of the wires cannot be made closer than 0.010-0.015 in. to the joint. The contact resistances appear normal and show only minimal changes after test times in excess of 700 hours.

The testing of the four special test samples was interrupted due to a plant-wide power failure. With the loss of vacuum, the instrumentation leads were severely oxidized, necessitating their removal from the life test station. Visual inspection of the samples showed no damage or bond separation resulted after this thermal shock. Room temperature resistance readings of the bonds and bulk material were normal.

Prior to the power failure, the four test samples had completed 700 hours at design temperature conditions and various bond resistances were normal (see Table XXI). The data in Table XXI show the hot and cold shoe bonds to be quite stable. These values are listed under R_1 , R_2 (hot end) and R_3 (cold end) and show only minimal changes to date. Two of the four samples, No. 1 and 4A, were reinstrumented and placed back on test and completed 3971 hours with stable performance.

As a result of all the testing conducted in the development of the thermocouple structure and more specifically the testing of the special test samples above, the conventional air-vac bond - p-SiMo--Ti-barrier--n-SiMo--n-SiGe - was selected as the bond system for the n-SiGe segment of the Hybrid couples that were fabricated for the two flat plate module test panels. Assembly process D, Figure 51, was the fabrication procedure used to produce the Hybrid thermocouple for the test panels.

TABLE XXI
BOND RESISTANCES OF SPECIAL TEST SAMPLES ON LIFE TEST

			Sampl	e No. 1					
Hours Hot Shoe Temp. Cold Shoe Temp.	(°C)	25 -	0 936 419	50 926 420	212 930 420	381 934 419	548 933 418	700 930 418	726
R ₁ (m R ₂ R ₃ R ₄ R _{total}	ohms) " " " "	.22 .25 4.29 .26 5.01	.37 .35 6.40 .28 7.38	.84 .66 12.46 .61 14.60	1.01 .53 13.05 .65 15.22	1.00 .51 13.12 .65 15.32	1.01 .54 13.32 .68 15.53	.98 .50 13.27 .67 15.42	Power Failure 7/1/71
			Sampl	e No. 3					
Hours Hot Shoe Temp. Cold Shoe Temp.	(°C)	25 -	0 939 419	50 930 417	212 934 416	381 937 416	548 939 415	700 935 414	726
R ₁ (m R ₂ R ₃ R ₄ Rtotal	ohms) " " " "	.25 .24 4.32 .27 5.08	.48 .34 7.48 .24 8.67	.66 .66 12.66 .61 14.58	.72 .64 13.08 .65 15.06	.72 .63 13.10 .64 15.14	.75 .64 13.34 .68 15.41	.72 .63 13.27 .66 15.33	Power Failure 7/1/71
			Sampl	e No. 4	<u>A</u>				
Hours Hot Shoe Temp. Cold Shoe Temp.	(°C)	25 -	0 928 419	50 918 424	212 922 424	381 925 424	548 927 415	700 923 422	726
R ₁ (m R ₂ R ₃ R ₄ R _{total}	ohms)	.22 .27 4.49 .25 5.33	.70 .65 12.28 .56 14.3	.84 .53 13.23 .46 15.06	.90 .53 13.58 .50 15.49	.94 .48 13.63 .50 15.58	.95 .49 13.83 .52 15.79	.94 .48 13.78 .52 15.72	Power Failure 7/1/71
			Sampl	e No. 4	B				
Hours Hot Shoe Temp. Cold Shoe Temp.	(°C)	25 -	934 415	50 925 419	212 928 419	381 932 414	548 933 413	700 930 412	726
R ₁ (m R ₂ R ₃ R ₄ Rtotal	ohms) " " " "	.32 .50 5.46 .22 6.48	.38 .52 8.38 .22 9.35	.70 .91 15.18 .44 17.23	.77 .82 15.59 .49 17.66	.77 .70 15.70 .47 17.68	.81 .71 15.90 .51 17.90	.78 .71 15.80 .48 17.83	Power Failure 7/1/71

Section V

FABRICATION OF MODULE PANELS

Two 9-couple panel sections of the Reference Design Hybrid module were fabricated and delivered to NASA Lewis for their testing and evaluation. This section describes the fabrication procedures used in making the panels. Detailed specifications for materials, parts and assembly processes were furnished to NASA Lewis and a list of these specifications is given in Tables XXII and XXIII. The assembly procedures developed in the thermocouple development task, Section IV-B, Sequence D, Fig. 51, were used to fabricate all couples for the panels. A detailed description of these assembly procedures is given at the end of this section.

The Hybrid couple components and a couple are illustrated in Figures 65 and 66, respectively. The Hybrid panel construction details are shown in Figures 67 and 68.

Photographs of the assembly sequences to fabricate the Hybrid panel are shown in Figures 69, 70 and 71. The heat receptor thermocouple location and power leads location are shown in Figure 72. The cold junction thermocouples, voltage taps and current leads location are shown in Figure 73. The couple location and individual couple room temperature resistance are presented in Figures 74 and 75 for Hybrid panels number 1 and 2, respectively. In both panels, the sum of the individual couple resistances were well within the required specification of 5% of the total couple resistance. The sum of the individual resistances versus the total for the two panels is given below.

Panel Number	Sum of Resistances	Total Couple Resistance
1 2	96.7 95.5	96.2 96.5

The couple leak rates for the couples used in the panel are given below.

P	anel Number 1	Pa	anel Number 2
Couple	Leak Rate	Couple	Leak Rate
Number	Std. cc Helium/sec	Number	Std. cc Helium/sec
8	4.0×10^{-8}	5	4.0×10^{-8}
10	4.0×10^{-8}	13	>1.0 x 10 ⁻⁵
11	4.2×10^{-8}	21	1.0×10^{-6}
17	4.4×10^{-8}	25	>1.0 x 10 ⁻⁵
18 20	4.0×10^{-6}	26	1.3×10^{-7}
	2.0×10^{-8}	28	1.0×10^{-8}
22	4.2×10^{-8}	30	1.0×10^{-6}
24	2.0×10^{-8}	36	1.0×10^{-6}
27	6.0×10^{-8}	37	1.0×10^{-8}

TABLE XXII

HYBRID THERMOCOUPLE MATERIAL AND PART SPECIFICATIONS

	Title	Material		Specification Number
1. 2. 3. 4. 5. 6. 7. 8.	Heat Receptor Hot Shoe Cylindrical Leg Segment (SiGe) Cold Shoe Hot Shoe Segment (PbTe) Cold Shoe	p-Type SiMo n-Type SiMo p-Type SiGe n-Type SiGe Tungsten 303 Stainless PbTe 303 Stainless		HTP 171-A5A HTP 10-J15D HTP 170-B3A HTP 10-I16D HTP 8-B17F HTP 4-D22K HTP 10-H13D HTP 4-D22K
9. 10. 11. 12. 13. 14. 15. 16. 17. 18.	Compensator Top Compensator Bottom Cold Shoe Ring Collar p-Type Elec. Connector n-Type Elec. Connector Insulator Insulator Mount Stud Pad Mount Stud Screw Exhaust Tubulation	Gold Gold Tungsten Copper Copper Copper Al ₂ O ₃ Al ₂ O ₃ Copper C.R.S. Copper		HTP 9-A23C HTP 172-A4A HTP 78-C5E HTP 148-A3A HTP 99-B42L HTP 99-B41L HTP 78-B3D HTP 78-B4D HTP 140-P20G HTP 154-D9E HTP 169-A2A
20. 21. 22. 23. 24. 25. 26. 27. 28. 29. 30. 31. 32. 33.	Standard 1/4" Washer Standard 1/4" x 28" Nut Braze Shims Braze Shims Braze Shims Braze Shims Brazing Ring Brazing Ring Brazing Disc (Compensator) Compensating Ring Insulation Block Mounting Plate Heat Transfer Plate Shoulder Screw Cooling Plate Connector Screw	Gold Min-K 2020 Aluminum Aluminum 303 Stainless Aluminum C.R.S.	Steel	- HTP 100-B5B HTP 100-B6B HTP 100-E6B HTP 55-D9C HTP 46-D10B HTP 46-D8B HTP 100-E7B HTP 175-A1A HTP 152-B44A HTP 159-B3A HTP 159-B4A HTP 178-A1A HTP 178-A1A HTP 159-B5A HTP 154-G25C

TABLE XXIII HYBRID THERMOCOUPLE ASSEMBLY SPECIFICATIONS

	Title	Specification Number
1.	Heat Receptor Assembly	HTS-1
2.	SiGe Assembly	HTS-2
3.	PbTe Element Assembly	HTS-3
4.	Thermocouple Assembly	HTS-4
5.	Cold Stack Assembly	HTS-5
6.	Hybrid Thermocouple Final Assembly	HTS-6
7.	Module Panel Assembly	HTS-7

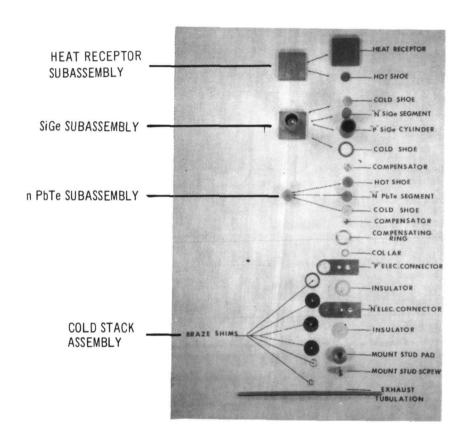


Figure 65. Hybrid Thermocouple Components

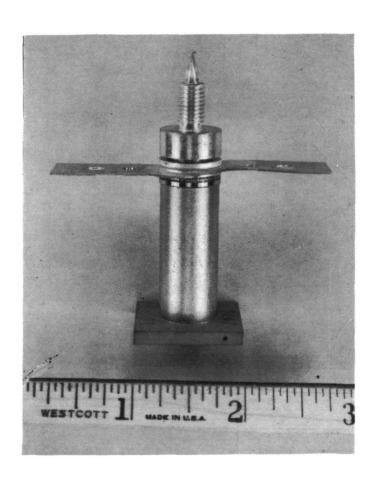


Figure 66. Reference Design Hybrid Couple

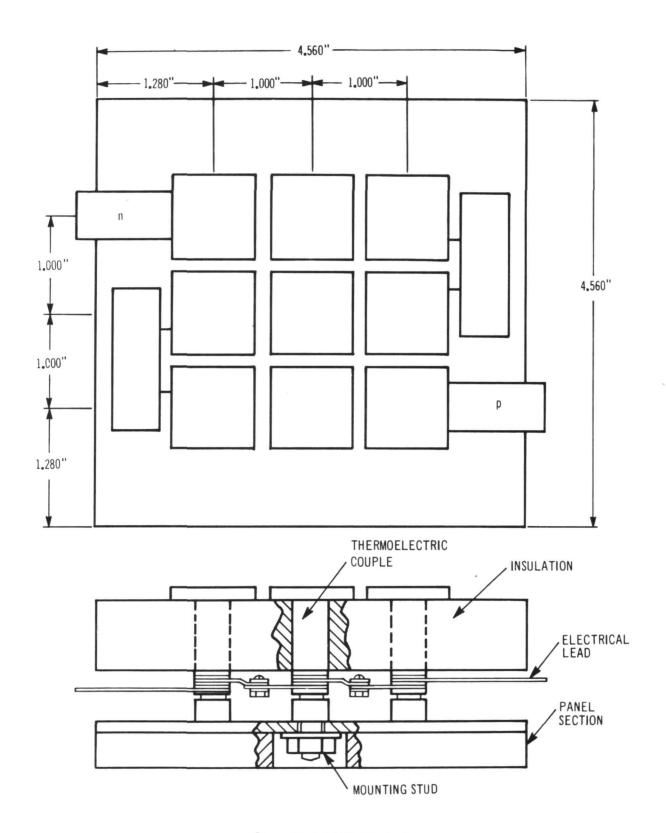


Figure 67. Hybrid Module Panel

Figure 68. Hybrid Structure Module Panel Assembly

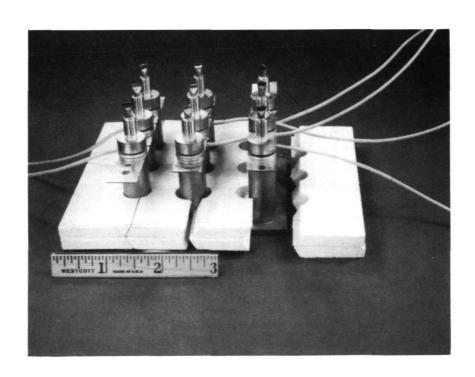


Figure 69. Hybrid Module Panel Strips

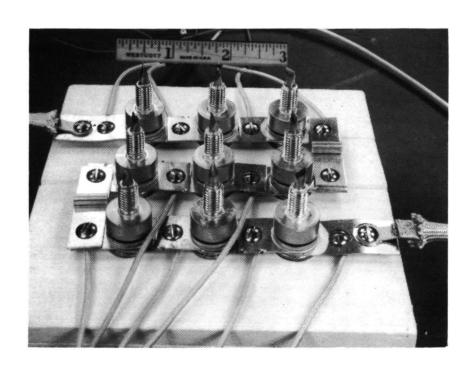
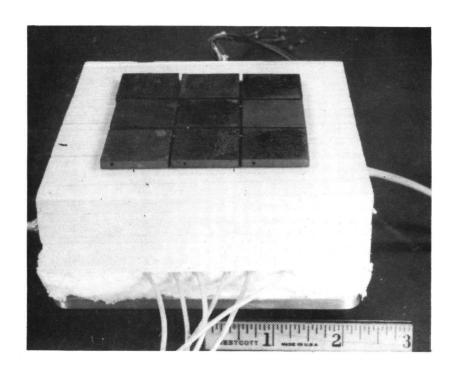


Figure 70. Hybrid Module Strips Instrumented - Cold Shoe Thermocouples



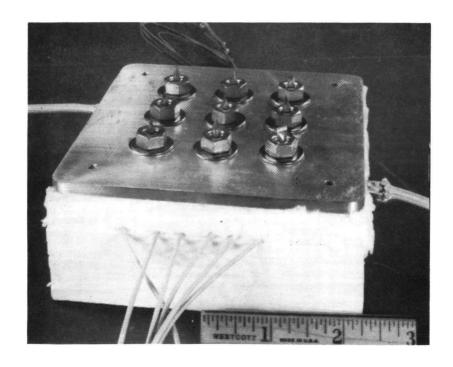


Figure 71. Completed Hybrid ${f F}$ lat Plate Panel Section

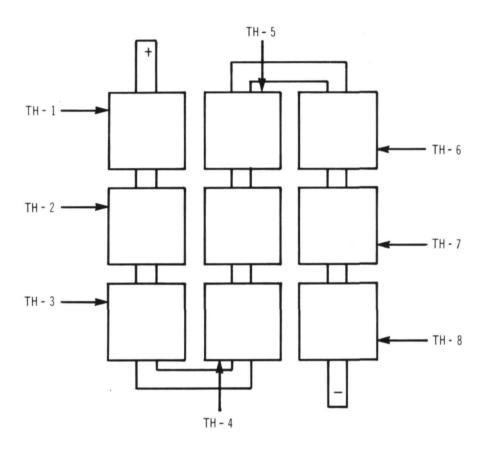


Figure 72. Heat Receptor Thermocouple Locations and Power Leads Locations, (W3Re/W25Re Thermocouples used)

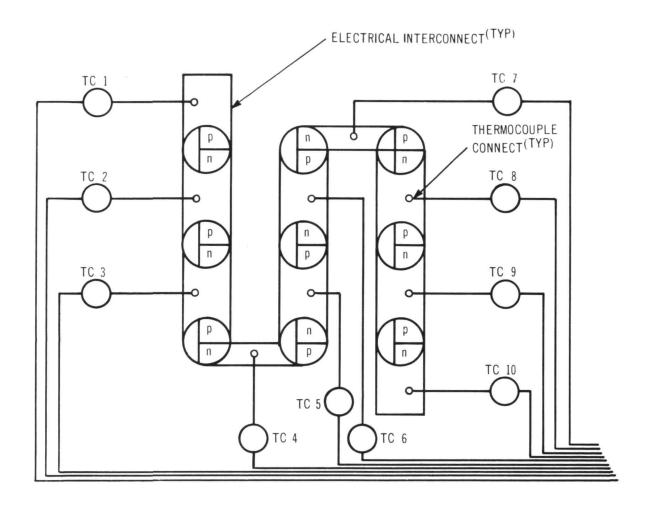
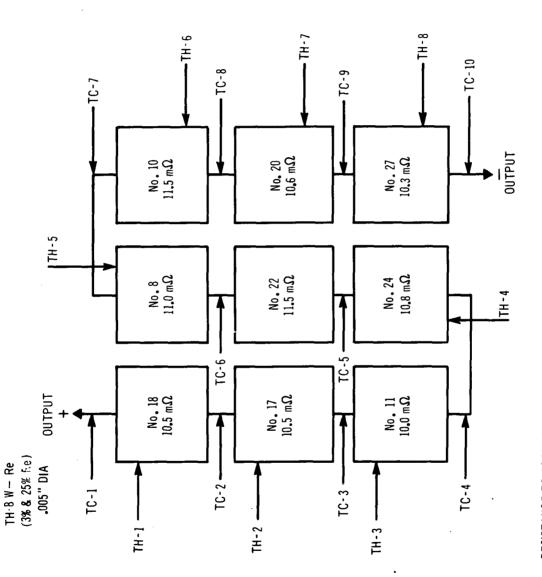


Figure 73. C/A Cold Junction T/C, Voltage Taps and Current Lead Location



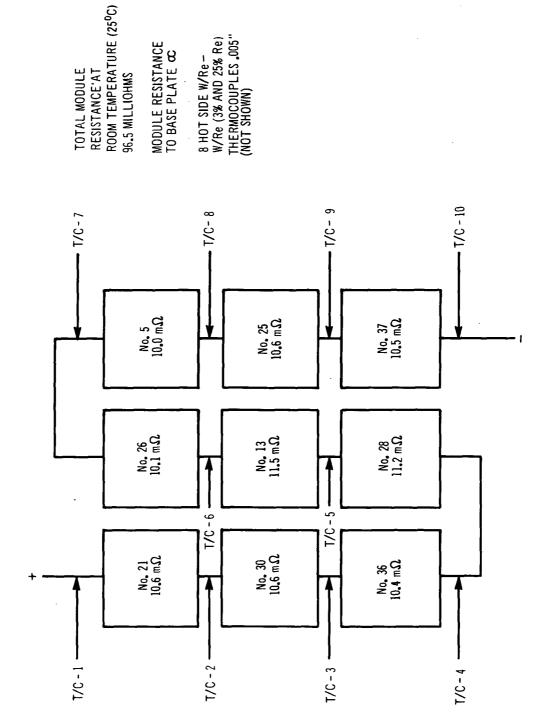
TC-1 - C/A

TC-10-

TH-] ₩ - Re

RESISTANCE TO GROUND CC (MODULE RESISTANCE AT 78° F 96.2 MILLIOHMS) VIEW: HOT SHOE SIDE OF MODULE

Figure 74. No.1 Panel Resistance Data



10 COLD SIDE C/A THERMOCOUPLES .005" DIA WIRE VIEW:HOT SHOE SIDE OF MODULE

A. Hybrid Couple and Panel Assembly Processes

- 1. HEAT RECEPTOR ASSEMBLY HTS-1A1A
- 1.0.0 <u>Heat Receptor Parts Assembly</u>

1.1.0 Material

Polarity Type	Composition	Dopant	Fabrication <u>Method</u>	
n p	85 wt.% Si; 15 wt.% Mo 85 wt.% Si; 15 wt.% Mo	1% Phosphorus	Hot pressed	

1.2.0 Equipment

- 1.2.1 Element alignment fixture HST-4
- 1.2.2 Small tools: tweezers, dropper, tungsten weight

1.3.0 Procedure

- 1.3.1 Slice 4.45 cm (1.75 in.) diameter p-SiMo casting to 2.44 cm (0.960 in.) x 2.44 cm (0.960 in.) x 0.318 cm (0.125 in.) thick heat receptor plates and the 5.08 cm (2.0 in.) diameter n-SiMo hot pressed disc to 0.813 cm (0.320 in.) diameter using Micromesh slicing equipment and metal bonded diameter wheels.
- 1.3.2 Measure and record the individual slice density using the water immersion technique. The material immersed in water is buoyed up with a force equal to the weight of the fluid displaced. Because the density of water at room temperature is one gram per cubic centimeter, the difference between the weight of the slice in air and in water represents its volume in cubic centimeters. Therefore:

Density
$$(g/cm^3) = \frac{Wt (gms) \text{ in air}}{Wt (gms) \text{ in air - Wt (gms) in water}}$$

Match n-type and p-type slices according to density.

Polish one surface of each slice on a Brinkman metallurgical polishing table using silk cloth with Linde "A" abrasive followed by Rayvel synthetic velvet cloth with Linde "A" abrasive. Wash in methanol; dry with an air blower.

1.3.3 Place the p-SiMo heat receptor on a ceramic plate with the smooth or ground side of the heat receptor facing up. Then place the alignment fixture over the heat receptor making

sure that the heat receptor fits snugly into the recess of the fixture. Apply one small drop of lucite cement onto the surface of the heat receptor at the hole in the boron nitride fixture, then immediately insert a titanium bonding shim into the hole and press firmly onto the heat receptor with tweezers. After a moment apply another drop of lucite to the surface of the bonding shim and apply an n-SiMo hot shoe and press the complete assembly with tweezers.

- 1.3.4 With pressure applied to the heat receptor assembly (tweezers or pick), remove alignment fixture.
- 1.3.5 Place the assembly into an air oven (approximate temperature 100°C) for drying for 3-5 minutes.
- 1.3.6 Inspect alignment. Titanium shim and n-SiMo must be located centrally on the p-SiMo heat receptor.

2.0.0 Heat Receptor Bonding

2.1.0 Equipment

- 2.1.1 Brew Furnace, Model 300MC or equivalent
- 2.1.2 Furnace rack
- 2.1.3 Ceramic spacers, tungsten weight (300 gms total)

2.2.0 Procedure

- 2.2.1 Determine temperature profile of furnace unit.
- 2.2.2 Load work onto the rack. Place a dummy n-SiMo hot shoe on the hot shoe being bonded, then position tungsten weight onto the dummy hot shoe. The tungsten weight must be supported to prevent shifting of the assembly.
- 2.2.3 Insert the work rack into the furnace, being certain that it is fully seated. Replace the furnace cover.
- 2.2.4 Evacuate the furnace to a pressure not to exceed 1×10^{-4} Torr.
- 2.2.5 Metallurgically bond assembly.

3.0.0 Inspection and Measurement

- 3.1.0 Lightly sandblast the bonded assembly after removing from the furnace. The bond is inspected under 30X magnification.
- 3.1.1 Bond resistance is measured using a Keithly milliohmeter.

2. SiGe ASSEMBLY - HTS-2AlA

1.0.0 SiGe Stage Assembly

1.1.0 Material

1.1.1 SiGe Alloys

Polarity Type	Composition	Dopant	Fabrication <u>Method</u>	
n	70 at.% Si; 30 at.% Ge		Zone leveled	
p	80 at.% Si; 20 at.% Ge		Zone leveled	

1.1.2 Cold Shoes

Tungsten Bar Stock - density: 18.9 gm/cm³

1.1.3 Heat Receptor Assembly HTS-1A1A

1.2.0 Equipment

- 1.2.1 Alignment Fixture HST-7
- 1.2.2 Small tools: tweezers, dropper, clips, etc.

1.3.0 Procedure

- 1.3.1 Slice the n-ingot on the Micromesh slicing equipment, using a metal-bonded diamond wheel. Cylindrically grind the n-type SiGe elements with the proper radius to 0.798 cm (0.314:in.) diameter, using an 80-grit silicon carbide resin bonded wheel. End grind the n-type SiGe element to 1.82 cm (0.716 in.) length on the Sanford surface grinder, using an 80-grit silicon carbide resin-bonded wheel. Microfinish is not critical to subsequent processing. Weigh and inspect. Wash in 50% methanl-50% acetone. Package and identify individual pellets as to ingot number and position.
- 1.3.2 Slice the p-ingot on the Micromesh slicing equipment, using a metal-bonded diamond wheel, into blanks 1.52 cm; (0.600 in.) diameter x 3.81 cm (1.500 in.) length. Send blanks to vendor to be diamond core drilled to finished dimensions, 1.34 cm (0.527 in.) 0D x 1.1 cm (0.435 in.) ID x 3.17 cm (1.250 in.) length. Each machined p-SiGe cylinder is then checked for hermeticity, using a Veeco MS-9 leak detector and a dye penetrant leak test. Weigh and perform visual inspection. Wash in 50% methanol-50% acetone. Package and identify individual cylinders as to ingot number and position.

- 1.3.3 Fabricate n-SiGe cold shoe to a disc, 0.81 cm (0.320 in.) diameter x 0.102 cm (0.040 in.) thick. Fabricate p-SiGe cold shoe ring, 1.4 cm (0.550 in.) OD x 1.08 cm (0.425 in.) ID x 0.102 cm (0.040 in.) thick via Elox drilling technique (shoe fabricated by outside vendor). All shoes for the n- and p-type SiGe legs are to be rigidly inspected for dimensions and flatness, then washed in hot Blacosolv, hot water, and methanol to ensure a clean surface for bonding. The tungsten cold shoes are to be stored in clean, properly identified envelopes.
- 1.3.4 Place a heat receptor assembly on a ceramic pad with the hot shoe facing up.
- 1.3.5 Assemble n-SiGe pellet to heat receptor assembly. Use one small drop of lucite cement. Assemble a tungsten cold shoe to end of n-SiGe pellet using one small drop of lucite cement. Dry assembly with hot air gun.
- 1.3.6 To the above assembly, place the alignment fixture over the heat receptor making sure the heat receptor fits snugly into the recess of the fixture; the n-SiGe pellet facing up.
- 1.3.7 Apply two drops of lucite cement on the heat receptor at the recess of the fixture. Immediately place a p-SiGe cylinder over the n-leg and into the recess of the alignment fixture. Press slightly to squeeze the lucite to thinnest possible layer.
- 1.3.8 Apply lucite cement to the upper edge of the p-SiGe cylinder; Then place a tungsten ring on the edge and align the ring while pressing down.
- 1.3.9 To speed drying, place under lamp or in oven (maximum temperature 100°C). Allow to dry for approximately 5 minutes.
- 1.4.0 Remove alignment fixture. Inspect alignment.

2.0.0 SiGe Stage Bonding

2.1.0 Equipment

- 2.1.1 Brew Furnace, Model 300 MC or equivalent
- 2.1.2 Work rack
- 2.1.3 Ceramic spacers, tungsten weights

2.2.0 Procedure

- 2.2.1 Determine temperature profile of furnace unit.
- 2.2.2 Place tungsten weight (cylindrical) onto the tungsten cold shoe

ring of the p-leg. Insert tungsten rod into the cylinder weight so that it rests on the tungsten cold shoe of the n-leg.

- 2.2.3 Load the above assembly onto the rack so that no shifting occurs.
- 2.2.4 Insert the work rack into the furnace, being certain it is fully seated. Replace furnace cover.
- 2.2.5 Evacuate the furnace to a pressure not to exceed 1 x 10^{-4} Torr.
- 2.2.6 Metallurgically bond assembly.

3.0.0 Inspection and Measurement

- 3.1.0 After removing the bonded SiGe stage from the furnace, all bonds are inspected under 30X magnification.
- 3.1.1 Bond resistances are measured on a Keithley milliohmeter.

3. n-TYPE PbTe ASSEMBLY - HTS-3AlA

1.0.0 N-Type PbTe Parts Assembly

1.1.0 Material

3M Co. 3N PbTe pellet 303 stainless steel hot and cold shoes

1.2.0 Equipment

- 1.2.1 Commercial chemical vapor deposition equipment
- 1.2.2 Split bonding die

1.3.0 Procedure

- 1.3.1 The 303 stainless steel alloy hot shoe, 0.953 cm (0.375 in.) diameter x 0.102 cm (0.040 in.) thick, and cold shoes, 0.953 cm (0.375 in.) diameter x 0.051 cm (0.020 in.) thick, are fabricated by punching from stock strip. They are then coated with a tungsten diffusion barrier layer and bonding layer.
- 1.3.2 Coat n-PbTe segment TP10H13D on cylindrical surface with Aquadag.
- 1.3.3 Place hot and cold shoes TP4D22K and TP4D23K with W-plated surface toward PbTe segment TP1OH13D.

1.3.4 Align unbonded element in split bonding die with bond interfaces in line with bond overflow relief channels.

2.0.0 n-Type PbTe Element Bonding

2.1.0 Equipment

2.1.1 Vacuum Furnace

2.2.0 Procedure

- 2.2.1 Insert loaded die assembly in bonding furnace. Evacuate chamber to 1×10^{-5} Torr. Admit 95 Ar-5 H gas. Maintain positive pressure.
- 2.2.2 Metallurgically bond.
- 2.2.3 Remove bonded element from die. Sandblast to remove Aquadag and bond overflow.
- 2.2.4 Record T_1 , T_2 , T_3 (hot, cold contact and total element resistance).
- 2.2.5 Measure and record length.
- 2.2.6 Surface grind both shoes.
- 2.2.7 Nickel plate both shoes 0.0001. Fire 575°C 30 minutes in dry H2.
- 2.2.8 Surface polish both shoes to mirror finish with Linde "A" immediately prior to couple assembly.

4. THERMOCOUPLE ASSEMBLY - HTS-4AlA

1.0.0 Thermocouple Parts Preparation

1.1.0 Materials, Components

- 1.1.1 SiGe Assembly, HTS-2A1A
- 1.1.2 n-PbTe Assembly, HTS-3AlA
- 1.1.3 Upper and lower gold compensator

1.2.0 Procedure

- 1.2.1 Sandblast the face of the TP8B17E tungsten shoe and the edge of the p-cylinder of the SiGe subassembly.
- 1.2.2 Wash subassembly ultrasonically in 50% methanol-50% acetone for 5 minutes.

- 1.2.3 Nickel plate and sinter the surface of the tungsten cold shoe and edge of the p-tungsten cold shoe ring.
- 1.2.4 Ultrasonically wash upper and lower gold compensators in Blacosolv for 5 minutes. Rinse in distilled water and then methanol.

2.0.0 Bonding Operation

2.1.0 Equipment

- 2.1.1 Diffusion bonding fixture (CST-40) 2.1.2 316 SS pressure pads
- 2.1.3 Molybdenum pad
- 2.1.4 303 SS shim 2.1.5 Nichrome shim 2.1.6 TC structure
- 2.1.7 316 SS shim

2.2.0 Bonding Procedure

- 2.2.1 Glue the top compensator gold disc to the nickel-plated hot shoe of the PbTe element with lucite cement. Center the two with respect to each other using the locating sleeve "A".
- 2.2.2 Drop the glued subassembly into the cylinder with the gold top compensator against the TP8B17E nickel-coated shoe and secure with lucite. Use sleeve "B" for centering. Load into diffusion bonding fixture.
- 2.2.3 Apply 30 psi 0.05 mil deflection of pressure to the subassembly.
- 2.2.4 Metallurgically diffusion bond.
- 2.2.5 Remove from furnace and check for bond resistance.
- 2.2.6 Glue the slotted side of the bottom gold compensator (TP172A4A) to the cold shoe (TP4D22K) of the PbTe element with lucite cement. Center the two using locating sleeve "B".
- 2.2.7 Assemble into bonding clamp CST40 and repeat procedure 2.1.0 through 2.2.4.

3.0.0 <u>Inspection and Measurement</u>

3.1.0 Remove from furnace and check for n- and p-joint resistances and n- and p-element resistances after each bonding cycle.

5. COLD STACK ASSEMBLY - HTS-5AlA

1.0.0 Cold Stack Assembly (HTS-5)

1.1.0 Equipment

- 1.1.1 Fixture Assembly (HST-5)
- 1.1.2 Carbon block with boron nitride insert
- 1.1.3 Nichrome pin coated with boron nitride 1.1.4 Ceramic ring
- 1.1.5 Tungsten wire clips

1.2.0 Cold Stack Assembly Procedure

1.2.1 Load the following component parts in the prescribed order onto the boron nitride insert of the jig assembly:

TP148A3A	copper collar
TP99B42L	p-type electrical copper connector
TP100B6B	braze shim
TP78B4B	ceramic insulator
TP100B5B	braze shims

1.2.2 Load the following onto the nichrome pin set into the jig subassembly in the prescribed order:

TP99B41L	n-type electrical copper connector
TP100B5B	braze shims
TP78B3B	ceramic insulator
TP100B5B	braze shims
TP140D206	copper mount stud pad

1.2.3 Load the following into the TP140D2OG mount stud pad:

TP46D10B	brazing ring	
TP154D9E	CRS mount stud	screw
TP46D8B	brazing ring	
TP169A2A	copper exhaust	tubulation

- 1.2.4 Slide the ceramic ring of the jig assembly over the mount stud screw onto the mount stud pad.
- 1.2.5 Use four tungsten wire clips to secure the assembly by clipping to the ceramic and carbon block.
- 2.0.0 Metallurgically braze assembly.

3.0.0 <u>Inspection and Measurement</u>

3.1.0 Check the assembly for continuity to determine that no short exists between the n- and the p-type electrical connectors.

3.1.1 Check assembly for leaks on the Veeco. Cap across the cold shoe ring and check for leaks from outside to the tubulated region.

Make sure that tubulation is not blocked. If blocked, open with a drill.

6. FINAL ASSEMBLY - HTS-6AlA

1.0.0 Final Parts Assembly

1.1.0 Materials, Components

- 1.1.1 Thermocouple Assembly HTS-4AlA
- 1.1.2 Cold Stack Assembly HTS-5AlA
- 1.1.3 Gold Compensator Ring

1.2.0 Procedure

- 1.2.1 Fill the area between the inner wall of the p-cylinder and internal parts with powdered alumina (Al203) to the level of the p-tungsten cold ring. Pack Microquartz fibers on top of the alumina to secure it and prevent spillage.
- 1.2.2 Place brazing shims (TP100E7B) inside the collar (TP148A3A).
- 1.2.3 Place brazing shims (TP100E6B) onto the p-type electrical connector of the Cold Stack Assembly and secure with lucite cement.
- 1.2.4 Place the gold compensating ring (TP175AlA) on top of the brazing shims.
- 1.2.5 Place an additional brazing shim (TP100E6B) on top of the gold compensating ring and secure with lucite cement.
- 1.2.6 Assemble the Thermocouple Subassembly (HST-8) onto the Cold Stack Assembly (HTS-7) being careful that the bottom gold compensator fits into the collar brazed to the n-electrical connector. Align the brazing rings with the tungsten cold shoe ring.
- 1.2.7 Secure both assemblies with four tungsten wire clips.
- 1.2.8 Inspect the assembly for alignment of hot shoe, location of thermocouple holes with respect to the n- and p-electrical connectors.

2.0.0 Bonding Operation

2.1.0 Metallurgically braze the assembly in a vertical position with the cold stack on top.

3.0.0 Inspection and Measurement

- 3.1.0 Inspect for braze quality and leak test the assembly on the Veeco Leak Detector.
- 3.1.1 Check the assembly for shorts and bond resistance.
- 3.1.2 Back fill the assembly with argon at 1/3 atmosphere and pinch off tubulation close to the screw.

7. MODULE PANEL ASSEMBLY HTS-7Ala

1.0.0 Module Panel Assembly

1.1.0 Equipment

1.1.1 Fixture Assembly HTS-7

1.2.0 Procedure

- 1.2.1 Form the electrical connectors using jig assembly HST-7. Both n- and p-end connectors on TC #1 and #9 are to be left long.
- 1.2.2 Load thermocouples 1, 2 and 3 into holes 1, 3 and 5 of plate jig HST-7, respectively. Trim the ends of the connectors to the proper length, approximately 0.440 in. from the edge of the p-cold shoe ring.
- 1.2.3 Clamp the assemblies to the plate by securing the nut on the mount stud screw.
- 1.2.4 Clamp the electrical connectors between TC 1 and 2 and TC 2 and 3, using TP154G25C screw and nut.
- 1.2.5 Clamp a copper connector strap at right angles to the p-connector of TC #3.
- 1.2.6 Loosen the nuts from the mount stud screw and remove from plate.
- 1.2.7 Assemble the remaining two strips consisting of TC 4, 5 and 6 and TC 7, 8 and 9 in a similar manner. Care should be taken to maintain the proper orientation of the thermocouple holes in the heat receptor with respect to panel location.
- 1.2.8 Instrument all eight outer p-SiMo hot shoes with W-Re thermocouples. Instrument all cold shoes with Cr-Al thermocouples at the electrical connectors between couples.

2.0.0 Insulating Module Panel Assembly

- 2.1.0 Load module strip #1 into the outer insulation blocks (TP152B44A). Bring the cold shoe thermocouples outside the area of the mounting plate.
- 2.1.1 Load the mating segment of inner insulation blocks (TP152B44A) on the other side of module strip #1 and lock into outer insulation block.
- 2.1.2 Repeat this procedure for module strips 2 and 3. Use the other outside insulation blocks TP152B44A for securing module strip #3.
- 2.1.3 Bring the hot shoe instrumenting thermocouples along the outside of the insulating block in channels provided at this time.
- 2.1.4 Band the insulating material around the outside to secure the insulation in place.

Section VI

SUMMARY OF RESULTS

The results of the work performed under the Hybrid Thermocouple Development Program are as follows.

- 1. An analytical study was conducted to estimate Hybrid couple efficiency and define practical Hybrid couple geometries. The study indicated that the Hybrid couple efficiency would be 10 to 15 per cent better than that of all-SiGe (63 at.% Si alloy) couples.
- 2. A preliminary design of a planar generator using Hybrid thermocouples and a water heat pipe radiator was prepared. A specific power of 3.5 watts/kg (1.6 watts/lb) was estimated for a generator using current-design Hybrid couples operating at a hot shoe temperature of 941°C (1726°F) (hot junction temperature of 926°C (1700°F) and a cold junction temperature of 232°C (450°F)). A specific power of 5.3 watts/kg (2.4 watts/lb) is projected assuming the use of improved thermoelectric materials and operation at a hot junction temperature of 1000°C (1832°F).
- 3. A total of 64 Hybrid couples were built, using a number of different assembly techniques. Couples fabricated early in the program were, in general, characterized by high initial resistance (from 10 to 30 per cent higher than design), and, in some cases, bond separation after several thousand hours of testing. Couples made later in the program, using improved fabrication techniques, exhibited much lower initial resistance, closely approximating design values, and after limited testing, i.e., 700 hours, exhibit a resistance increase which is consistent with that expected as a result of phosphorus precipitation in the n-SiGe segment. Further testing and evaluation, presently being performed at NASA-LeRC, will be required to fully assess the performance of the Hybrid couple.

APPENDIX I

THERMOELECTRIC MATERIAL PROPERTIES

TABLE XXIV

PHYSICAL PROPERTIES - THERMOELECTRIC MATERIALS

(approximate values)

PbTe	19.5 to 20.5 x 10^{-9} °K		1000 psi	1000 psi	10,000 ps1	$2 \times 10^6 \text{ ps1}$	$5 \times 10^{-5} \text{ at } 932^{\circ} \text{F } (500^{\circ} \text{C})$	8.15 g/cm ³
Si-Alloy Hot Shoe	same as SiGe		l		1	1	ł	2.6 g/cm ³
S1Ge	$4.8 \text{ to } 5.0 \text{ x } 10^{-6} ^{\circ}\text{K}$		3900 psi	18d 0077	150,000 ps1	3600 psi	3 x 10 ⁻⁹ at 1472°F (800°C)	3 to 3.5 g/cm ³
Property	Thermal Expansion	Tensile Strength	p-type: 932° F (500° C)	$n-type: 932^{\circ}F (500^{\circ}C)$	Compressive Strength	Avg. Modulus of Rupture	Vapor Pressure	Density

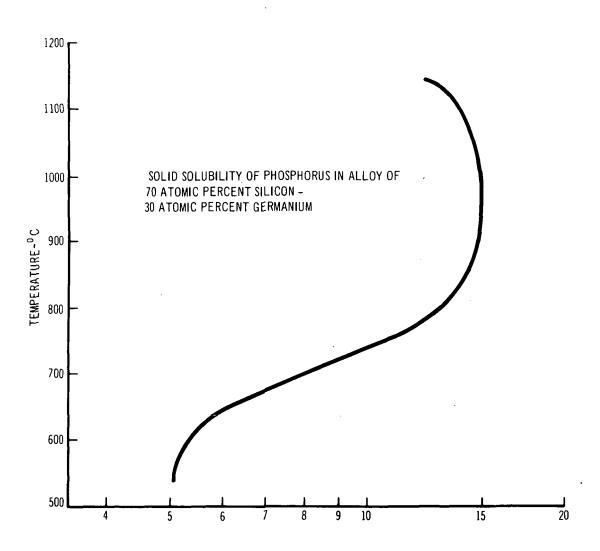
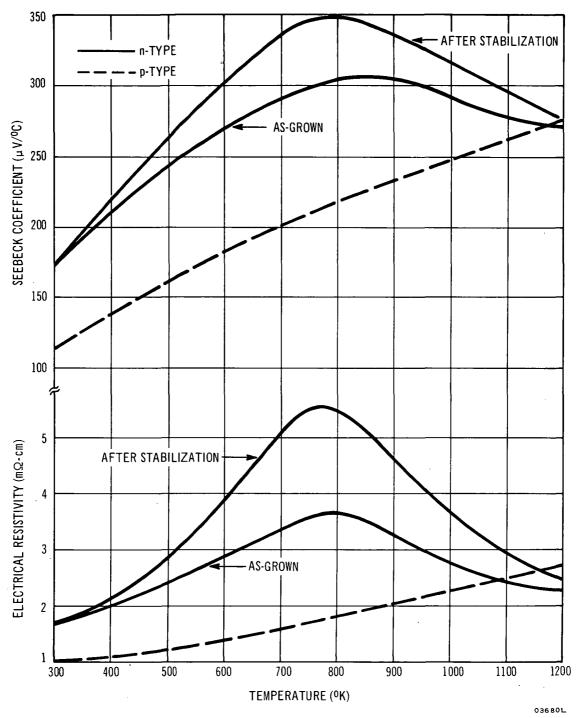


Figure 76. Carrier Concentrations (Atoms per Cubic Centimeter x 10^{19})



A. SEEBECK COEFFICIENT AND ELECTRICAL RESISTIVITY VS. TEMPERATURE

Figure 77. 63.5 At. % SiGe, Thermoelectric Properties (Sheet 1 of 2)

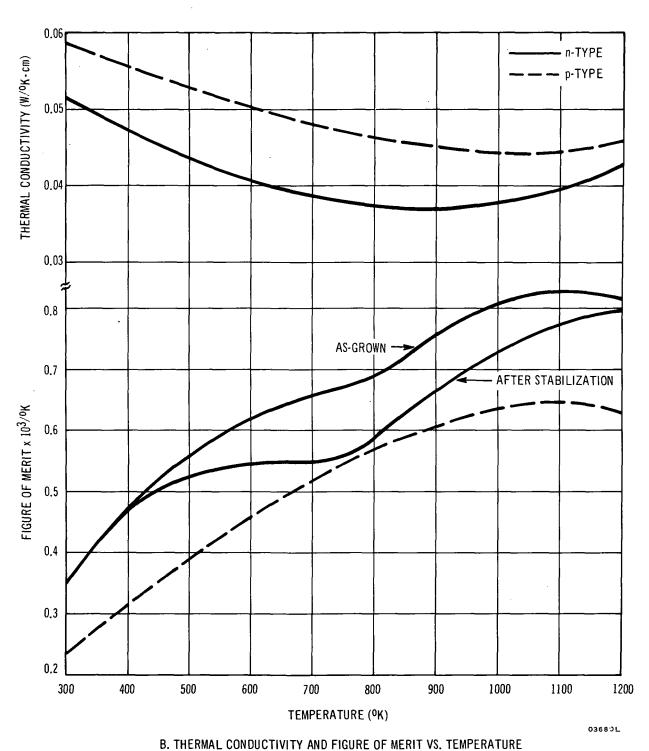


Figure 78. 63. 5 At. % SiGe, Thermoelectric Properties (Sheet 2 of 2)

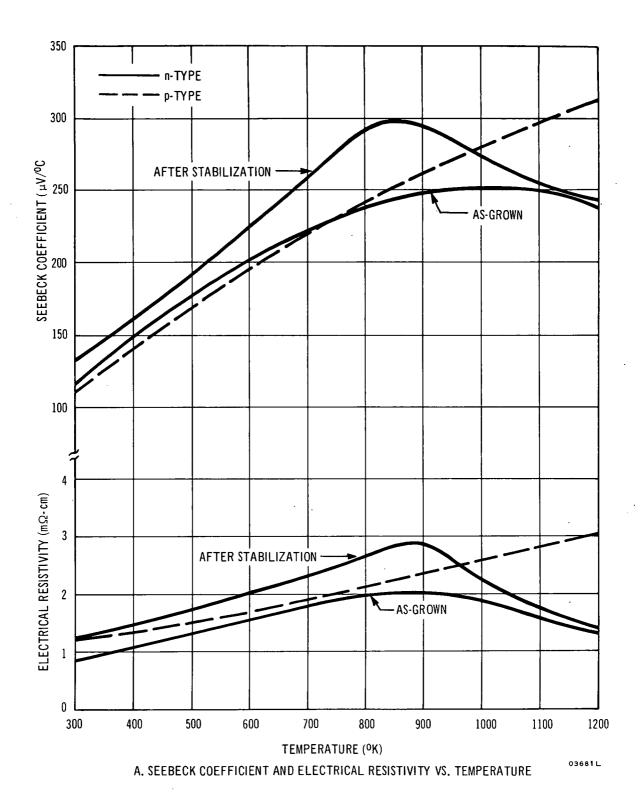


Figure 79. 80.0 At. % SiGe, Thermoelectric Properties (Sheet 1 of 2)

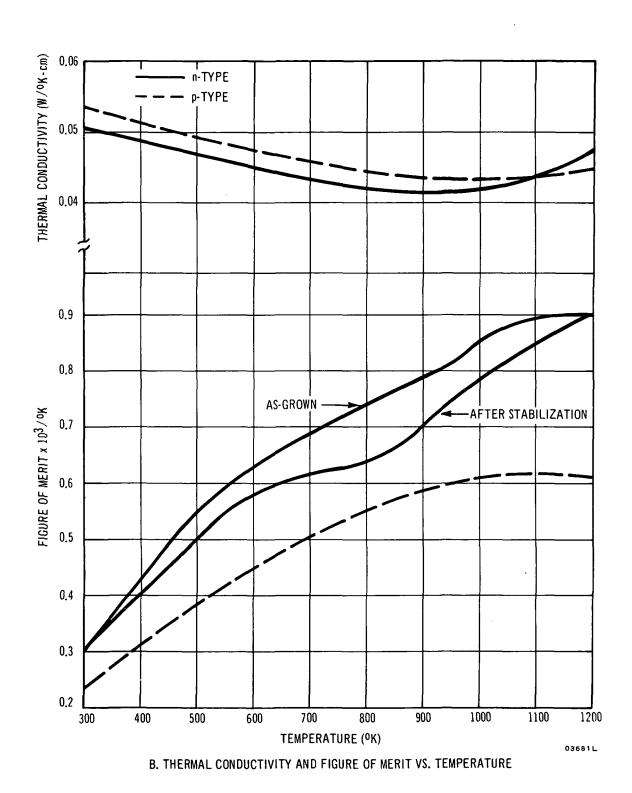
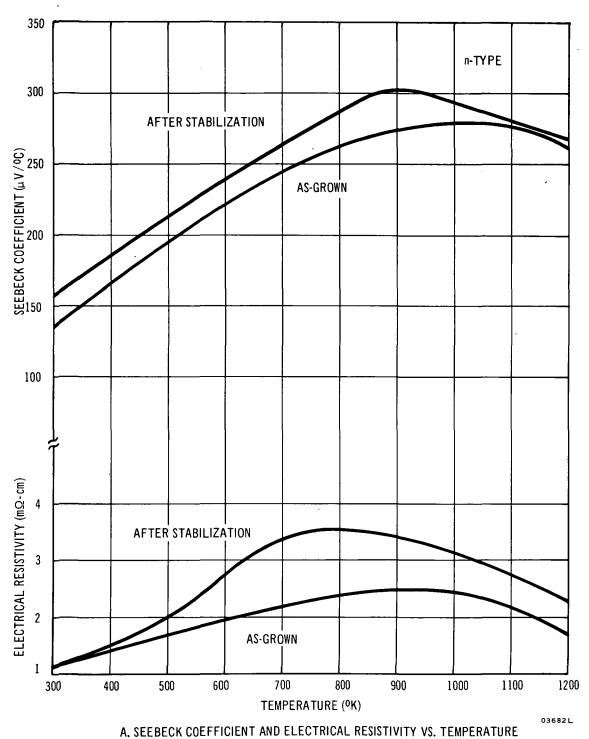


Figure 80. 80.0 At. % SiGe, Thermoelectric Properties (Sheet 2 of 2)



A. SELDEON OULT TOTALL AND ELECTRONE RESIDENTITY VS. TEMILENATURE

Figure 81. 70.0 At. % SiGe, Thermoelectric Properties (Sheet 1 of 2)

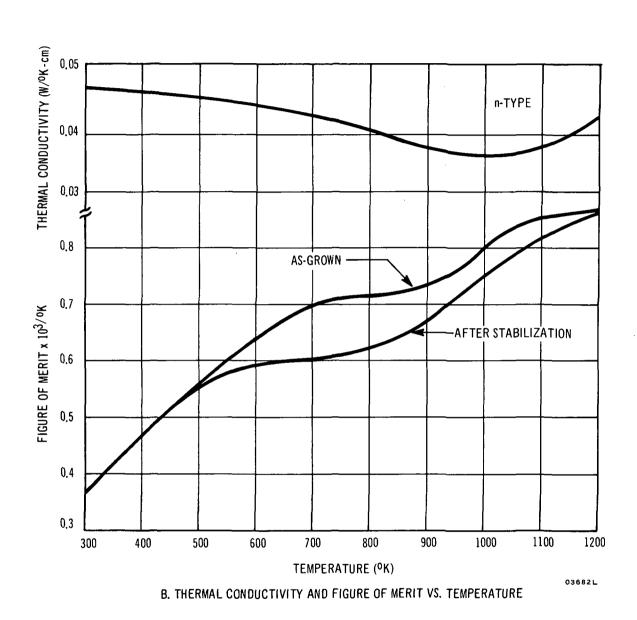


Figure 82. 70.0 AT. % SiGe, Thermoelectric Properties (Sheet 2 of 2)

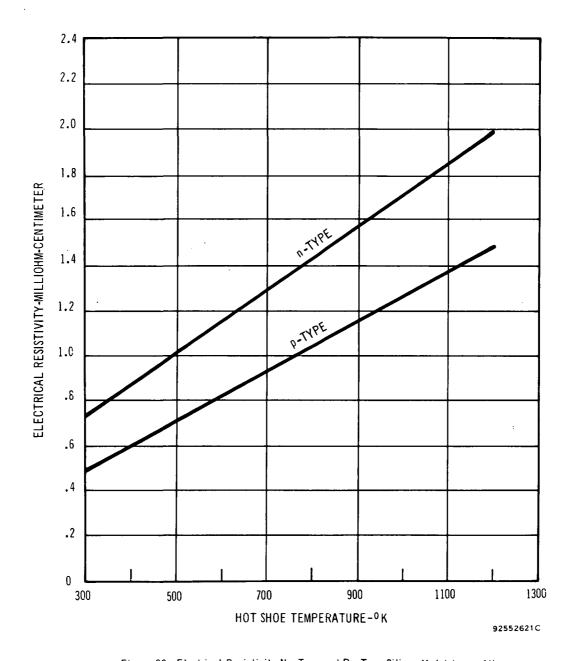


Figure 83. Electrical Resistivity N- Type and P- Type Silicon Molybdenum Alloy

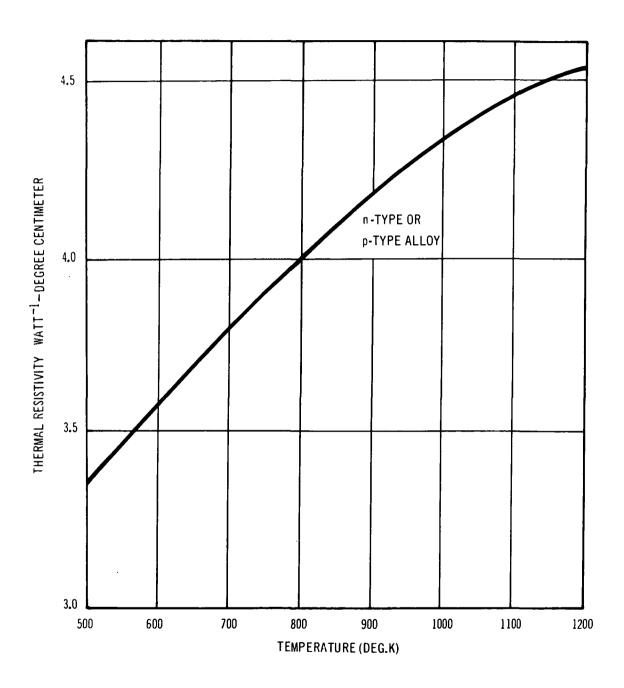


Figure 84. Thermal Resistivity of Silicon Molybdenum Alloy

APPENDIX II

COMPUTER DESIGN PROGRAMS

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APPENDIX II-A HYBRID COUPLE COMPUTER PROGRAM

```
/EDIT HYBRID
1-99999
          10 ALLOCATE 9,13,1,6,3
                       REAL TEMP(51), XSN(51), XPN(51), XRN(51), XSP(51), XPP(51), XRP(51)
                       REAL TEMP2(31), XS N2(31), XPN2(31), XR N2(31), XSP2(31), XPP2(31), XRP
                  2(31)
          40 REAL FXI(11)
          50
                  REAL FUNCTION MAXI(A,B)
          50
                    MA XI = A
          72
                     IF (B.ST.A) MAXI =B
                    RETURN
          83
          93
                    END
       123
                    INTEGER FUNCTION XMAXI(A,B)
       113
                   XMAX1=A
       127
                   IF(B.GT.4)XMAXI=B
       130
                    RETURN
       147
                    END
       153
                   INTEGER FUNCTION XMINI(A.B)
       1 53
                    XMINI =A
       179
                   IF(B.LT.A) MINI =B
       182
                    RETURN
       192
                   END
                    REAL FUNCTION MINI(A,B)
       200
       219
                    -MI N1 ±4
       220
                    IF(B.LT.A)MINI =B
       230
                   RETURN
       242
                    END
       250 C FORM ARRAY OF INPUT VARIABLES FOR SN AND SP
                       DO 192 J=1,51
       2.53
       277
                       READ 94X, (L) 9XX, (L) 1XX, (L
       283 9332 FOR MAI(F13.0,6F10.6)
                               TEMP(J) = TEMP(J) + 273.
       290 100
       320
                       DO 110 J=1.31
                      READ 9009, TEMP2(J), XSN2(J), XRN2(J), XPN2(J), XSP2(J), XRP2(J), XPP2
       310
                  (J)
       320 9009
                                FORMAT(F10.0,6F10.6)
      330 110
                            TEMP2(J) = TEMP2(J) + 273.
       340 DO 398 I=1,11
      350 398 READ 9236, FK1 (I)
       369 9235 FOR MAT(E12.5)
       379 GKHP = .22
       380 3K4 = .22
      390 GKB = . 36
       409 GKC = . 43
       410 GKD=.332
       429 SKCP=2.37
       430 TTA =2.54*.920
       440 TTB =2.54*.120
       45% TTC =2.54*.250
       46% TID=2.54*.190
       47% TTCP=2.54*.125
       430 TTHP = 2.54*.075
       49? RHOA = .0011
       590 RH03 = .2000216
       512 RHOC = .0000009
       520 RHOD=.3300057
       537 RON = . 3892
       548 RCN2 = .00005
       550 RCP = . 7071
       559 RSH = . 931
       573 J=13
       530 PI = 3.1415926
     590 SIGP=.557E-11
```

```
610 ETC = .3
      620 THP=273.+1010.
      630 TCP =273.+190.
      640 THN2 = 273.+550.
      650 EM=1.2
      669 GAP =2.54*.030
      679 TCN2=TCP
      680 THN=THP
      690 TCN=THN2
      700 QSO=0
      710 QSI =0
      720
          QTA =3.
      739 QT3 =10.
     740 QT=30.
      75@ FL2B = 0.6
      760 QH = 20.
      773 AHA = 45.
      730 K3=0
      790 PRINT 8000
     800 8002 FORMAT(//28X, 'HYBRID-1F')
     810 DO 1000 J4=14,14,2
     820 ANP = J4/10.
     830 PDUMP ANP
     840 DO 1000 J1=10,10
     850 ANR =J1/10.
     850 PDUMP ANR
     (870 DO 1000 J3=125,125,25
     880 FLP =2.54*J3/100.
     392 POUMP FLP
     900 DO 1000 J2=6,6
     910 DPHI=J2*1.
     920 PDUMP DPHI
     930 PHI =DPHI
     940 DO 1000 J5=20,20,4
     950 RRN=2.54*J5/100.
     960 PDUMP RRN
     970 K2 = 0
     980 AN=PI*RRN**2
     990 ANI =AN*ANR
    1000 DTRR = (RRN+GAP)*((1.+1./ANP*(RRN/(RRN+GAP))**2)**.5-1.)
    1010 AT=10.
    1920 RROP=RR N+GAP+DTRR
    1030 RRPO=RRN+GAP
    1040 ARROP = .5* (RROP+RRPO)
    1050 AP=2.*PI*ARROP*DTRR
    1950 AD=AP*2.
    1070 FII = 5.0* (THN - TCN) / (THN2 - TCN2)
    1939 FL2 = (FLP+ITD-TTA-ITB-TTC)/(FII+1)
    1298 FLI=FLP-FL2-TTA-TTB-TTC-TTD
    1100
           101 XN=(1273.0-THP)/20.0+1.0
    1113
            XM=(1273.0-TCP)/20.0+1.0
    1129
            XX = XM - XM + I
    1132
          YM = MAX1(0., XM-51.)
          XM = MINI(XM + 9., 51.)
    1140
    1150
           YN = XA X1 (9., (1.-XN))
    1150
          XN = MAXI(XN + \emptyset ... 1.)
    1170 M=XM
    1133 RM=XM-M
    1193 N=YN
    1200 RN=XN-1
    1210 SHP = XSP(N)
    1220 SCP =XSP (M)
    1230
           SSN=XSN(1)*YN+XSN(51)*YM+XSN(M)*RM-XSN(N)*RN
    1242
           SPN = XPN (1) * YN+ XPN (51) * YM+ XPN (M) * RM- XPN (N) * RN
P2 1250
           SRN=YRN(1)*YN+XRN(51)*YM+XRN(M)*RM-XRN(N)*RN
```

```
TYOU
           つうと ニュント イナンネイル ナスシア インエンネイ ロナスシア イロンネド ローグシト (か) キドロ
           SPP = XPP (1) * YN + XPP (51) * YM + XPP (M) * RM - XPP (N) * RN
    1279
           SRP =XRP(1)*YN+XRP(51)*YM+XRP(M)*RM-XRP(N)*RN
    1230
   1290
           DO 200 II = N.M
           SSN=SSN+XSN(II)
   1300
   1319
           SPN=SPN+XPN(II)
   1322
           SRN=SRN+XRN(II)
           SSP=SSP+XSP(II)
   1330
   1342
           SPP = SPP+XPP(II)
   1358
           200 SRP = SRP+XRP(II)
           SP =SSP/XX
   1360
   1370
           PP =SPP/XX
   1383
           RHOP =SRP/XX
   1350 UP = 3
   1400 DO 320 I=N.M-1
   141@ 32@ UP = UP + ((TEMP(I) + TEMP(I+1))/2) * (XSP(I) - XSP(I+1))/20.
   142@ UP = UP / (XX-1.)
   1430 XM=(1273.-TCB)/100.+1.
   1447 XN=(1273.-THP)/100.+1.
   1450 XX = XM - XN+1.
   1450 YN=MAX1(0..1-XN)
   1478 XN=MAXI (XN.1.)
   1480 YM=MAXI(0.,XM-II)
   1497 XM=MINI(XM,11.)
   1500 M=XM
   1510 N=XN
   1520 RM=XM-M
   1530 RN=XN-N
   1540 G=FKI(M)*RM-FKI(N)*RN+FKI(1)*YN+FKI(11)*YM
   1550 DO 311 I=N.M
   1563 311 G=G+FKI(I)
   1572 FKI =G/XX
            XN = (1273.0 - THN) / 20.0 + 1.0
   1539
   1593
           XM = (12.73.9 - TCN) / 20.0 + 1.0
   1507
           XX = XM - XN + I
         YM=MAX1(0..XM-51.)
   1 51 2
          XM = MINI(XM + \emptyset...51.)
   1623
   1630
           YN = MAX1(0..(1.-XN))
   1540
          XN = MAXI(XN + \emptyset ... I.)
   1650 M=XM
   1660 RM=YM-M
   1670 N=XN
   1637 RN=XN-N
   1590 SHN=XSN(N)
   1700 SCN=XSN(M)
           SSN =XSN(1)*YN+XSN(51)*YM+XSN(M)*RM-XSN(N)*RN
   1713
           SPN = YPN(1) * YN+ XPN(51) * YM+ XPN(M) * RM - XPN(N) * RN
   1722
   1739
           SRN=XRN(1)*YN+XRN(51)*YM+XRN(M)*RM-XRN(N)*RN
           SSP =XSP(1)*YN+XSP(51)*YM+XSP(M)*RM-XSP(N)*RN
   174%
   1752
           SPP =X PP (1)*YN+XPP (51)*YM+XPP (M)*RM-XPP (N)*RN
   1753
           SBP = XBP (1) * YN + XBP (51) * YM + XBP (M) * RM - XBP (N) * RN
           DO 275 II = V.M
   1772
           SST=SSN+YSN(II)
   1732
   1797
           SPM =SPM+YPM(II)
   1322
           SRV =SRN+XRN(II)
   1313
           SSP =SSP+XSP(II)
           SPP =SPP +XPP (II)
   1327
   1333
           2 75 SRP =SRP+XRP(II)
           SW=SSW/XX
   1343
   135%
           PM=824/XX
   1353
           XXVM SC=1, OH S
   1377 UN=3
   1337 DO 321 I=N.M-1
   1390 321 UM=UN+((TEMP(I)+IEMP(I+1))/2)*(XSN(I)-XSN(I+1))/20.
   1927 UN=UN/(XX-1.)
R3 1919
             Y (=(373.0-THN2)/20.0+1.0
```

```
大きこくち/3 TIUNZT/20.0TI.の
    1727
    1930
           XX = XM - XN + I
    1940 YM=MAX1 (0.,XM-31.)
    1950 XM=MINI(XM+Ø.,31.)
    1969
          YN = MAX1(Ø.,(1..-XN))
    1970 XN = MAXI(XN + \emptyset...1.)
    1980 M=XM
    1999 RM=XM-M
    2000 N=XN
    2010 RN=XN-N
    2020 SHN2 = XSN2 (N)
    2030 SCN2 = XSN2 (M)
           SSN2=XSN2(1)*YN+XSN2(31)*YM+X$N2(M)*RM-XSN2(N)*RN
    2747
    2959
           SPN2=XPN2(1)*YN+XPN2(31)*YM+XPN2(M)*RM-XPN2(N)*RN
    2050
           SR N2 = XR N2 (1) * YN + XR N2 (31) * YM + XR N2 (M) * RM - XR N2 (N) * RN
    2270
           SSP2=XSP2(1)*YN+XSP2(31)*YM+XSP2(M)*RM-XSP2(N)*RN
           SPP2=XPP2(1)*YN+XPP2(31)*YM+XPP2(M)*RM-XPP2(N)*RN
    2930
    2290
           SRP2=XRP2(1)*YN+XRP2(31)*YM+XRP2(M)*RM-XRP2(N)*RN
    2133
           DO 210 II = N, M
    2119
           SSN2=SSN2+XSN2(II)
    2123
           SPN2=SPN2+XPN2(II)
    2139
           SRN2=SRN2+XRN2(II)
    2143
           SSP2 =SSP2+XSP2(II)
    21.50
           SPP2=SPP2+XPP2(II)
           210 SRP2 = SRP2 + XRP2(II)
    2150
    2170
           SW2 =SSW2/XX
           PN2 = SPN2/XX
    2133
    2190
           RHON2 = SRN2/XX
    22 30 UN2 =3
    2213 DD 322 I =N,M-1
    2220 322 UN2=UN2+((TEMP2(I)+TEMP2(I+1))/2)*(XSN2(I)-XSN2(I+1))/20.
    2239 UN2 = UN2/(XX-1.)
   2240 HKA = GKA*ANI/TTA
   2257 HKB = GKB*AN/TTB
    22 SØ HKC =GKC*AN/TIC
   2270 HKD=GKD*AD/TTD
   2280 RA=TTA/ANI*RHOA
   2299 RB =TTB /A N*RHOB
   2300 RC = TTC /A N*RHOC
   2310 RD=TTD/AD*RHOD
    2329 EOC = SP*(THP-TCP)+SN*(THN-TCN)+SN2*(THN2-TCN2)
    2330 RN=FL1/AN1*RHON+2.*RCN/AN1+FL2/AN*RHON2+2.*RCN2/AN+RA+RB+RC
    23 40 RP =FLP/AP*RHOP+2.*RCP/AP+RD
   23 50 R = R N+RP+RSH
    23 SØ CUR =EOC /((1+EM)*R)
   2370 RL=EM*R
   2389 PO=CUR**2*RL
    2393 DTP =THP -TCP
    2400 DIN = THN - TCN
    2417 DIN2=THN2-TCN2
   2 42 0 QOC =PP*AP/FLP*DTP+CUR*SCP*TCP+ .5*CUR**2*RP+ .5*CUR*UP*DTP
   2430 DI=DTP+QOC/HKD
   244@ QI2C =PN2*AN/FL2*DTN2+CUR*SCN2*TCN2+.5*CUR**2*RN+.5*CUR*(UN*DIN+UN
         2*DIN2)
   2457 ICN2 =TCP -QOC/HKD+QI2C/HKC
   2460 TC3=ICP-QOC/HKD
   2470 QI 1H=PN*AN1/FL1* DIN+CUR*SHN* THN -.5*CUR**2*RN -.5*CUR*(UN* DIN+UN2*D
         TN2)
   2430 THN=THP-QIIH/HKA
   2 49 @ QIIC =P B * 4 BI/FLI* DIN+C UR*SC N* TC N+ .5*C UR**2*R N+ .5*C UR*(UN* DIN+ UN2* D
         TN2)
   25@@ QI2H=PN2*AN/FL2*DTN2+CUR*SHN2*THN2-.5*CUR**2*RN-.5*CUR*(UN*DTN+UN
         2*DTN2)
   2513 TCN=THN2+QI IC/HKB
   .2529 FL24=FL2
79 2539 FL2=(P V2*A N*DIN2)/(PN*A N1/FL1*DIN+CUR*(SC N*TC N-SHN2*THN2)+1.0*CUR
```

```
<u> 本本乙本RガキしつR本( リガギ リエNギ リカ2米 ロエNと)丿</u>
   2540 FL23 =FL2
   2550 FLI =FLP+TTD-FL2 -TTA-TTB-TTC
   2560 HKP=PP*AP/FLP
   2570 HKN=PN*ANI/FL1
   2580 HKN2 =PN2*4N/FL2
   2599 HKT=(HKP*HKD)/(HKP+HKD)+((HKN*HKN2*HKA*HKB*HKC)/((HKN*HKN2*HKA*HK
         3)+(HK N*HK N2*HK 4*HKC)+(HK N*HK N2*HKB*HKC)+(HK N*HK A*HKB*HKC)+(HK N2*
         HKA*HKB*HKC)))
   2600 QHA =QH
   261 @ QH =HK T* DT+C UR* (SHP* THP+SHN* THN) -.5*C UR**2*R -.5*C UR* (UP* DTP+ UN* DT N
         (SNIG*SNU+
   2620 QHB = QH
         715 IF (ABS(QHA-QH)/QHA .LE. (.001)) IF (ABS(FL2A-FL2B)/FL2A.LE.(.
   2 53 7
         @@1))GO TO 7@3
   2543 K2=K2+1
   2550 IF(K2.3T.90) PDUMP K2.FL2.QS.QH.QI2C,QI1H,QI1C,QI2H,QS0,QT,QSI
   2660 IF (K2.GE.150) GO TO 9000
   2578 GO TO 181
   2580 AT=((SQRT(AH))+2.54*.020)**2.
   2590 703AH=QH/PHI
   2799 739 CONTINUE
   271% AHH = ((SQRT(AT))-2.54*.020)**2.
   2720 RH=((QT*(1-(PI*RROP**2)/AH)-QSO)/(AH-PI*RROP**2))/ITHP
   2730 IF(AH.LE.(.1))AH=PI*RROP**2+.2
   2740 RRTO = SQRI(AH/PI)
   2750 QX =0
   2760 DO 302 NNN=1.10
   2770 RR = RROP+((NNN - . 5) * (RRTO - RROP))/10.
   2730 THR =RH / (2 .* GKHP) * (RRTO**2* LOG (RR /RROP) - .5* (RR**2 -RROP**2)) + THP
   2790 RIC =-((QT-PO)*(1.-(PI*RROP**2)/AH)-QSO/(AH-PI*RROP**2))/TICP
   2333 TCR =RIC/(2.*GKCP)*(RRTO**2*LOG(RR/RROP)-.5*(RR**2-RROP**2))+TCB
   2810 IF (TCR.LE.50.) TCR =50.
   2320 DR = (RR TO - RR OP) / 10.
   2330 802 QK = QK + (THR - TCR) * RR * DR
   2840 QSO = (2.*PI* FKI) / (FLP+ ITD) *QK
   2850 EI = 1./(1./EHP+1./ETC-1.)
   2867 OSI =PI*(RRPO**2.-RRN**2)*EI*SIGP*(THP**4.-TCB**4.)
   2870 QS = QS O+QS I
   2880 QT=QH+QS
   2890 PHIA=PHI
   2900 PHI =QT/AH
   2910 AHB = AHA
   2920 AHA = AH
   2930 AH = AHA+ (DPHI - PHI)* (AHA - AHB) / (PHI - PHIA)
         IF(ABS(AHA-AH)/AHA .LE. (.001))IF(ABS(DPHI-PHI)/DPHI .LE.(.001))
         GO TO 795
   2950 K3=K3+1
   2969 IF(X3.EQ.99) IF(AH.LT.(PI*RROP**2)) GO TO 9112
   2970 IF(K3.GE.100)GO TO 9112
   298% GO TO 73%
   2990 735EL =PO/CUR
   3999 PDUMP AH
   3010 ETA =PO/QT
   3929 THPC =THP -273.0
   3030 TCPC =TCP -273.0
   3342 TC3C =TCB -273.0
   3952 THNZC =THN2 -273.0
   3767 TCN2C =TCN2 -273.0
   3970 ICNC=ICN-273.0
   3030 THNC = THN - 273.0
   3798 PRINT 9172, THPC, AN, FLI, ETA
                            THPC = 'F7.1,7X, 'AN= 'F7.4,6X, 'FL1 = 'F7.4,6X, 'ETA = 'F
   3130 9130 FORMAT('
         7.5)
   3110 PRINT 9101, TCPC, AP, FL2, PO
P5 3128 9181 FORMAT("
                           TCPC = 'F7.1.7X, 'AP = 'F7.4, 6X, 'FL2 = 'F7.4, 7X, 'P0 = 'F7
```

```
3130 PRINT 9102, TOBC, DTRR, QT, EL
3140 9102 FORMAT('
                       TCBC = ' F7.1,5X, 'DTRR = 'F7.4,7X, 'QT = 'F7.3,5X, 'VOL T =
     'F7.4)
3150 PRINT 9103, THNC, THN2C, QS, R
3169 9193 FORMAT(*
                       THNC = 'F7.1,4X, 'THN2C = 'F7.2,7X, 'QS = 'F7.3,8X, 'R = 'F
3170 PRINT 9104, TCNC, TCN2C, QSO, PHI
3139 9194 FORMATC"
                      TCNC='F7.1,4X,'TCN2C='F7.2,6X,'QSO='F7.3,6X,'PHI
     ='F7.4//)
3190 1000 CONTINUE
3230 STOP
3219 9000 PRINT 9001
3222 9001 FORMAT(' TOO MANY SEARCHES FOR QH')
3230 GO TO 1000
3240 9110 PRINT 9111
325% 9111 FORMAT(' TOO MANY SEARCHES FOR AH')
3259 30 TO 1999
3270 9112 PRINT 9113
3280 9113 FORMAT (' AH.LT.COUPLE AREA')
3290 30 TO 1000
```

APPENDIX II-B HYGEN CONVERTER COMPUTER PROGRAM

```
/RESEQ HYGENI
REA DY
/EDIT
1-500
    10 C T 01/08/70
         ALLOCATE 7.10.1.11.3
           REAL TEMP(51), XSN(51), XPN(51), XRN(51), XSP(51), XPP(51), XRP(51)
    39
            REAL TEMP2(31), XS N2(31), XPN2(31), XR N2(31), XSP2(31), XPP2(31), XR
      40
         P2(31)
    50
          REAL FUNCTION MAXI (A,B)
    60
          MA XI =A
    77
          IF(B.GT.A)MAXI=B
          RETURN
    80
    90
          END
   100
          INTEGER FUNCTION XMAXI(A.B)
   112
          XMA XI =A
          IF(B.GT.A) XMAX1 =B
   120
   130
         RETURN
   140
          END
   150
          INTEGER FUNCTION XMINI(A.B)
   160
          XMINI =A
   170
          IF (B.LT.A) MINI =B
   189
         RETURN
   190
          END
   200
         REAL FUNCTION MINI(A.B)
   210
          MI NI =A
   220
         IF(B.LT.A)MINI=B
   230
         RETURN
   2 40
          END
           FORM ARRAY OF INPUT VARIABLES FOR SN AND SP
   250 C
   260
           DO 100 J=1,51
   270
           READ 9002, TEMP (J), XSN(J), XRN(J), XPN(J), XSP (J), XRP(J), XPP (J)
   280
         9002
               FOR MA T ( F1 0.0.6F1 0.6)
   290
              TEMP (J) = TEMP (J) +273.
         100
   300
           DO 110 J=1.31
   310
           READ 9009.TEMP2(J).XSN2(J).XPN2(J).XPN2(J).XPP2(J).XRP2(J).XPP
         2(J)
   320
         9009
               FORMAT(F10.0.6F10.6)
   330
         110
              TEMP2(J) = TEMP2(J) + 273.
   3 40
         GKHP = .22
   350
        SKA = .22
   360
         GKB = .86
   370
         GKC = . 43
   380
         GK D=.332
   390
         GK E = . 455
   400
        GKCPBE=1.3
   410
         TTA =2.54*.020
   420
         TTB =2.54*.120
   430
         TTC =2.54*.250
   448
         TTD=2.54*.190
         TTCP =2 .54* .250
   450
   453
         TIHP =2 .54* .075
   470
        TTCASE =2 .54* .200
   487
        RHOA = . 0011
   490
        RHOB = 0000216
```

500

RHOC = .0000009

```
510-1020
         RHOD = . 00000057
   510
   520
         RCN=.0002
   530
         RC N2 = .00005
   540
         RCP = . 0001
   550
         RSH = . 991
         J=10
   560
   578
         PI =3 .1 41 5926
         SIGP = . 567E-11
   580
   590 ER = .85
         EHP=.6
   600
   610
         ETC = .3
         EFC = .85
   620
         THP =273.+926.
   630
   640
         DO 1000 I1=500.538.38
   650 THN2 = I 1+273.
   660 THN2C =THN2 -273.
   670 PDUMP THN2C
         DO 1000 I2=122,232,55
   680
   690
         TCP =12+273
         TCPC =TCP -2 73.
   700
   710
         PDUMP TCPC
   720
         TC V = 273.+25.
   730
         TCA = 73.
   7 40
         ANR = . 7
   750
         EM =1 .2
   769
         DTRR =2.54*.055
         FAS =2.54*.040
   770
   780
         GAP =2.54*.030
   790
         FLP = 2.54 * 1.250
         RHCU=.65E-8* TCP-.35E-6
   800
         RR N=2.54*.1875
   810
         AN =PI*RR N**2.
   82Ø
   830
         ANI =ANR *AN
   843
         RROP = RRN+GAP+DIRR
   850
         RRPO = RRN+GAP
         ARROP = .5*(RROP + RRPO)
   860
         AP =2 .*PI *ARROP*DTRR
   870
   875 ANP =AN/AP
   880
         AD=2.*AP
         FKIE=.00016
   890
         FK IC =. 001
   900
         CAPFL =2 .54* 6.75
   910
         CAPW = 2.54*3.50/.866
   920
   930
         CAPH =2.54*3.50
         ACAP = CAPFL * CAPW
   940
         CWFUEL =2.5
   9 50
   9 50
         PZ =635.
         D=(GKHP*TTHP)/(EHP*SIGP*THP**3.*PI*RROP**2.)
   979
   980
         FLG =FLP+TTD+TTHP+2.54*Ø.5+CAPH
         FLINS =FLP
   990
  1000
         FH =2.54*1.01
  1010
         AH =FH**2.
         AT=(FH+FAS)**2.
  1020
```

```
1030-1250
```

- 1930 ASTUD=2.*PI*(.550**2-.200**2)*6.45
- 1040 QH=5.0
- 1050 C INITIALIZED CONDITIONS
- 1060 (THR =THP+30; THN=TCP; THF=THR+50.; TCN=THN2; TCN2=TCP; FI=5.*(THN-TCN)
 /(THN2-TCN2); FL2=(FLP+TTD-TTA-TTB-TTC)/(FI+1.0); FL1=FLP-FL2-TTA-T
 TB-TTC+TTD: ETAN=0.5: ETAF=.5: GQT=4000.; GPO=250.; GEL=28.;)
- 1070 K1=0
- 1080 K3=0
- 1090 K5=0
- 1100 101 XN=(1273.0-THP)/20.+1.0
- 1110 (XM=(1273.0-TCP)/20.0+1.0;XX=XM-XN+1;YM=MAX1(0.,XM-51.);XM=MIN1(X M+0.,51.);YN=MAX1(0.,(1.-XN));XN=MAX1(XN+0.,1.);M=XM;RM=XM-M;N=XN ;RN=XN-N;SHP=XSP(N);SCP=XSP(M);SSN=XSN(1)*YN+XSN(51)*YM+XSN(M)*RM -XSN(N)*RN;SPN=XPN(1)*YN+XPN(51)*YM+XPN(M)*RM-XPN(N)*RN;)
- 1120 (SR N=XR N(1)*YN+XR N(51)*YM+XRN(M)*RM-XRN(N)*RN;SSP=XSP(1)*YN+XSP(5 1)*YM+XSP(M)*RM-XSP(N)*RN;SPP=XPP(1)*YN+XPP(51)*YM+XPP(M)*RM-XPP(N)*RM-XPP(N)*RN;SRP=XRP(1)*YN+XRP(51)*YM+XRP(M)*RM-XRP(N)*RN;DO20011=N,M;SSN=SSN+XSN(II):SPN=SPN+XPN(II):SRN=SRN+XRN(II):)
- 1130 (SSP=SSP+XSP(II);SPP=SPP+XPP(II);200SRP=SRP+XRP(II);SP=SSP/XX;PP= SPP/XX;RHOP=SRP/XX;UP=0;D0320I=N,M-1;320UP=UP+((TEMP(I)+TEMP(I+1))/2)*(XSP(I)-XSP(I+1))/20.;UP=UP/(XX-1.);XN=(1273.0-THN)/20.0+1.0 ;XM=(1273.0-TCN)/20.0+1.0;XX=XM-XN+1;YM=MAX1(0.,XM-51.);)
- 11 40 (XM=MIN1(XM+0.,51.);YN=MAX1(0.,(1.-XN));XN=MAX1(XN+0.,1.);M=XM;RM =XM-M;N=XN;RN=XN-N;SHN=XSN(N);SCN=XSN(M);SSN=XSN(1)*YN+XSN(51)*YM +XSN(M)*RM-XSN(N)*RN;SPN=XPN(1)*YN+XPN(51)*YM+XPN(M)*RM-XPN(N)*RN ;SRN=XRN(1)*YN+XRN(51)*YM+XRN(M)*RM-XRN(N)*RN;)
- 1150 (SSP=XSP(1)*YN+XSP(51)*YM+XSP(M)*RM-XSP(N)*RN;SPP=XPP(1)*YN+XPP(5 1)*YM+XPP(M)*RM-XPP(N)*RN;SRP=XRP(1)*YN+XRP(51)*YM+XRP(M)*RM-XRP(N)*RN;DO205II=N,M;SSN=SSN+XSN(II);SPN=SPN+XPN(II);SRN=SRN+XRN(II);SSP=SSP+XSP(II);SPP=SPP+XPP(II);205SRP=SRP+XRP(II);)
- 1160 (SN=SSN/XX;PN=SPN/XX;RHON=SRN/XX;UN=0;D0321I=N,M-1;321UN=UN+((TEMP(I)+TEMP(I+1))/2)*(XSN(I)-XSN(I+1))/20.;UN=UN/(XX-1.);XN=(873.0-THN2)/20.0+1.0;XM=(873-TCN2)/20.0+1.0;XX=XM-XN+1;YM=MAX1(0.,XM-31.);XM=MIN1(XM+0.,31.);IF(XM.LE.1.0)XM=1.0;)
- 1170 (YN = MAX1(0.,(1.-XN)); XN = MAX1(XN+0.,1.); M=XM; RM=XM-M; N=XN; RN=XN-N; SHN2=XSN2(N); SCN2=XSN2(M); SSN2=XSN2(1)*YN+XSN2(31)*YM+XSN2(M)*RM-XSN2(N)*RN; SPN2=XPN2(1)*YN+XPN2(31)*YM+XPN2(M)*RM-XPN2(N)*RN; SRN2=XRN2(1)*YN+XRN2(31)*YM+XRN2(M)*RM-XRN2(N)*RN;)
- 1180 (SSP2=XSP2(1)*YN+XSP2(31)*YM+XSP2(M)*RM-XSP2(N)*RN;SPP2=XPP2(1)*YN+XPP2(31)*YM+XPP2(M)*RM-XPP2(N)*RN;SRP2=XRP2(1)*YN+XRP2(31)*YM+XRP2(M)*RM-XRP2(N)*RN;D021011=N,M;SSN2=SSN2+XSN2(II);SPN2=SPN2+XPN2(II);SRN2=SRN2+XRN2(II);SSP2=SSP2+XSP2(II);)
- 1190 (SPP2 = SPP2+XPP2(II);210SRP2 = SRP2+XRP2(II); SN2 = SSN2/XX; PN2 = SPN2/XX; RHON2 = SR N2/XX; UN2 = 0; DO322I = N, M-1;322 UN2 = UN2+((TEMP2(I)+TEMP2(I+1))/2)*(XS N2(I)-XS N2(I+1))/20.; UN2 = UN2/(XX-1.);)
- 1200 (HKA=GKA*ANI/TTA; HKB=GKB*AN/TTB; HKC=GKC*AN/TTC; HKD=GKD*AD/TTD; RA= TTA/ANI*RHOA; RB=TTB/AN*RHOB; RC=TTC/AN*RHOC; RD=TTD/AD*RHOD;)
- 1210 EOC = SP*(THP-TCP)+SN*(THN-TCN)+SN2*(THN2-TCN2)
- 1220 RN=FL1/AN1*RHON+2.*RCN/AN1+FL2/AN*RHON2+2.*RCN2/AN+RA+RB+RC
- 1230 RP =FLP/AP*RHOP+2 .*RCP/AP+RD
- 1240 RCONN=RHCU*SQRT(AT)/(RROP*2.54*.020)
- 1250 RSH = 2*PI*RHOA* TTHP/ALOG(RROP/(RRN*ANR))

```
1260-1680
  1260
        R = RN+RP+RSH+RCONN
  1270
         CUR = EOC/((1+EM)*R)
  1280
         RL =EM*R
  1290 EL = CUR * RL
  1300
         PO = C UR * * 2 * R L
  1310
         DTP =THP -TCP
  1320
         DTN=THN-TCN
  1330
         DTN2 = THN2 -TCN2
         QOC =PP*AP/FLP*DTP+CUR*SCP*TCP+.5*CUR**2*RP+.5*CUR*UP*DTP
  1349
  1350
         DT = DTP+QQC/HKD
  1369
         Q12C=PN2*AN/FL2*DIN2+CUR*SCN2*TCN2+.5*CUR**2*RN+.5*CUR*(UN*DIN+U
        N2*DIN2)
  1377
         TC N2 =TCP -QOC /HKD+QI2C /HKC
  1389 TCB = TCP - QOC / HKD
         QI 1H=PN*AN1 /FL1*DIN+CUR*SHN* THN-.5*CUR**2*RN-.5*CUR*(UN*DIN+UN2*
  1390
       DIN2)
  1400
         THN=THP-QIIH/HKA
         QIIC=PN*ANI/FLI*DIN+CUR*SCN*TCN+.5*CUR**2*RN+.5*CUR*(UN*DIN+UN2*
  1419
        DTN2)
  1 42 3
         QI2H =PN2*AN/FL2*DIN2+CUR*SHN2*THN2-.5*CUR**2*RN-.5*CUR*(UN*DIN+U
        N2*DTN2)
  1430
         TC N = TH N2 +Q I 1 C / HKB
  1442
         FL2 = (PN2*AN*DIN2)/(PN*AN1/FL1*DIN+CUR*(SCN*TCN-SHN2*THN2)+1.0*CU
       R**2*R N+C UR* ( UN* DT N+ UN2* DT N2 ) )
  1450
         FI.1A = F1.1
         FLI =FLP+TTD-FL2-TTA-TTB-TTC
  1460
  1470
          FLIB =FLI
  1480
          FIL = (FLIA+FLIB) /2.
  1490
         HKP =PP*AP/FLP
  1500
         HKN=PN*ANI/FLI
  1518
         HK N2 =P N2 *A N/FL2
         HK T = ( HKP*HKD) / ( HKP+HKD) + ( ( HK N* HK N2*HKA* HKB* HKC) / ( ( HK N* HK N2*HKA*H
  1 52 8
        KB)+(HKN*HKN2*HKA*HKC)+(HKN*HKN2*HKB*HKC)+(HKN*HKA*HKB*HKC)+(HKN2
       *HKA*HKB*HKC)))
  1 53 3
        HP= AHD
         QH =HK T* DT+C UR* (SHP* THP+SHN* THN) -.5*C UR**2*R-.5*C UR* ( UP* DTP+ UN* DT
  1548
        N+UN2*DTN2)
  1559
         QHB =QH
  1560
           IF (ABS(QHA-QH)/QHA .LE. (.001))IF(ABS(FLIA-FLIB)/FLIA.LE.(.00
        1))GO TO 703
  1572
        K1 = K1+1
  1580
          IF(KI.GT.180) PDUMP KI.QH.QHA.QHB.FLI.FL2.FLIA.FLIB
  1593
         IF(K1.GE.200) GO TO 9000
  1600
         30
            TO 101
  1613
             QSI =PI*(RRPO**2.-RRN**2.)*(THP-TCB)*FKIC/(FLP+TTD-TTA)
        793
  1620
         IGCPL=GEL/EL
  1630
         GPO1 =2 .* IGCPL* PO
  1640
         SCPL=2.*IGCPL
  1650 505 IGCPL=SQRT(GCPL+1)+K3
  1660 GCPL=IGCPL**2.
  1670
         GPOI =PO* GCPL
  1680
         GELI =EL* GCPL/2.
```

```
1699-2179
       IF(GELL.GE.GEL) IF(GPOL.GE.GPO) GO TO 500
  1690
  1799
        K3 = K3 + 1
        IF(K3.GE.100) GO TO 9005
  1717
  1722
        GO TO 505
  1730
        500 GPO = GPO1
  1742
        GEL =GELI
  1750
        GAT=GCPL*AT
  1760 GQSO=(GAT-GCPL*AH)*FKIE*(THF-TCP)/FLP+(GCPL*AH-GCPL*PI*RROP**2.)*
       FKIE*(IHR -TCP)/FLP
  1779
        GQH =GCPL*QH
  1782
        GAC =GAT
 1793 ACASE=(SQRT(GAC)+2.*FLP)**2.+4.*FLG*(SQRT(GAC)+FLP)
        GQCASE = ACASE* FKIE*1.25* (THF-TCV) /FLINS
  1897
 1810
        GQSI =GCPL*QSI
        GQ T = (GQH + GQSO + GQSI + GQCASE) *1.05
 1829
 1837
        B=RROP**2.*PI/AH
 1840
        Z =D** .1 *B** .125
 1850
        IF(Z.LE.(0.9)) GO TO 525
 1857
        ETAN=-14.971+42.821*7-43.221*7**2.+19.457*7**3.-3.295*7**4.
        GO TO 53 Ø
 1870
 1889
        525 ETAN=3.1126*Z+.6579*Z**2.-1.365**Z**3.+1.4342*Z**4.
 1890 530 IF(ETAN.GE.1.0) ETAN=1.0
        EN =1 ./(1 ./EFC+1 ./(EHP*ETAN) -1 .)
 1900
 1913
        TFC = ((GQH+GQSO+GQSI)/(SIGP*EN*GAC)+THP**4.)**.25
        THR = THP + 420 .* (1 . - ETAN) /1 .8
 1920
 1930 535 TCRR =TCB -QH/(GKE*ASTUD) - (GQH+GQSO+GQSI -GPO)* TTCP/(GKCPBE* GAC)
       AFIN=(GQH-GPO+GQSO+GQSI)/(SIGP*ETAF*ER*(TCRR**4.-TCA**4.))
 1940
 1950 IF(ETAF.GE.1.0) ETAF=1.0
 1960 IF(ETAF.LE.(.01)) ETAF=.01
 1970
        540 IF(ETAF.LT.(0.58))GO TO 545
 1980
        ZF = 360.516-1846.222* ETAF+3502.456* ETAF**2.-2916.253* ETAF**3.+901
       .906* ETAF** 4.
 1999 GO TO 559
 2000 545 ZF=0.002+5.504* ETAF-17.629* ETAF**2.+29.950* ETAF**3.-19.505* ET
       AF**4.
 2010 550 BF=GAC/AFIN
 2020 DF = (7F/BF**.125)**10.
 2030 TTFI N=ER*SIGP*TCRR**3.*GAC*DF/GKCPBE
 2040 IF(ABS(TTCP-TTFIN).LE.(.01*TTCP))GO TO 555
 2050 ETAFI = ETAF
 2050 ETAF=ETAF1-ETAF1*(TTFIN-TTCP)/TTFIN
 2070 ETAF=(7*ETAF1+ETAF)/8.
 2080 IF(ETAF.GE.1.0)ETAF=1.0
 2090 IF(ETAF.LE.(.01)) ETAF=.01
 2130
       IF(K5.GE.98)GO TO 9114
 2110 K5=K5+1
 2123 GO TO 535
 2130 C WEIGHT CALCULATIONS FOLLOW: W PREFIX=WEIGHT, D PREFIX=DENSITY
 2140 555 DSIMO = 2.80
 2159 DSIGE=3.535
 21 60 DP3 TE = 3.15
 2173 DNI =8.20
```

```
21 80 -2 61 9
  2180 DAL203=3.85
  2190 DALMN=2.82
  2200 DSS =8.02
  2210 DBRYL=1.85
  2223 DCU=8.95
  2230 DMINK = .32
  2240 DASTRO=.838
  2250 DMOLY=10.2
  22 50 DPLAT=21.5
  2270 DZR02=.485
  2230 DSI 92 = .056
  2290 DQ UR T7 = . 16
  2300 DZCAR =.225
  2310 DAU=19.6
  2320 DW=19.3
  2330 C ASSUME FOR TH = 1000 DEGREES C,30% MOLY,60% NI,AND 10% AL.
       TH = 1100 DEGREES C USE 30% MOLY-, ZRO2, 15% MOLY-ASTRO AND 10%
       AL-S192
  2340 C FOR TH=1100 DEGREES C,352 PLATINUM WITH DZRO2,452 PLATINUM WITH
        DASTRO, AND 20% DNI WITH DASTRO.
  2350 C DF0IL=.35*(.5*DZRO2*9.75+DPLAT*.25)/10.+.45*(.5*DASTRO*9.75+DPL
       AT* .25)/10.+.20*(.5* DASTRO*9.75+ DNI*.25)10.
 2350 DF0IL=.30*(.7*DZR02*8.33+DM0LY*.25)/8.58+.15*(.5*DASTR0*10.+DM0LY
       *.25)/10.25+.48*(.5*DASTRO*10.+DNI*.25)/10.25+.07*(DSI92*8.33+DAL
       MN*.25)/8.58
 2370 WHP =DSI MO* TTHP*AH
 2380 WSIGE=DPBTE*AN*FL2+DSIGE*AN1*FL1+DSIGE*AP*FLP
 2390 C PAR TS =A U PI EC E+A U SPACER+S.S.SHOES+SIMO NSHOE+W COLD SHOE
 2400 WPARTS=.573+DAU*.040*PI*.25**2.*16.38/4.+DSS*.060*2.54*AN+DSIMO*A
       N1*2.54*.020+DW*.020*AN1*2.54
 2410 C WEIGHT OF NUT = WNUT = S.S. NUT AND WASHER + COLD STACK WITH W P.C.S.
 2420 WNUT=5.71+18.846
 2439 WCONN =DCU*(FH+FAS)*2.*RROP*2.54*.020
 2440 WIOI = WHP+WSIGE+WPARIS+WNUT+WCONN
 2450 WGCPL =W TO T* GCPL
 2460 WFIN=DBRYL*AFIN*TTFIN
 247@ WFOIL =DFOIL*FLP*((SQRT(GAC)+2.*FLP)**2.+4.*(SQRT(GAC)+FLP)*FLG+GA
       T-GCPL*PI*RROP**2.)
 2480 WCASE =DBRYL*ACASE* TTCASE*1.3
 2490 WGEN=(WGCPL+WFIN+WFOIL+WCASE)/453.59
 2500 WGPO=GPO/WGEN
 2510 GETA =GPO/GQT
 2523 TFCC = TFC - 273.
 2530 THRC = THR - 273.
 2540 THPC =THP -273.0
 2550 TCPC=TCP-273.0
 2560 TCRRC = TCRR - 273.
 2570 TCBC =TCB -273.0
 2580 THN2C = THN2 - 273.0
 2598 TCN2C = TCN2 - 273.0
 2600 TCNC=TCN-273.0
 2610 THNC=THN-273.0
```

```
2620-99999
 2620 TCVC=TCV-273.0
 2630 TCAC =TCA -273.0
  2647 FLI =FLI /2.54
 2650 FL2=FL2/2.54
       DTRR =D TRR /2.54
 2662
 2670 GAT=GAT/6.45
 2680 PHI = (GQH+GQSO+GQSI)/(GCPL*AH)
 2690 AH = AH/6.45
 2700 AFIN=AFIN/6.45
 2710 ETA =PO/(QH+QSI+GQSO/GCPL)
 2720 PDUMP WGPO, WGEN, GQSO, GQH, GQSI, GQCASE, GQT, GCPL, GAT, GPO, GEL, THRC, TF
       CC.ETAN.TTFIN.ETAF.AFIN.WGCPL,WTOT,WFIN,WFOIL.WCASE,ETA,TCRRC,DFO
        IL
 2730 PRINT 9100. THPC.AN.FLI.GETA
 2740 9100 FORMAT('
                          THPC = 'F7.1,7X,'AN='F7.4,6X,'FL1 = 'F7.4,5X,'GETA = "
        77.5
 2750 PRINT 9101, TCPC, AP, FL2, PO
                          TCPC = 'F7.1.7X, 'AP = 'F7.4.6X, 'FL2 = 'F7.4.7X, 'P0 = 'F7
 2750 9101 FORMAT(*
       .4)
 2770 PRINT 9102, TCBC, DTRR, GQT, EL
                          TCBC = 'F7.1.5X.'DTRR = 'F7.4.6X.'GQT= 'F7.0.7X.'EL = '
 2780 9102 FORMAT('
       F7.4)
  2790 PRINT 9103, THNC, THN2C, WGPO, R
 2800 9103 FORMAT('
                          THNC = 'F7.1, 4X, 'THN2C = 'F7.2, 5X, 'WGPO = 'F7.3, 8X, 'R =
       'F7.5)
 2810 PRINT 9104.TCNC.TCN2C.WGEN.PHI
 2820 9104 FORMAT('
                          TCNC = 'F7.1, 4X, 'TCN2C = 'F7.2, 5X, 'WGEN = 'F7.2, 6X, 'PH
       I = F7.4//)
 2830 1000 CONTINUE
 2840 STOP
 2850 9000 PRINT 9001
 2860 9001 FORMAT(' TOO MANY SEARCHES FOR QH')
 2870 9005 PRINT 9006
 2880 9006 FORMAT('
                         TOO MANY SEARCHES FOR GCPL')
 2890 GO TO 1000
 2900 9110 PRINT 9111
                         FUEL CAPSULE EXCEEDS GAT')
 2910 9111 FORMAT('
 2929 GO TO 1990
 2930 GO TO 1000
 2940 9114 PRINT 9115
 2950 9115 FORMAT('
                          TOO MANY SEARCHES FOR ETAF')
 2967 GO TO 1070
```

APPENDIX II-C COMPUTER PROGRAM DATA TABLES

1. <u>HYGEN Data Tables</u>
a. 63.5 at.% Si - n- and p-SiGe

Seq	ν ος	<u>a. 63.</u>			_	^
#	Kp Temp	<u>s</u> n	$\rho_{\rm N}$	$K_{\rm N}$	<u> </u>	
10	SIGEL 500					
20	1000 .951590	.000263	.001860	.047000	.000284	.002870
30	980	.000267	.001950	.045400	.000282	.002820
40	969	.000270	.002050	.044200	.000279	.002770
50	949	.000274	.002150	.043100	.000277	.002720
50	920 .045900	.000278	.002260	.042200	.000274	.002670
7 €,	900 .045100	.000281	.002370	.041400	.000272	.002620
89	880 .044500	.000285	.002490	.040800	.000270	.002580
98	860 .044200	.000289	.002620	.040190	.000268	.002530
100	840 .043900	.000293	.032780	.039600	.000265	.002480
110	820 .043800	.000297	.002920	.039000	.000263	.002440
120	800 .043600	.000302	.003080	.038500	.000260	.002400
130	780 .043600	.000306	.003260	.038100	.000257	.002340
140	760 .043700	.000310	.003420	.037700	.000255	.002300
150	740	.000314	.003620	.037350	.000252	.002260
160	720	.000318	.003800	.037100	.000249	.002220
170	700 .044200	.000322	.003990	.036850	.000246	.002170
180	680 •044400	.000326	.004180	.036700	.000243	.002130
190	660 .044700	.000330	.00 43 60	.036600	.999249	.002080
200	640 .044900	.000333	.004540	.036500	.000236	.002040
210	623 •245133	.000337	.004720	.036600	.000233	.002000
229	699 • 945499	.000340	.004890	.03 663 0	.000230	.001960
239	580 .045700	. 0003 43	.005080	.036680	.000226	.001920
249	560 •345000	.000346	.005240	.936850	.000223	.001880
259	549 •945339	.000349	.005420	.037100	.000220	.001840
260	520 .046500	.000350	.005530	. 03 73 50	.000217	.001795

a. 63.5 at. Si - n- and p-SiGe (contd.)

Seq #	$\mathcal{K}_{\mathtt{p}}$	°C Temp	5 _N	$\rho_{\rm N}$	K _N	_Sp	Pp
278	3 47 9 9	500	.000349	.005590	.037650	.000213	.001700
289	.04732	430	.000348	.005570	.037950	.000209	.001710
293	.24772	460	.000345	.005440	.038300	.000206	.001670
323	.04810	448	.000339	.005240	.038600	.000202	.001635
312	.04850	420	.000333	.005020	.039050	.000199	.001590
320	.84538	433	.000326	.004790	.039500	.000195	.001550
339		380	.000320	.004520	.039950	.000191	.001510
3 40	.04940	3 50	.000312	. 00 42 60	.043430	.000187	.001480
359	.74930	3 40	.000305	.004000	.040900	.000183	.001440
359	.05020	320	.000297	.003740	.041400	.000179	.001400
370	.05070 .05120	300	.000289	.003510	.042000	.000175	.001370
380		280	.000280	.003290	.042550	.000171	.001340
390	.05170	260	.000272	.003100	.043100	.000167	.001300
400	.05220	240	.000264	.002920	.043700	.000162	.001260
410	.05250	220	.000256	.002760	.044400	.000158	.001240
42 9	.05370	200	.000247	.002600	.044900	.000154	.001200
430	. 875 70	130	.000239	.002450	.045650	.000149	.001170
440	.04700	160	.000230	.002320	.046400	.000145	.001140
45@	.05529	140	.000221	.002190	.047000	.000140	.021110
450	.85579	123	.000213	.002080	.047750	.000136	.001080
473	.05620	100	.000205	.001980	.048500	.000130	.001055
489		89	.000195	.001880	.049300	.000126	.001030
490	.05589°	63	.022186	.001790	.750000	.000122	.031090
500	.05720	42	.000177	.001710	.050800	.000117	.000980
510	.35783	2.3	.3331 69	.001650	.051600	.000113	. 000969
528	.05840 .05930	91	. 0001 61	.001590	.052400	.000109	.000940

b. 3M - 3N PbTe

530 C PBTE-3N DATA RICHARDS (AUG '68) P-TYPE SIMILAR TO 2P

Seq.	Kp Temp	S _N		\mathcal{K}_{N}		$\mathcal{P}_{\mathtt{p}}$
550	600	.000226	.004640	.0208	.000224	.00454
560	.0144 580	.000231	.004420	.0197	.000226	.00440
579	.9135 569	.000234	.004160	.0188	.000227	.00433
582	.0127	.000235	.093890	.0180	.000228	.99427
590	.0121 520	.000234	.003640	.0174	.000227	.00418
ସେଷ	.0117 500	.000233	.003 400	.0170	.000225	.00406
610	.0112 480	.000231	.003150	.0167	.000223	.00393
62 Ø	.0109 460	.000227	.002910	.0164	.000220	.00379
630	.0106 440	.000223	.002680	.0163	.000216	.00364
640	.0104 420	.000219	.002460	.0162	.000211	.00348
650	.0106 400	.000214	.002240	.0162	.000205	.00331
660	.0105 380	.000208	.002030	.0162	.000199	.00313
670	.0106	.000203	.001830	.0163	.000192	.00297
682	.0108 340	.000197	.001670	.0164	.000184	.00280
690	.1111	.000190	.001520	.0165	.000176	.00264
700	.0115 300	.000184	.001380	.0168	.000168	.00248
710	.0119 230 .0124	.000178	.001270	.0171	.000160	.00234
720	260 .0130	.900172	.001160	.0175	.000151	.00220
73 Ø	240	.000166	.001040	.0179	.000144	.00206
743	.0136 220 .0142	.000160	.000930	.0184	.000136	.00193
75%	200	.000154	.000840	.0190	.000129	.00182
763	.9156	.999147	.000750	.0196	.000121	.00170
773	163	.000141	. ØØØ66Ø	.0203	.000114	.00159
788	140 .3171	.900134	.000580	.3211	.000107	.00149
790	129 .0178	.000128	.900520	.0222	.000101	.00140
899	100	.030121	.000470	.0230	.000094	.00131

b. 3M - 3N PbTe (contd.)

Seq Kp °C Temp	5' _N	P_{N}	$\mathcal{K}_{\mathtt{N}}$	$\mathcal{S}_{\mathtt{p}}$	Pp
810 89	.000114	.000420	.0243	.000088	.00124
.0194 820 50 .0203	.000106	.0003610	.0253	.000083	.00116
330 40 .0211	.000099	.000320	.0275	.000078	.00110
8 40 29 •9220	.000001	.000280	.0292	.000074	.00104
850 Ø •3229	.000082	.000240	.0312	.000070	.00100
855 /ENDO 860 MK202469 870 1.04E-3 880 0.83E-3 890 0.67E-3 900 0.54E-3 910 0.34E-3 920 0.34E-3 930 0.27E-3 940 0.20E-3 950 0.14E-3 950 0.10E-3 970 0.05E-3	MinK 2020 Fibro	ous Insulatio	n		

2. <u>HDATAl Data Tables</u>
a. 80.0 at.% Si - n- and p-SiGe

_		<u>a.</u>	80.0 at.%	S1 - n- and	<u>p-SiGe</u>		
Seq #	<u> R</u> p	°C <u>Temp</u>	S_{N}	$\rho_{\rm N}$	$K_{\rm N}$	_ S p	Pp
19-273	61056	2					•
13 28	SIGE89	1103	.000225	.301300	.746296	.0003339	.293378
3.9	.34878	1989	.999224	.001320	. 245977	.000336	.003340
42	.04791	1252	.20224	.001340	•Ø45558	.000332	.203310
57	.7475?	1943 15	.000224	.001350	.045248	.000329	.003280
ସେ	. 94583	1020 3	.9.9322.6	.001360	.044742	.000325	.003240
7.8	• Ø 4 52 S	1000 6	. 000228	.001380	.044444	.999321	.003190
87	. 24575	980 SS	.998229	001420	.043917	.000318	.003150
98	. 9 4 5 2 4		.999231	.001 450	.043764	.000314	.993119
133	. 94434		.000234	.001 490	.043478	.000311	.003070
110	. 84444		.000236	.031 480	.043196	.000308	. 9 93 93 9
123	. 3 4 40 5	_	.000239	.001520	.042918	.000304	.002970
130	. 34376		.000242	.001 580	.042553	.000301	.022930
1 40	.04338		.078246	.001650	.042372	.000298	.072380
1 60	. 24329		.000248	.001720	.042105	.000296	.002840
1 70	. 94317		.000251	.001800	.041841	.000293	.002790
180	.94319	800 3 780	.000254 .000257	.001850	.041666	.000290	.002750
198	. 94317		.237263	.001960 .002050	.041580	.000286	.002 69 0
299	. 24312		.900256	.002160	.041322	.000283	.002640
210	. 3,431 3		.000271	.002250	.041152	.000279	.002 500
220	. 34314		.000276	.032420	.041152	.000273	.002510
230	. 943233	3 683	.999282	.702.569	.349816	.000269	.002470
2 40	. 94329:		.300286	.972710	.049816	.000265	.002470
2 53	. 843291	648	.000299	.002860	.041152	.000262	.992389
2 SØ	. 34353	62 3	.000294	.002980	.041322	.900258	.002340
279	. 24370	699	.000293	.002930	.041493	.000255	.992299
170	.343378	3					

a. 80.0 at.% Si - n- and p-SiGe (contd.)

Seq #	$-\mathcal{K}_{p}$	°C Temp	S _N	ρ_{N}	K _N	_S _p	Pp
. 239-53: 289	•	533	.000291	.072920	.041701	.000251	.002250
299	.44752	569	.030288	.092810	.041963	. 999248	.002210
300	. 4424	549	.000285	.982728	.042194	.898244	.002160
313	.74444	52.9	.999231	.002620	.042462	.000241	.002120
329	.34496	533	.037275	.002560	.942698	.999237	.002080
337	. 452A	436	.092271	.002500	.042918	.000233	.002040
3 49	.045454	46%	.030266	.072437	.943196	.009230	.001990
3 5 3	.345632	449	.773251	.082368	.043478	.000226	.001950
369	. 845999	420 3	.পপ্ৰপ্ৰ255	.002300	.943763	.909222	.001910
373	.04529	4 ଅଅ S	.070249	.032240	.044952	.000218	.001870
33%	. 445533		.0002 43	.002170	.044365	.030214	.001830
398	.845882		.000237	.002110	.044642	.000210	. 991 799
400	.047163		.000231	.992050	.044984	.000206	.001750
41 9	.7471 69		.000225	.001990	.045310	.000202	.001710
42 7	.047845		.993219	.001930	.045662	.000198	.401670
43 %	. 48216		.030214	.031880	.045977	.000193	.001630
449	.348543		.989287	.001820	.046446	.000188	.001590
463	.348971	240	.000201	.991719	.046728	.000184	.001560
47%	.249235		.000199	.001660	.047393	.000179	.001490
483	.349751		.000182	.001610	.047801	.030168	.001460
49.7	.353359		.000175	.001560	.048169	.992162	.001430
539	.757479		.933169	.931529	.948543	.999156	.001400
51 7	.151899	120 120	.000162	.001470	.948947	.700149	.001370
52 A	.951232		.933156	.001430	.049309	.700142	.001350
53 3	.351733	33	. 9991 49	.001400	.049751	.000135	.091320
	.252113	1					

80 at. Si - n- and p-SiGe (contd.) °C Pp $\mathcal{K}_{\mathtt{N}}$ K_p S_{N} $\rho_{\rm N}$ Sp Temp Seq_# 549-899 5 40 59 .999142 .001360 .050175 .000125 .001300 .05257S 550 43 .000135 .001330 .050556 .000116 .001280 . 852966 568 23 .000128 .001290 .051020 .000106 .001260 .053475 579 0 .999121 .001270 .051493 .000095 .001240 .054054 589 b.RCA PBTE DATA TABLE n and p alloys 59% PBTEO 528 592 .000253 .095600 .013240 .000288 .007850 .012848 613 537 .000257 .005150 .012650 .007400 .000295 .013415 528 563 .000260 .004950 .012340 .000300 .007100 .039551 63% 543 .000264 .004790 .011620 .000302 .376630 . 009259 543 52 8 .000266 .004620 .011290 .000302 .006200 .009000 650 500 .000268 .004430 .010983 .030300 .005850 . 223849 669 433 .000268 .004260 .010750 .000297 .005500 .003771 670 460 .000266 .004070 .010630 .000293 .005150 .003710 689 443 .000263 .003800 .010540 .000287 .004850 .003733 690 42 3 .000259 .003590 .010430 .000280 .004500 .008849 700 400 .000254 .003350 .010520 .000272 .004150 . 0039 63 712 330 .000248 .003100 .010700 .000262 .003840 .309132 729 360 .033242 .002830 .010920 .000254 .003480 . 339339 732 348 .000235 .992579 .011230 .399243 .003140 .009708 742 320 .0002229 .092350 .011590 .000231 .002830 .310333 759 .999221 303 .992100 .712190 .000217 .002550 .010449 7.53 237 .000214 .001930 .012730 .000202 .002340 .012928 770 26% .000206 .001720 .013150 .000187 .001920 .311627 780 243 .002193 .001560 .213799 .000173 .001 750 .312422 793 223 . 0 0 0 1 1 1 .031 42 3 .014490 .000161 .001550 .313227 832 200 .000193 .201259 .015150 .000148 .001350

. 91 42 35

	بقيتم	KCA PbTe Dat	<u>a Table - n-</u>	and p-Alloy	s (contd.)	
⊵eq #	$\underline{\mathcal{K}_{\mathtt{p}}}$ Temp	<u>5n</u>	PN	<u>Kn</u>	<i>S</i> p −	<u> ₽</u>
819-900						
810	130	.000175	.001100	.015870	.000135	.001180
82 8	.015197 160 .016393	.000166	. 3389 78	.016660	.999121	.001040
830	140	.000159	.000860	.017540	.999198	.000920
. 840	120 .418796	.000150	.000775	.018510	.000097	.000798
853	199	.0001 42	.900695	.019600	.000085	.000695
860	.021413	.999133	.000605	.020700	.000075	.000605
370	.922727	.000124	.000540	.921889	.000066	.000525
880	.024096	.000117	.003475	.023250	.%%%%58	.000450
890	.025316	.000138	.093440	.024590	.339858	.000380
900	.725315	.୭୭୭୮୭୭	.020370	.025310	.920042	.000350

APPENDIX II-D INPUT/OUTPUT COMPUTER PROGRAM SYMBOLS

1. Hybrid Couple Program Symbols

Area of Heat-Reception Plate

AΗ

AN	Cross-Sectional Area of n-type SiGe Thermoelement				
ANP	Cross-Sectional Area Ratio of n-Type PbTe Thermoelement/p-Type				
	SiGe Thermoelement				
ANR	Cross-Sectional Area of n-Type SiGe Element/ n-Type PbTe Element				
AP	Cross-Sectional Area of p-Type Thermoelement				
CUR	Current Flowing Through The Thermocouple				
DTRR	Wall Thickness of p-type Element				
EHP	Emissivity of Inner Surfaces of Heat-Reception Plates				
EL	Load Voltage				
EM	Ratio of Load to Internal Electrical Resistance				
ETA	Efficiency of Thermocouple				
ETC	Emissivity of Inner Surfaces of Heat-Rejection Plates.				
FLP	Length of SiGe p-type Element				
FL1	Length of SiGe n-type Element				
FL2	Length of PbTe n-type Element				
GAP	Distance Between Surface of n-type PbTe Inner Segment and Inner Wall				
	Surface of p-type Cylinder				
GKA	Thermal Conductivity of n-type SiMo Hot Shoe				
GK B	Effective Thermal Conductivity of Interface Materials Between n-type Segmen				
GKC	Effective Thermal Conductivity of n-type Cold Stack				
GKCP	Thermal Conductivity of Heat-Rejection Plate				
GKD	Effective Thermal Conductivity of p-type Cold Stack				

Thermal Conductivity of p-SiMo Heat-Reception Plate

Heat Flux Incident on Heat-Reception Plate

GKHP

PHI

- PO Electrical Power Output per Couple.
- QT Total Heat Incident on Heat-Reception Plate
- QSI Total Amount of Shunt Heat Loss Between Heat Reception and Rejection Plates
 Which Flows in the Gap Between these Two Surfaces.
- QSO Amount of Shunt Heat Flowing thru Insulation and External to the p-type Element
- QH Heat Absorbed by the Thermocouple Materials
- R Total Electrical Resistance of Couple
- RCN Contact Resistivity of SiGe n-type Thermoelement per contact
- RCNZ Contact Resistivity of PbTe n-type Thermoelement per contact
- RCP Contact Resistivity of SiGe p-type Thermoelement per contact
- RHOA Electrical Resistivity of n-type SiMo Hot Shoe
- RHOB Effective Electrical Resistivity of Metallic Interface Material
- RHOD Effective Electrical Resistivity of n-type Cold Stack to End of Electrical Connector
- RHOC Effective Electrical Resistivity of p-type Cold Stack to End of Electrical
 Connector
- RSH Assummed Resistance Value of SiMo n-type tod p-type Junction in Heat Receptor
- RRN Radius of n-type PbTe Element
- SIGP Stefan-Boltzmann Constant
- TCBC Heat Rejection Base Plate Temperature, °C
- TCNC Heat Rejection SiGe N-Type Plate Temperature, °C
- TCN2C PbTe n-type Cold Junction Temperature. °C
- TCPC SiGe p-type Element Cold Junction Temperature, °C
- THNC SiGe n-type Hot Junction Temperature, oc
- THNZC-PbTe n-type Hot Junction Temperature, oc
- THPC SiGe p-type Hot Junction Temperature, °C
- THRC SiMo p-type Hot Shoe Edge Temperature, °C

TTA Thickness of n-type Si-Mo Hot Shoe

TTB Composite Thickness of n-type Interface Materials

TTC Composite Thickness of n-type Cold Stack

TTCP Thickness of Heat Rejection Plate

TTD Composite Thickness of p-type Cold Stack

TTHP Thickness of p-SiMo Heat Reception Plate

2. Converter Program Symbols

A STUD Surface Contact Area Of Couple Cold Stack Incontact With

Mounting Plate

AH Area of Heat-Reception Plate

AN Cross-Sectional Area of n-Type SiGe Thermoelement

ANP Cross-Sectional Area Ratio of n-Type PbTe Thermoelement/p-Type

SiGe Thermoelement

ANR Cross-Sectional Area Ratio of n-Type SiGe Element/n-Type PbTe Element

AP Cross-Sectional Area of p-Type Thermoelement

CUR Current Flowing Through The Thermocouple

D Dimensionless Quantity

D FOIL Density of Multi-Layered Foil

DTRR Wall Thickness of p-Type Element

EHP Emissivity of Inner Surfaces of Heat-Reception Plates

EL Load Voltage

EM Ratio of Load to Internal Electrical Resistance

ETA Efficiency of Thermocouple

ETC Emissivity of Inner Surfaces of Heat-Rejection Plates

FAS Distance Between Adjacent Heat Receptors

FKIC Effective Thermal Conductivity of Dynaquartz Insulation

FKIE Effective Thermal Conductivity of Composite Foil Insulation

FLG Total Case Height Above Fin Mounting Plate

FLP Length of SiGe p-Type Element

FL1 Length of SiGe n-Type Element

FL2 Length of PbTe n-Type Element

GAP Distance Between Surface of n-Type PbTe Inner Segment and Inner

Wall Surface of p-Type Cylinder

GAT Planar Area Occupied By Couples Facing Heat Source

GCPL Number Of Couples

GEL Converter Output Voltage Delivered to Load

GETA Converter Efficiency

GKA Thermal Conductivity of n-Type SiMo Hot Shoe

GKB Effective Thermal Conductivity of Interface Materials Between

n-Type Segments

GKC Effective Thermal Conductivity of n-Type Cold Stack

GKCP Thermal Conductivity of Heat-Rejection Plate

GKD Effective Thermal Conductivity of p-Type Cold Stack

GKHP Thermal Conductivity of Heat-Reception Plate

GPO Converter Output Power

PHI Heat Flux Incident on Heat-Reception Plate

PO Electrical Power Output per Couple

QT Total Heat Incident on Heat-Reception Plate

QSI Total Amount of Shunt Heat Loss Between Heat Reception and

Rejection Plates Which Flows in the Gap Between These Two

Surfaces

QSO Amount of Shunt Heat Flowing Through Insulation and External to

the p-Type Element

QH Heat Absorbed by the Thermocouple Materials

R Total Electrical Resistance of Couple

RCN Contact Resistivity of SiGe n-Type Thermoelement

RCNZ Contact Resistivity of PbTe n-Type Thermoelement

RCP Contact Resistivity of SiGe p-Type Thermoelement

RECA Electrical Resistivity of n-Type SiMo Hot Shoe

RHOB Effective Electrical Resistivity of Metallic Interface Material

RHOD Effective Electrical Resistivity of n-Type Cold Stack to End

of Electrical Connector

RHOC Effective Electrical Resistivity of p-Type Cold Stack to End of

Electrical Connector

RSH Assumed Resistance Value of SiMo n-Type to p-Type Junction in

Heat Receptor

RRN Radius of n-Type PbTe Element

SIGP Stefan-Boltzmann Constant

TCBE Heat Rejection Base Plate Temperature, °C

TCNC SiGe n-Type Element Cold Junction Temperature, °C

TCN2C PbTe n-Type Cold Junction Temperature, °C

TCPC SiGe p-Type Element Cold Junction Temperature, °C

TFCC Fuel Source Surface Temperature, °C

THNC SiGe n-Type Hot Junction Temperature, °C

THNZC PbTs n-Type Hot Junction Temperature, °C

THPC SiGe p-Type Hot Junction Temperature, °C

TT CASE Thickness of Case Protecting Insulation Around Heat Source

TTA Thickness of n-Type Si-Mo Hot Shoe

TTB Composite Thickness of n-Type Interface Materials

TTC Composite Thickness of n-Type Cold Stack

TTCP Thickness of Heat Rejection Plate

TTD Composite Thickness of p-Type Cold Stack

TTHP Thickness of Heat Reception Plate

W CASE Weight of Case

THRC SiMo p-Type Hot Shoe Edge Temperature, °C

W FIN Weight of Fin

W FOIL Weight of Foil

WGCPL Weight of Couples

WGPO Specific Power - Watts/lb.

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